



WFIRST Coronagraph Instrument

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On behalf of the CGI instrument & science investigation teams

Astro2020 EOS1 Panel

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Jet Propulsion Laboratory
California Institute of Technology

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Outline



- Overview of CGI
- Status of key technologies
- Comparison to future mission needs
- Path to delivery

Outline



- Overview of CGI
 - Tech demo, performance predictions, science impact
- Status of key technologies
 - All will be TRL6 by November
- Comparison to future mission needs
 - several key areas are “in family”
- Path to delivery
 - on schedule, on budget

CGI is step on the path toward biomarkers on an Earth-like planet

- Exoplanet Science Strategy recommendations:
 - NASA should lead a large strategic direct imaging mission capable of measuring the reflected-light spectra of temperate terrestrial planets orbiting Sun-like stars.
 - NASA should launch WFIRST to ... demonstrate the technique of coronagraphic spectroscopy on exoplanet targets.

“The most effective way to do it, is to do it.”

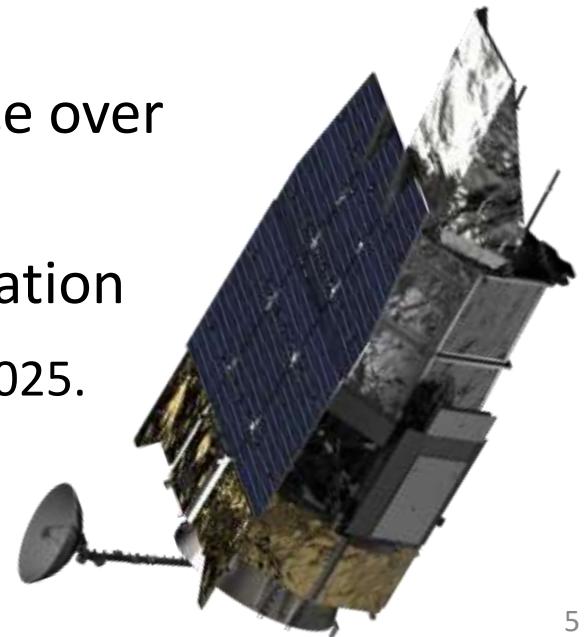
– Amelia Earhart

- System-level demonstration, on orbit

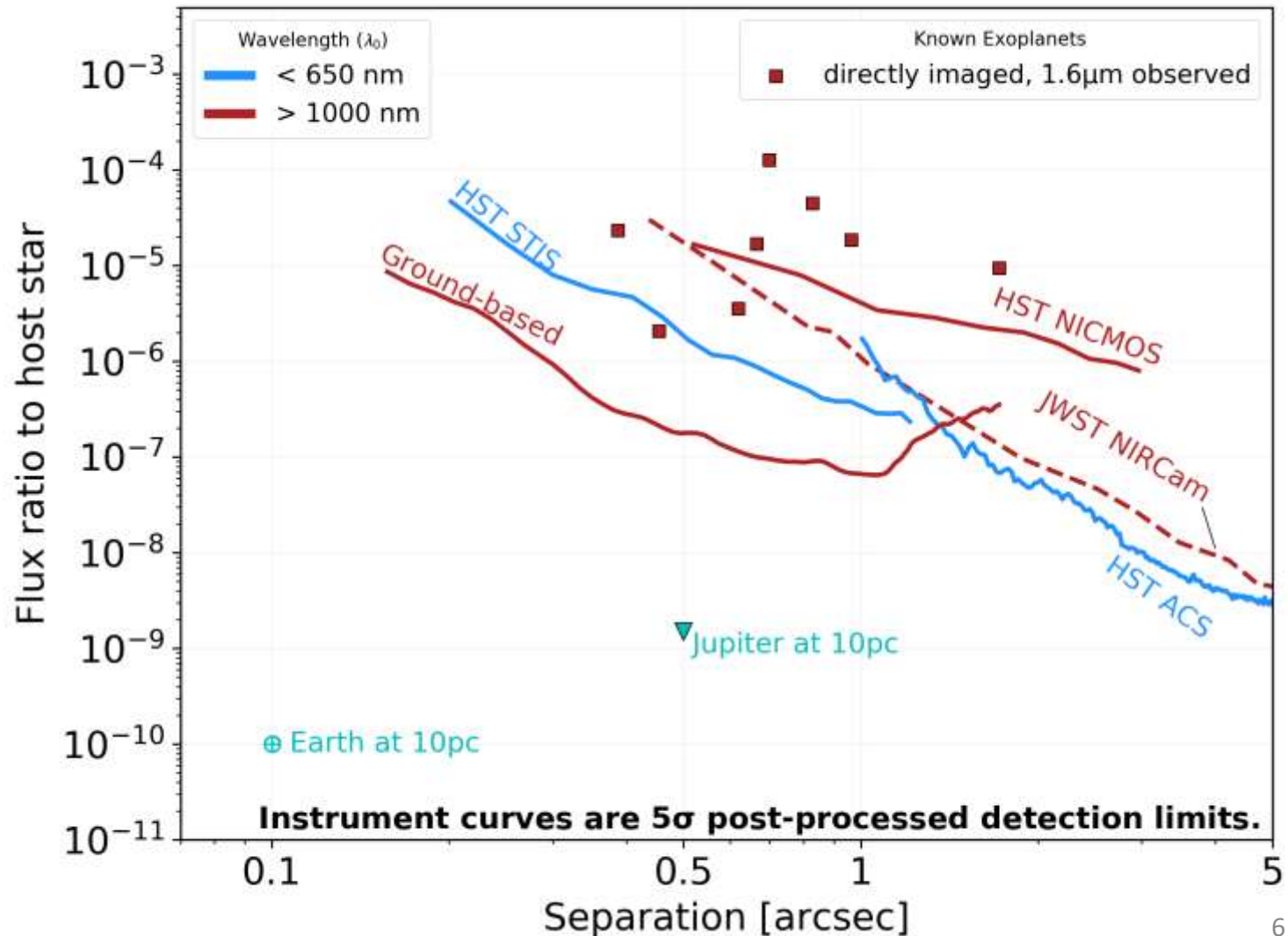


CGI paves the way for future direct imaging missions

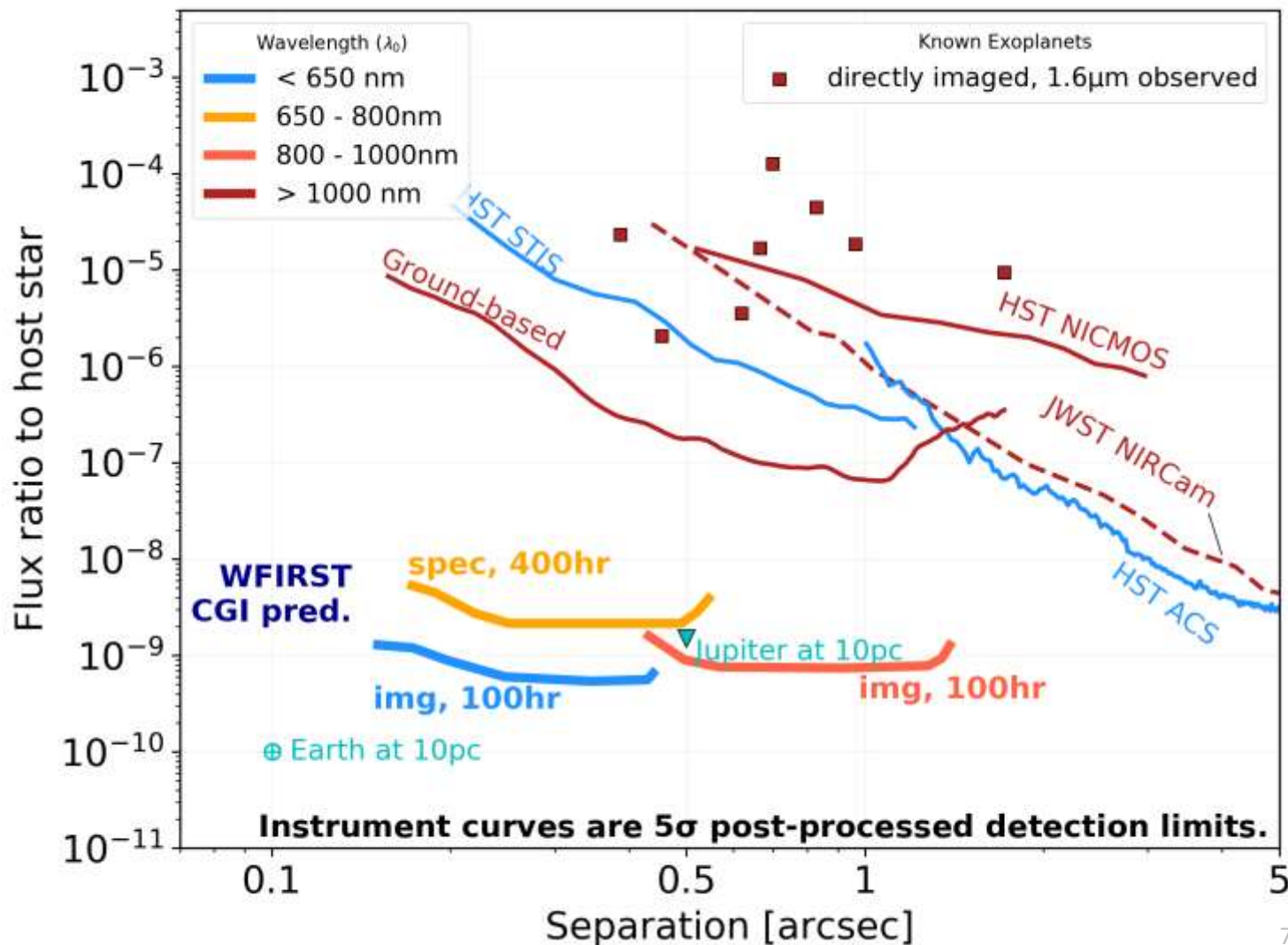
- CGI is:
 - a technology demonstration instrument on WFIRST
 - the first space-based coronagraph with active wavefront control
 - a **visible light** (545-865nm) imager, polarimeter and $R \sim 50$ spectrograph
 - a 1,000 times improvement in performance over current space facilities
 - confirmed by NASA, beginning implementation
 - Delivery to payload I&T in 2023. Launch late 2025.



Goal: bridge the gap between massive self-luminous planets (IR) and reflected light exo-Earths (visible)



CGI's predicted performance is 100-1000x better than SOTA



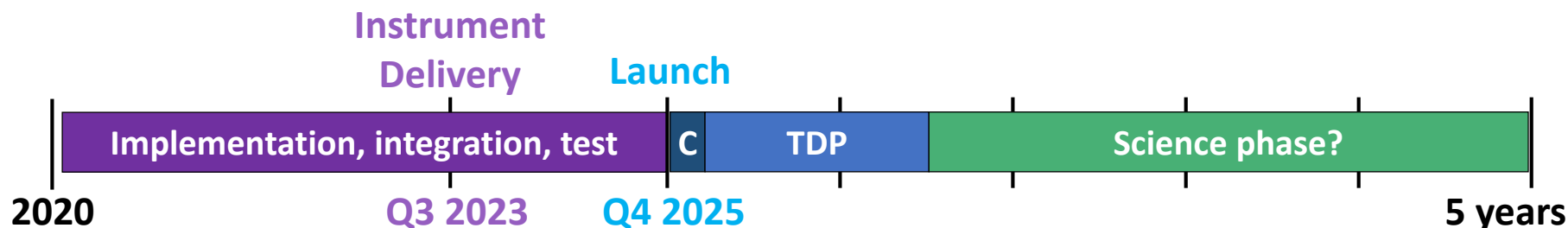
Based on lab demonstrations as inputs to high-fidelity, end-to-end models.

NASA terminology: MUF=1 predictions

Primary Observing Modes

Three modes will be fully tested prior to launch.

λ_{center}	BW	Mode	FOV radius	Polarimetry
575 nm	10%	Imager	0.14" – 0.45"	Y
730 nm	15%	Slit + R~50 Prism	0.18" – 0.55"	-
825 nm	10%	Imager	0.45" – 1.4"	Y



- Feb 2020: Entered implementation phase (Phase C)
- Q3 2023: Instrument delivery to payload integration & test
- Q4 2025: Launch
- Commissioning Phase
 - 450 hr in first 90 days after launch
- Technology Demonstration Phase (TDP)
 - ~2200 hr (3 months) baselined in first 1.5 years of mission
- If TDP successful, potential science phase
 - 10-25% of remainder of 5 year mission
 - Support community engagement with extended Participating Scientist Program and/or GO (with additional tools)
 - Requires additional resources
 - Starshade rendezvous, if selected

CGI will demonstrate key technologies for future missions

**Ultra-Precise
Wavefront Sensing
& Control**



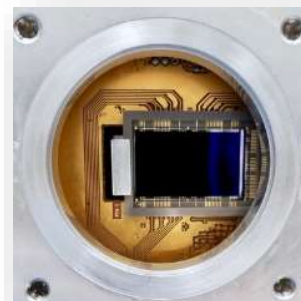
**Large-format
Deformable Mirrors**



**High-contrast
Coronagraph
Masks**



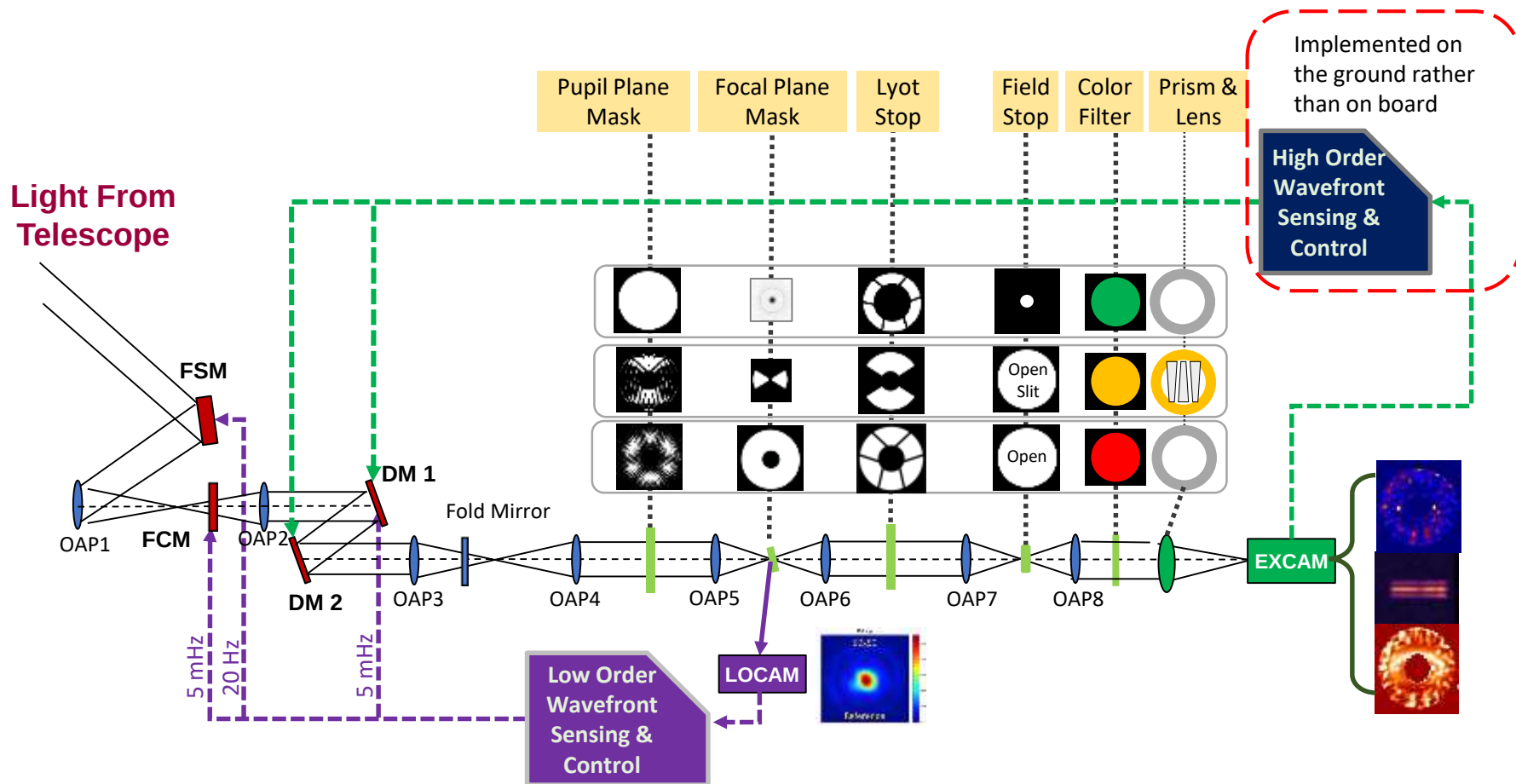
**Ultra-low-noise
photon counting
visible detectors**



**Photon-
Counting High-
Contrast Data
Processing**

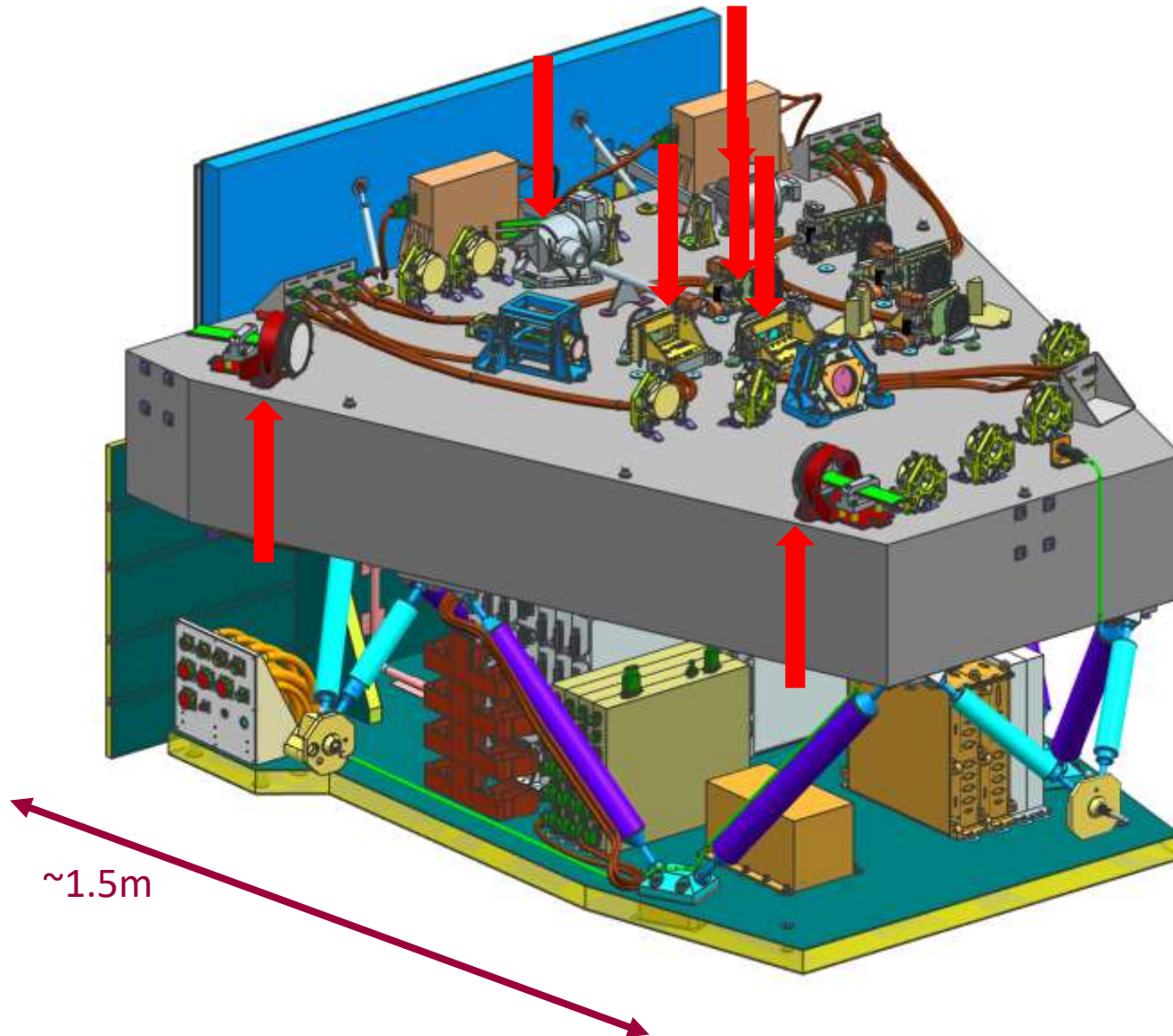


Key technologies work together as a system to deliver high performance



OAP = Off-Axis Parabolic [Mirror]

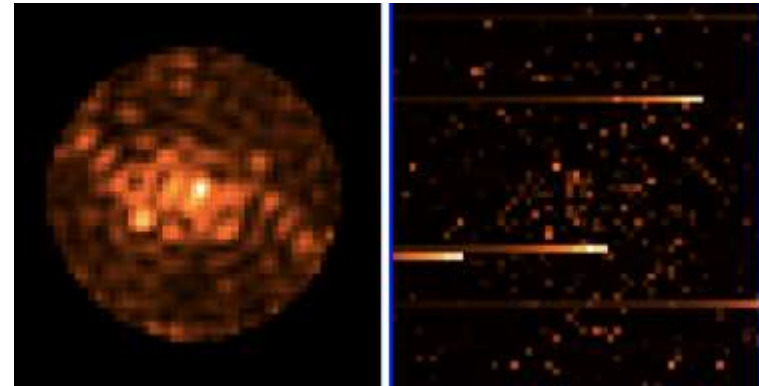
Key technologies work together as a system to deliver high performance



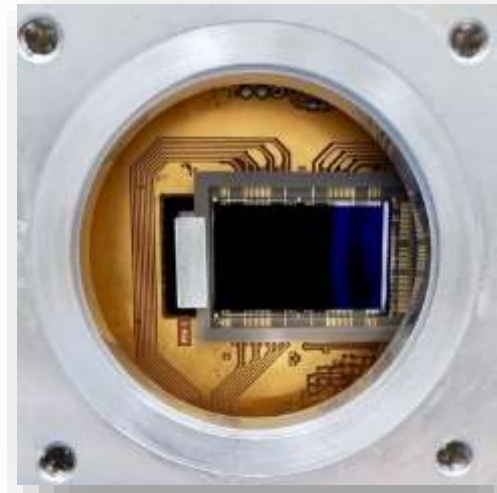
CGI is maturing photon-counting EMCCDs for space applications

Simulated images
John Krist (JPL)

- Low-flux images:
 - EXCAM: Jupiter analogs < 1 photon/min
 - LOCAM: 1kHz framerate
- EM $\Rightarrow$ $\sim$ no read noise
 - First space-qualified photon-counting EMCCD
- Tech & data processing development
 - mitigation and characterization of charge traps from radiation damage
 - Mitigation of cosmic ray effects (overspill)



Patrick Morrissey (JPL)

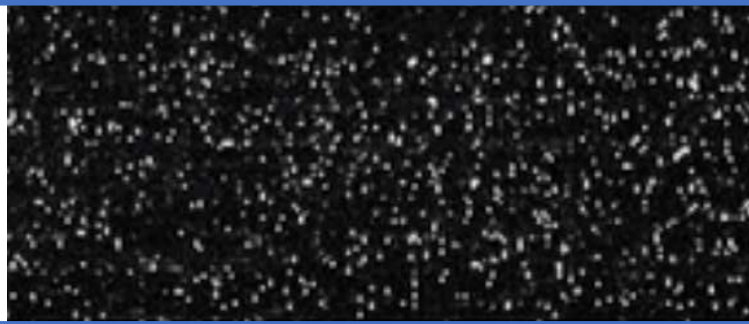


“notch” channels reduce effects of radiation damage up to $\sim 5x$

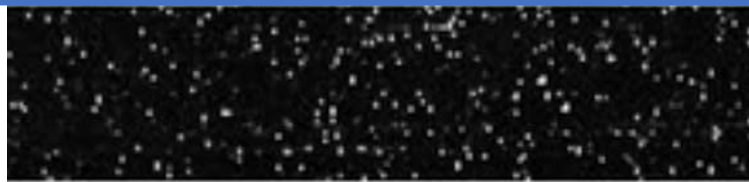
Patrick Morrissey (JPL)



Undamaged (shielded) region



commercial design: irradiated

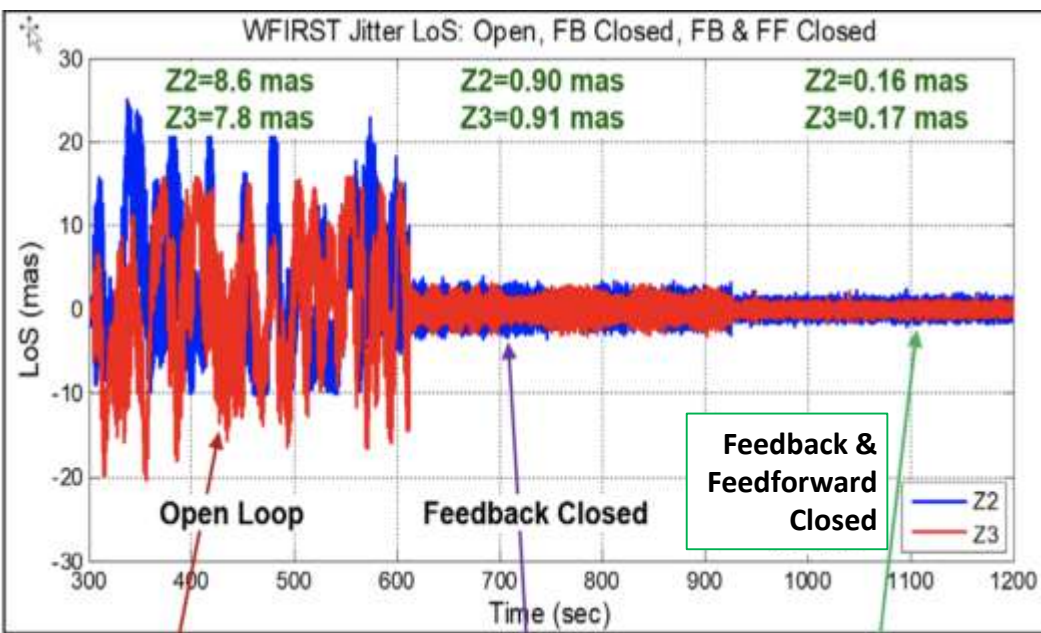


CGI “notch” design: irradiated

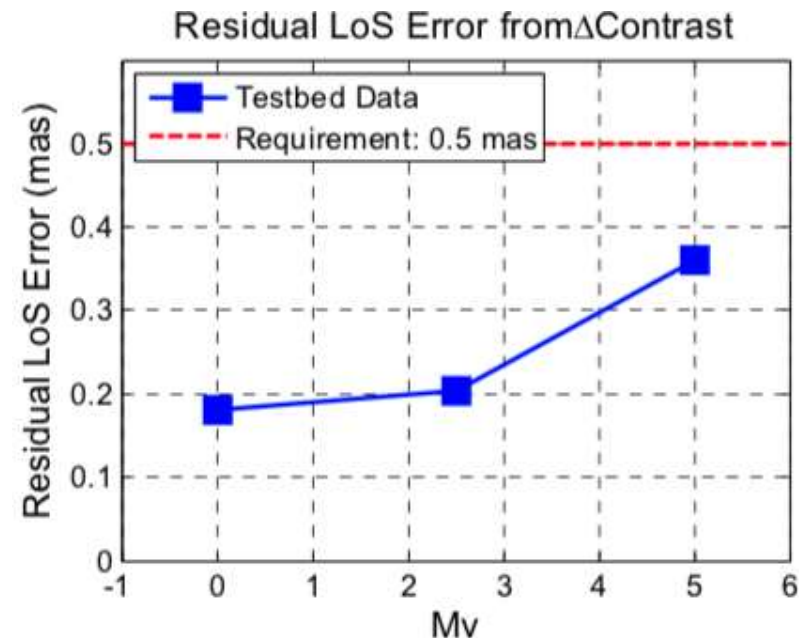
CBE effective QE @ 5 years $\sim 50\%$

CGI Low-Order Wavefront Sensor: 1st (in space) to use science light for control

Unlike HST & JWST fine guidance sensors, CGI LOWFS is designed to minimize non-common path errors & operate in low-photon regime.



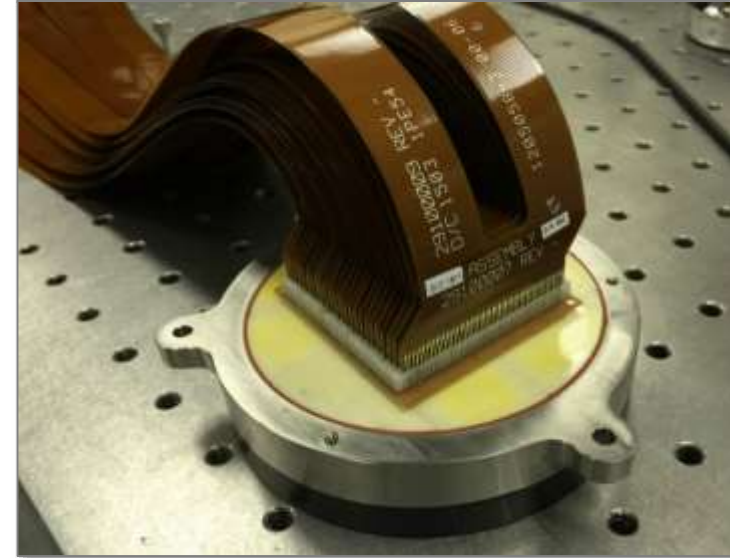
Shi+2019 lab demo: flight-like tip/tilt disturbances, bright “star.”



Shi+2018 lab demo

CGI is maturing high-actuator-count DMs for space applications

- CGI uses Northrop Grumman Xinetics Deformable Mirrors
 - 48X48 PMN (lead magnesium niobate) electro-strictive ceramics actuators
- Xinetics has strong **lab** heritage:
 - >10 years without failures
 - 4×10^{-10} contrast in testbed
- CGI is maturing to flight-ready:
 - Flight interconnect will be demonstrated to survive flight environment by November, 2020



All CGI coronagraph designs have achieved high contrast in the JPL testbed

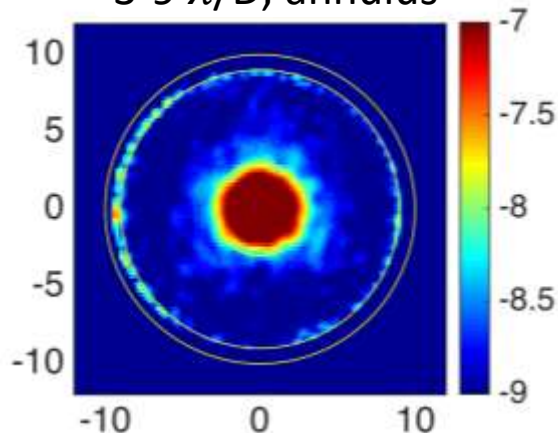
(results as of PDR)

Hybrid Lyot Coronagraph

Narrow-FOV Imaging

575 nm, 10% BW

3-9 λ/D , annulus



1.6e-9 raw contrast

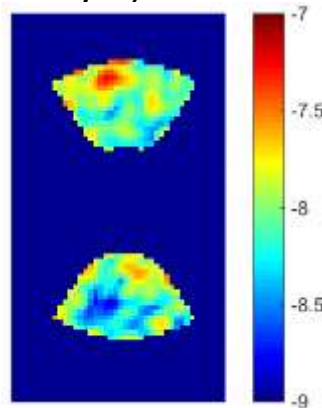
550 nm, 10% BW

“Bowtie” Shaped Pupil Coronagraph

Spectroscopy

730 nm, 15% BW

3-9 λ/D , Bowtie



1.0e-8 raw contrast

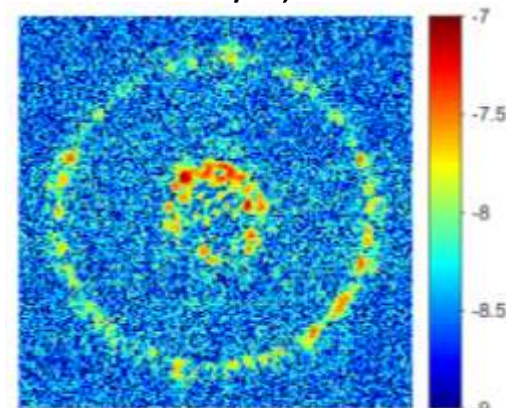
760 nm, 18% BW

“Wide” Shaped Pupil Coronagraph

Wide-FOV Imaging

825 nm, 10% BW

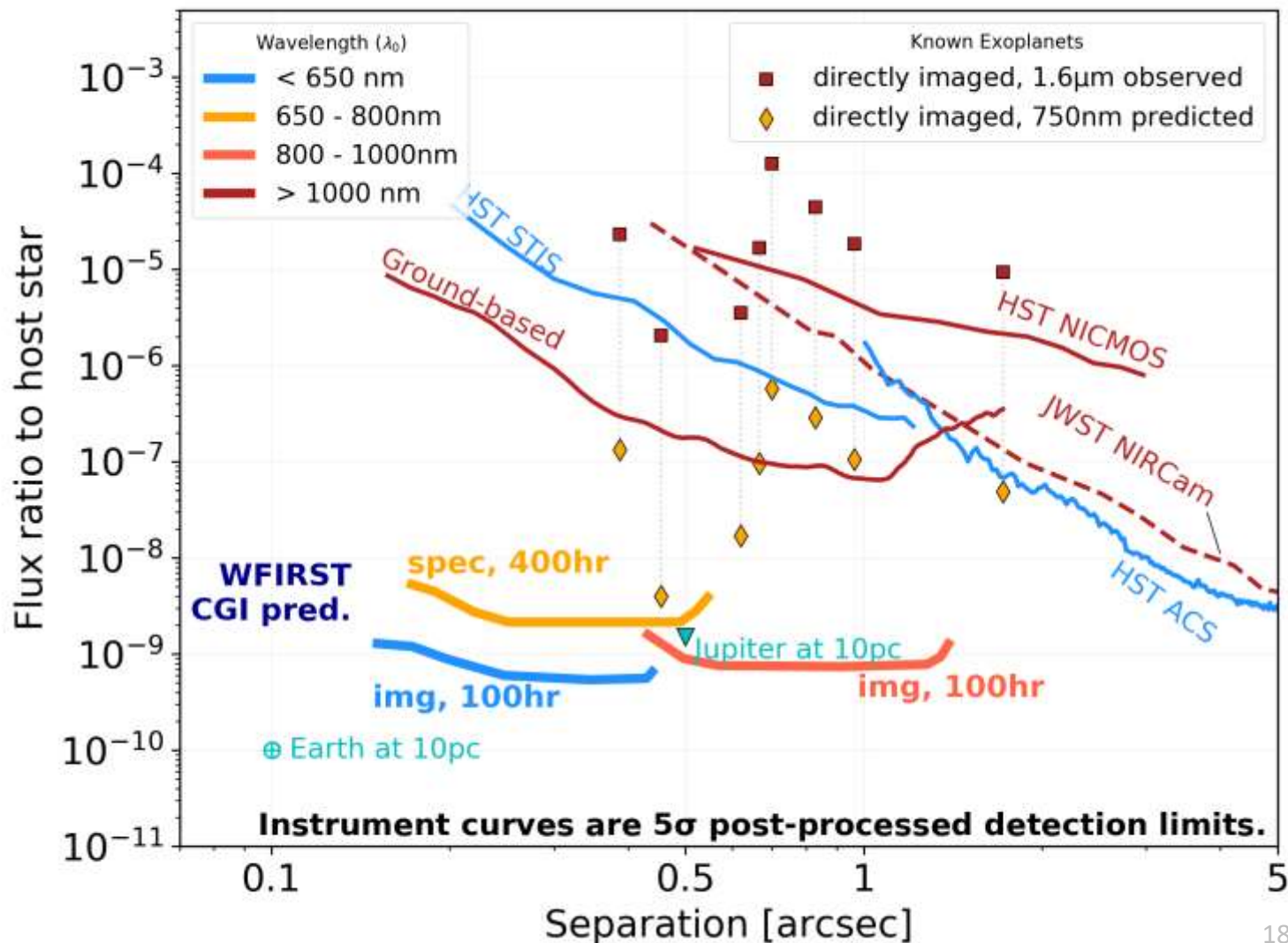
6.5-20 λ/D , annulus



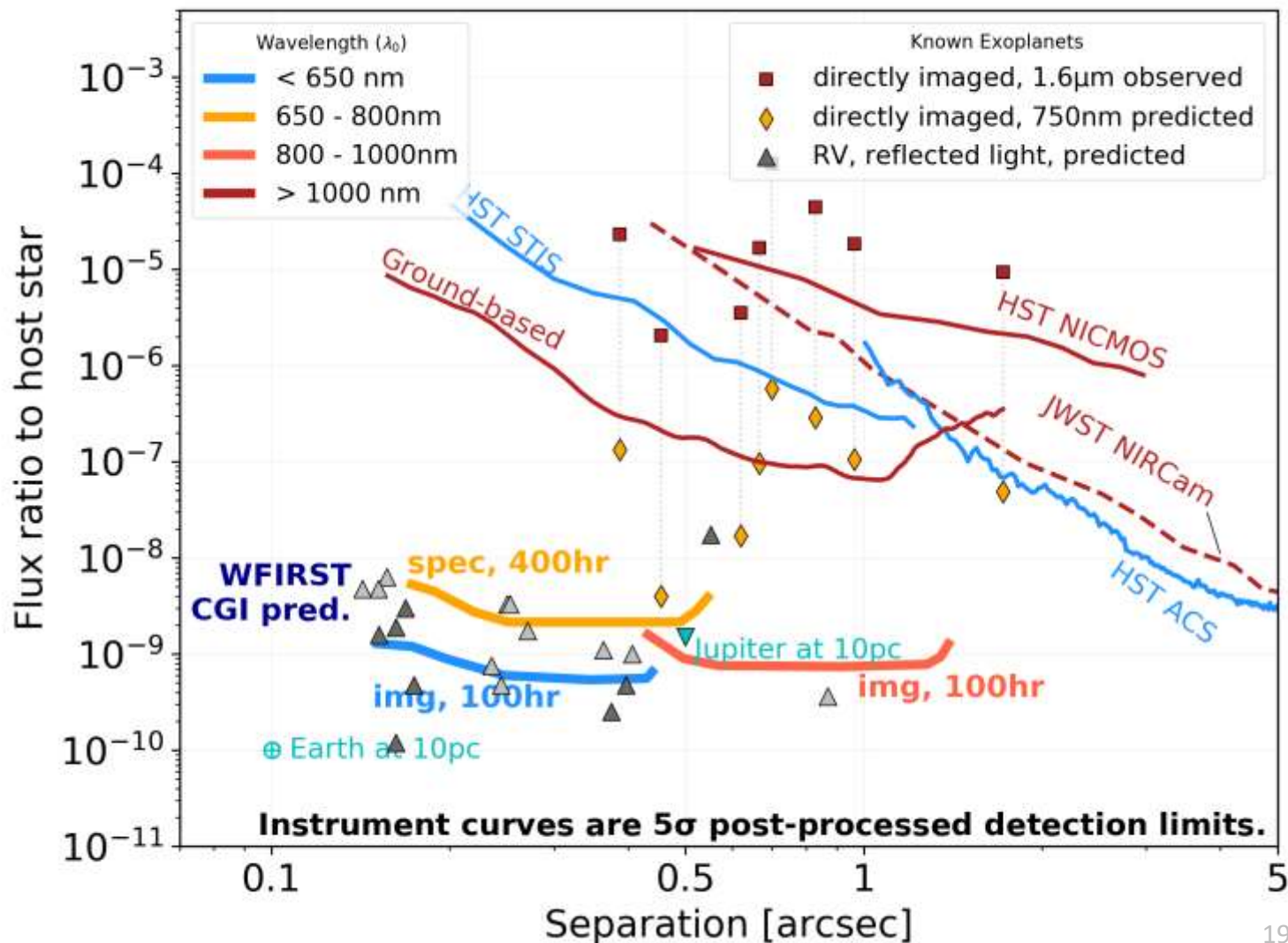
4.3e-9 raw contrast

565 nm, 10%

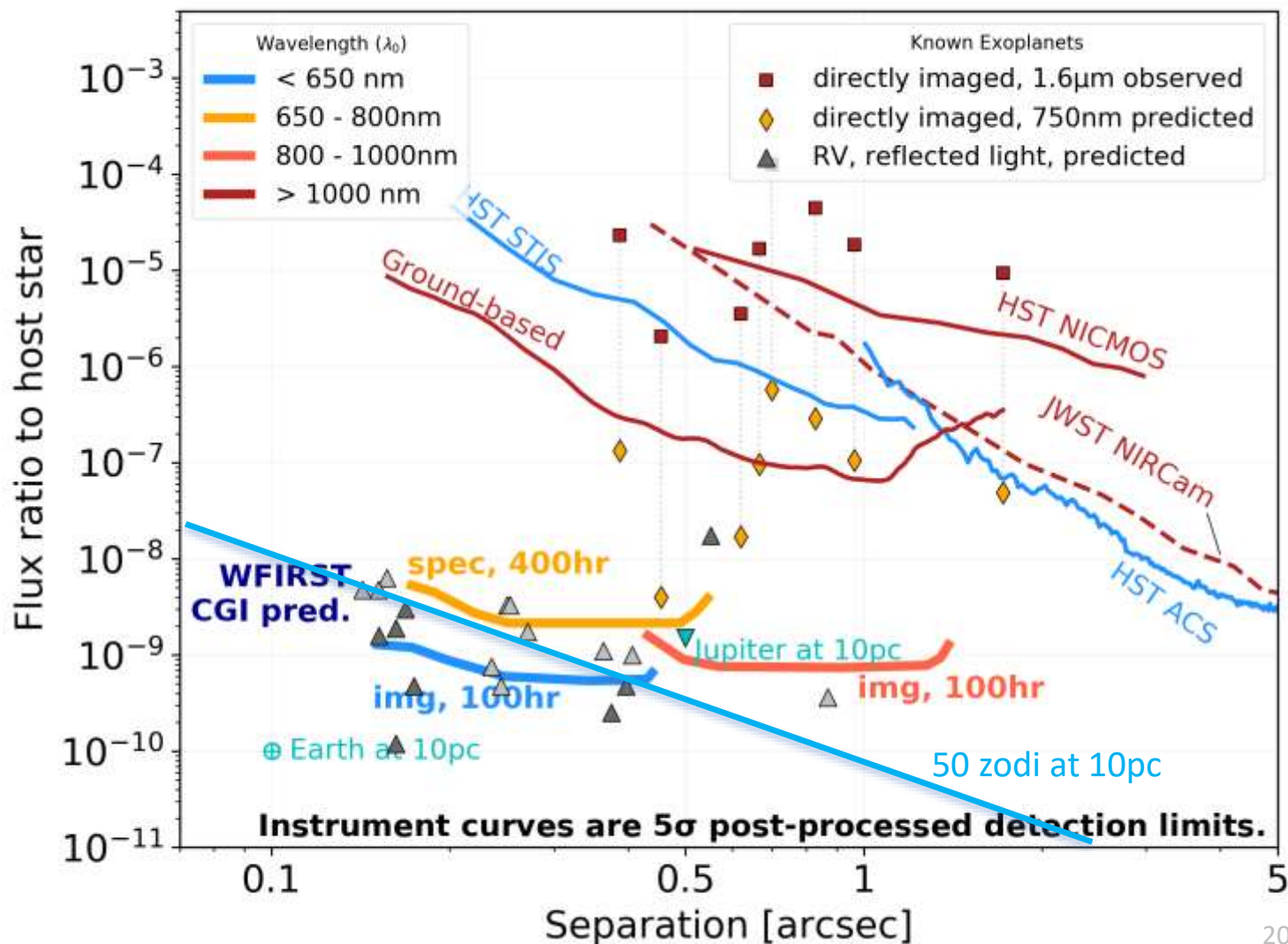
CGI can study young, self-luminous planets at new wavelengths



CGI can take the first reflected light images of true Jupiter analogs



CGI can study tenuous debris and exozodi disks at solar system scales



CGI predicted performance compares favorably to future missions' requirements

Parameter	CGI vs. Future missions <i>unobscured aperture: HabEx & LUVOIR B</i>
5σ Flux ratio at 3 λ/D (6 λ/D)	5$\cdot 10^{-9}$ (2$\cdot 10^{-9}$) * vs. 5$\cdot 10^{-11}$ ** WFIRST pupil is challenging

* model, typically without Model Uncertainty Factors (MUFs)

** NTE = not-to-exceed = requirement on max tolerable.

Future missions working group:

Bertrand Mennesson, Laurent Pueyo, Matt Bolcar, Bijan Nemati, Chris Stark, Stefan Martin, Aki Roberge²¹

CGI predicted performance compares favorably to future missions' requirements

Part 1

Parameter	CGI vs. Future missions <i>unobscured aperture: HabEx & LUVOIR B</i>
EMCCD	Comparable at V-band Bit better: dark current, clock-induced-charge Bit worse: QE at UV/red at 5 years (rad hard)
Pointing jitter control	Comparable CGI lab: ~ 0.35 mas RMS V=5 star, FM: 0.3 mas NTE**
Low order control (Z4-Z11)	~ 100x better ~ 10 pm RMS * vs ~ 1 nm NTE ** Challenging WFIRST pupil: trade low-order sensitivity for overall throughput

* model, typically without Model Uncertainty Factors (MUFs)

** NTE = not-to-exceed = requirement on max tolerable.

Future missions working group:

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CGI predicted performance compares favorably to future missions' requirements

Part 2

Parameter	CGI vs. Future missions <i>unobscured aperture: HabEx & LUVOIR B</i>
Wavefront error sources	Comparable Phase & “new physics” (amplitude & polarization)
High order drift ($\geq Z12$)	Comparable ($\sim 5\text{pm}$) CGI: 1σ prediction*, FM: NTE**
# of DMs	Same (2)
DM stroke resolution	$\sim 4\text{x worse}$ (7.5pm vs 2pm) Engineering problem, not physics problem
DM actuator count	48x48 vs 64x64

* model, typically without Model Uncertainty Factors (MUFs)

** NTE = not-to-exceed = requirement on max tolerable.

Critical to validate end-to-end system on sky, especially interaction with telescope

Implications of Tech Demo Designation



- CGI has technology, not science, requirements
- CGI cannot drive WFIRST mission cost or schedule
 - CGI may not be on the critical path
 - CGI is cost capped
 - No access to NASA or WFIRST Project reserves
 - CGI may not drive observatory design
 - CGI is now Class D

Plan to deliver a capable instrument on time and on budget

- Increased performance margin by removing L1 Baseline requirements, leaving only L1 Threshold requirement
 - $1 \cdot 10^{-7}$ flux ratio, $6 - 9 \lambda/D$, $\lambda \leq 600$ nm, bandwidth $\geq 10\%$, $V \leq 5$ star
- **CGI design has not changed as a result of L1 relaxation**
 - Re-design to Threshold would hurt both cost & schedule
- Improved cost & schedule robustness by identifying “off-ramps” and descopes that could be triggered if problems arise
 - May trade performance/risk for cost/schedule.
 - Assess with performance modeling tools.
- Granted CGI Project sole decision authority, unless the L1 threshold requirement is at risk
 - Key decisions advised by stakeholders

Summary

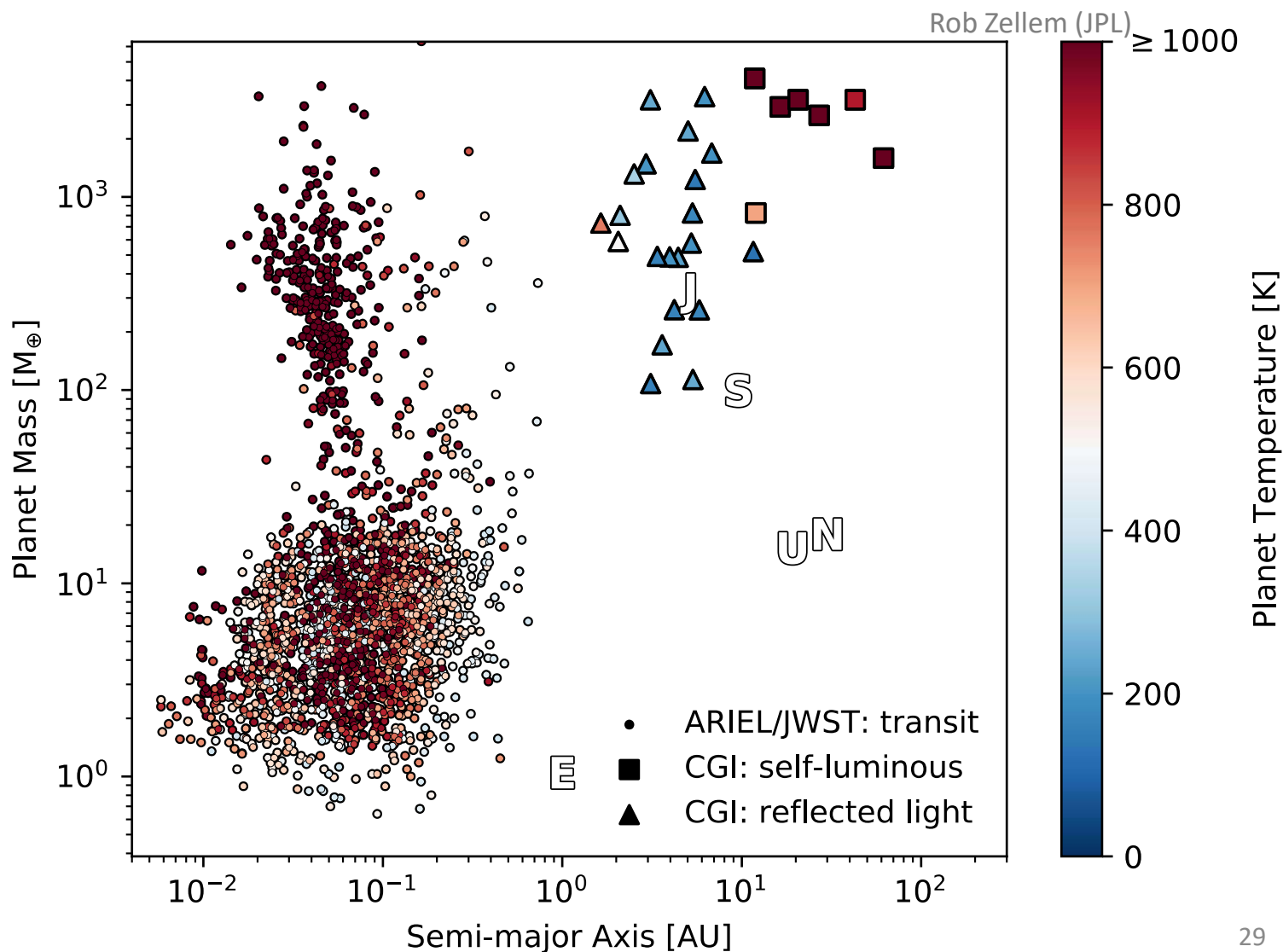


- First space-based coronagraph with active wavefront control
- Meaningful technology demonstrator
 - Lab & models are compelling, but need system-level on-sky test
 - Comparable to future missions' needs:
 - low order control, high-order stability, “new physics,” EMCCD noise
 - Improvement over SOTA, but more work needed:
 - high order wavefront control, DMs, EMCCD lifetime & UV/red sensitivity
- Capable of interesting science
 - Jupiter analogs in reflected light; young exoplanets at new λ
 - Tenuous debris/exozodi disks; perhaps protoplanetary systems
- Approved to begin implementation with a plan to stay on time and on budget

Questions?

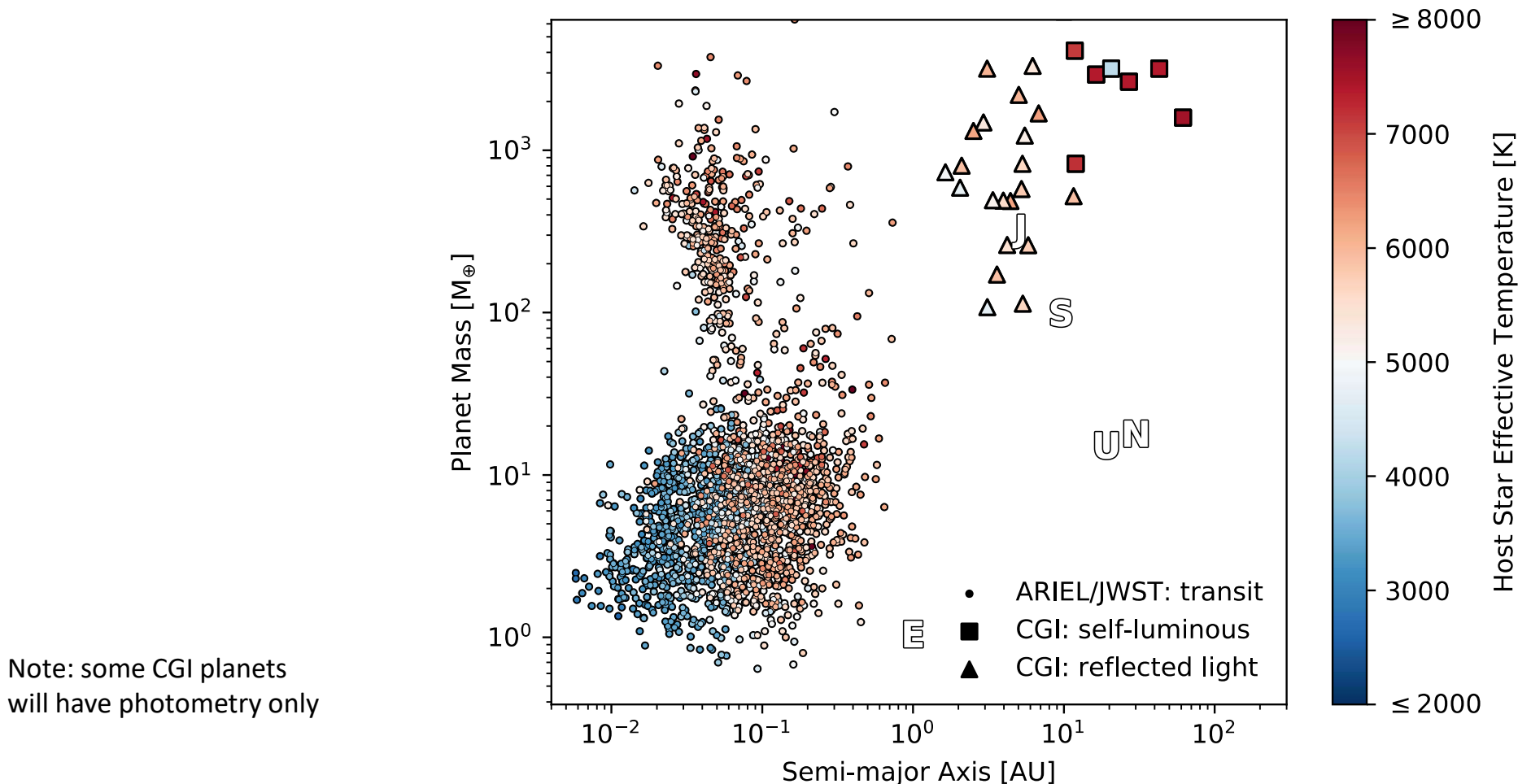
Intro backups

Transit spectroscopy probes a different class of planets

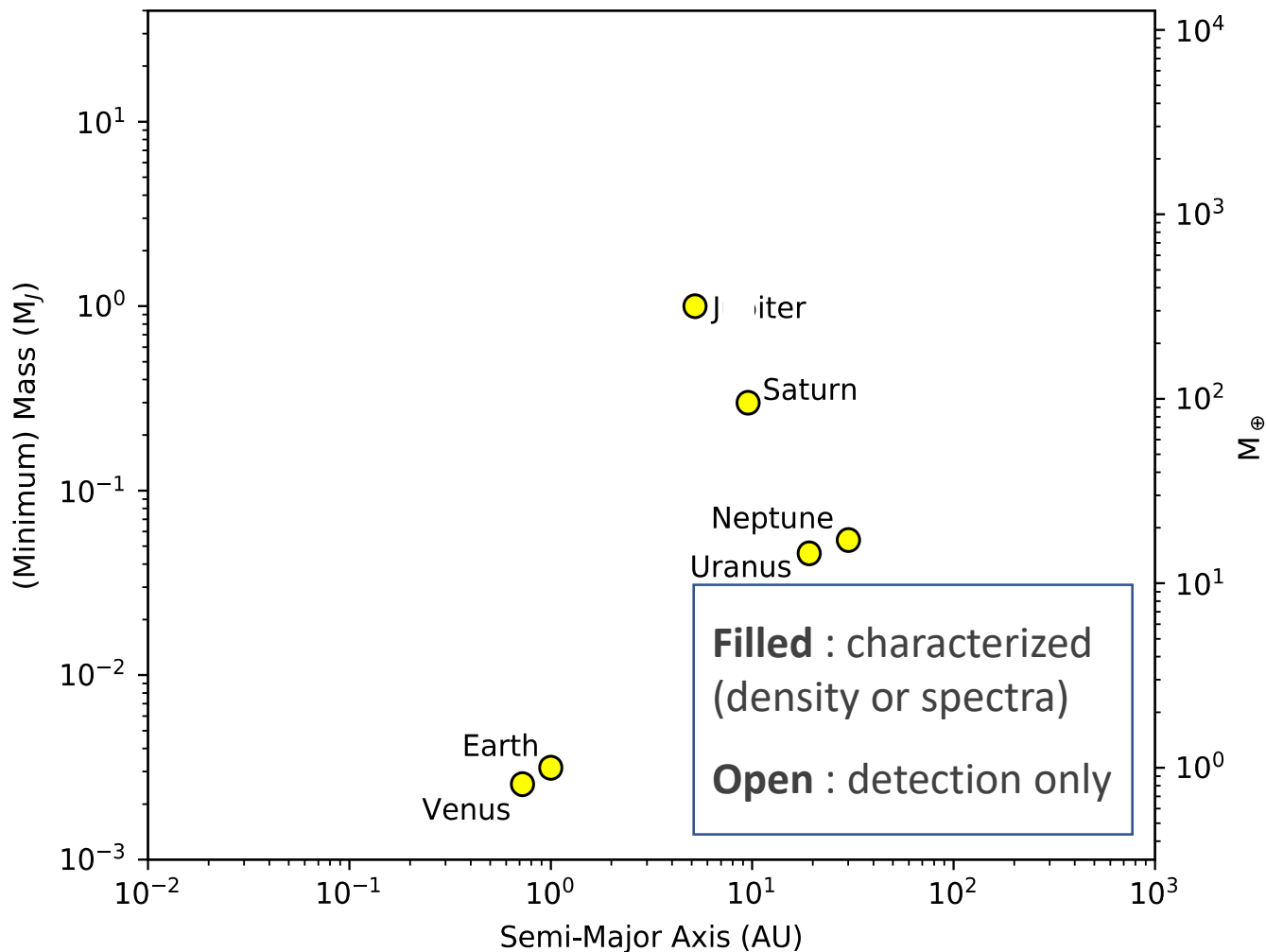


Note:
most CGI mature
planets will only
have photometry

Transit spectroscopy probes different class of planets

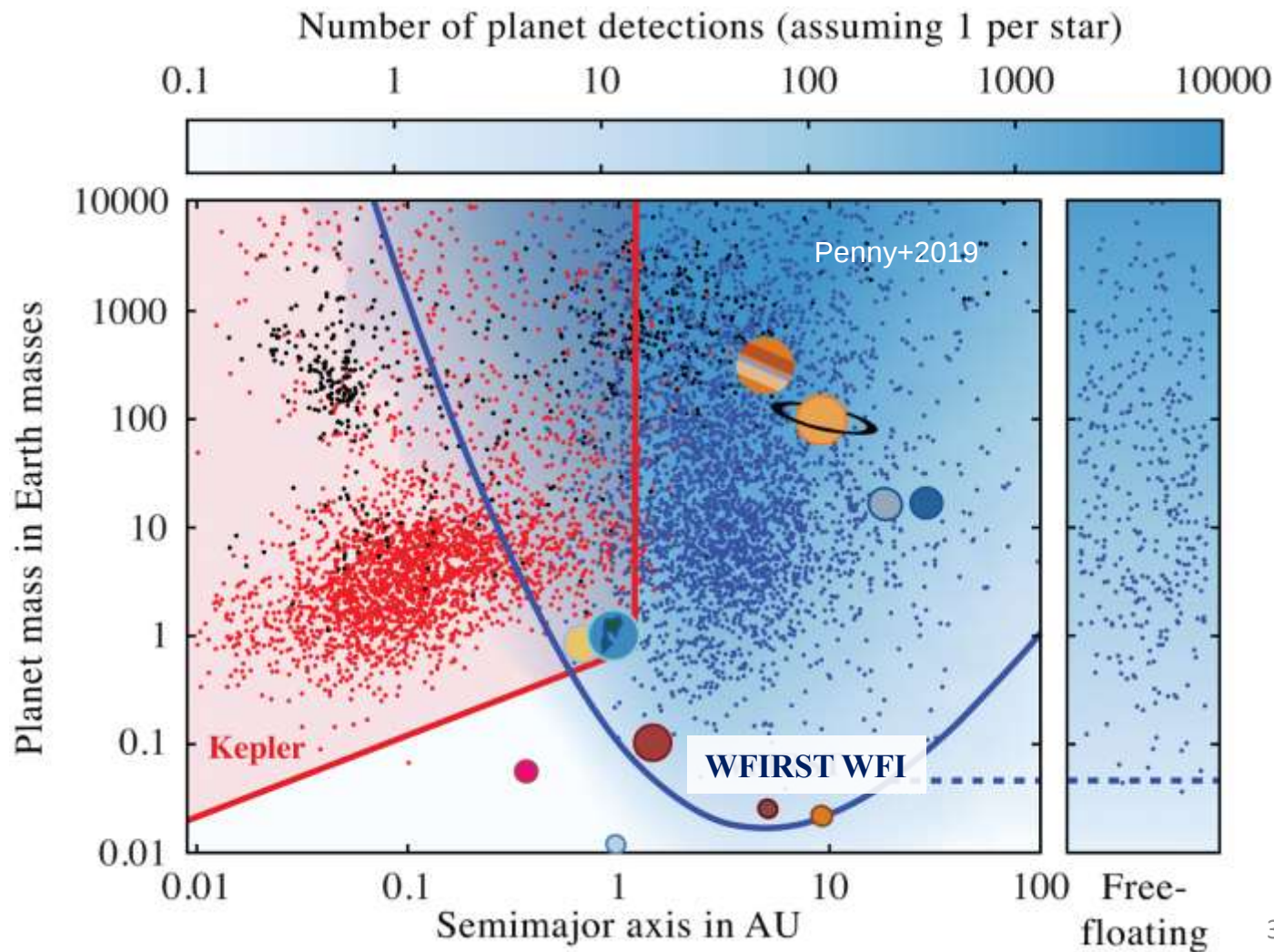


Only a small fraction of known exoplanets have been characterized



WFIRST *Wide Field Instrument*

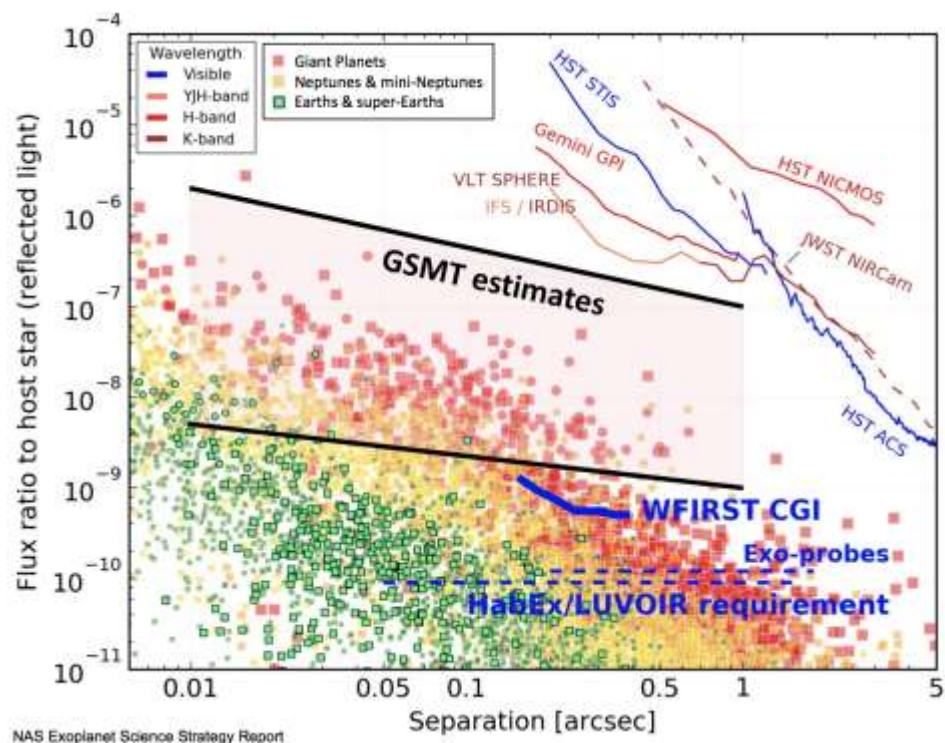
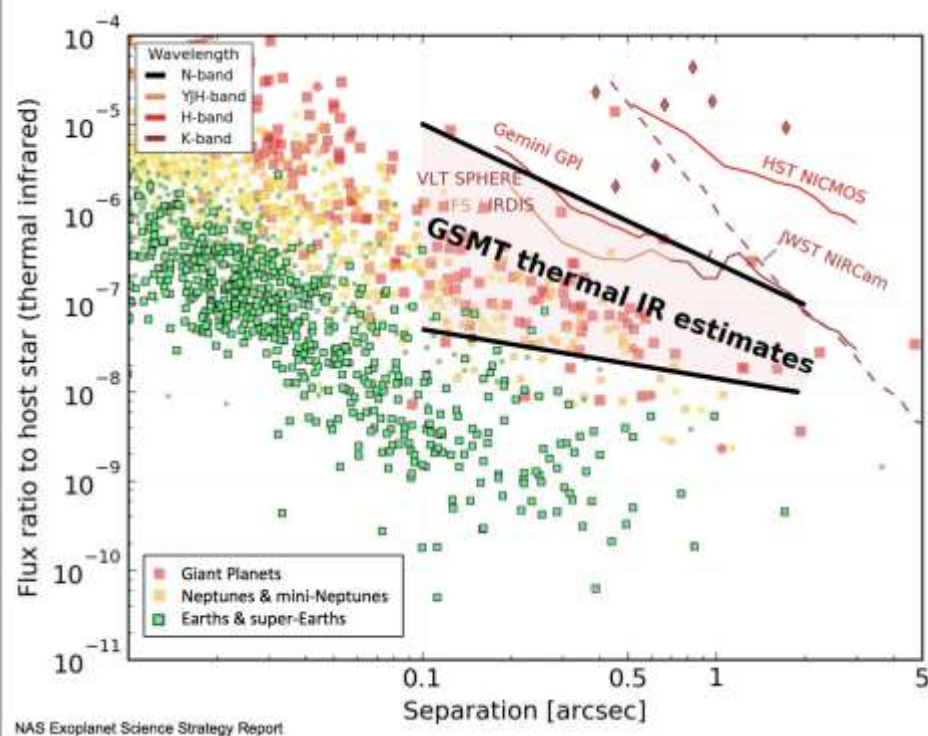
microlensing will discover 1000s of planets,
but they cannot be characterized



ELTs and space missions are complementary

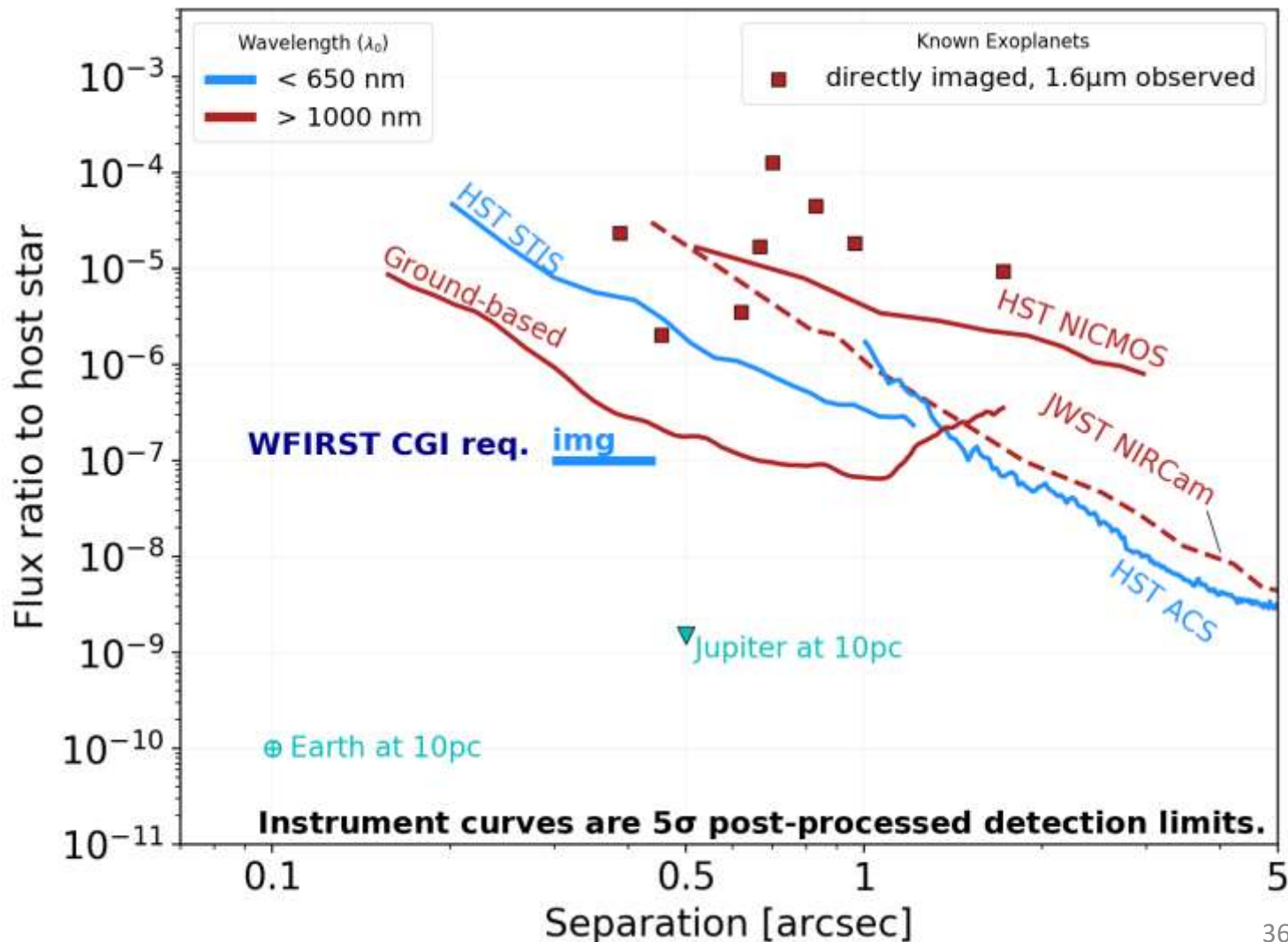
- CGI: Jupiter analogs around Sun-like stars
 - Visible, modest working angle, intermediate flux ratio
- ELTs: small, temperate planets around cool stars
 - Infrared, small working angle, shallower flux ratio
- Future space missions: Earth analogs around Sun-like stars
 - UV (ozone) and visible, intermediate working angle, deepest flux ratio

Synergies between Ground and Space (NAS ESS report)

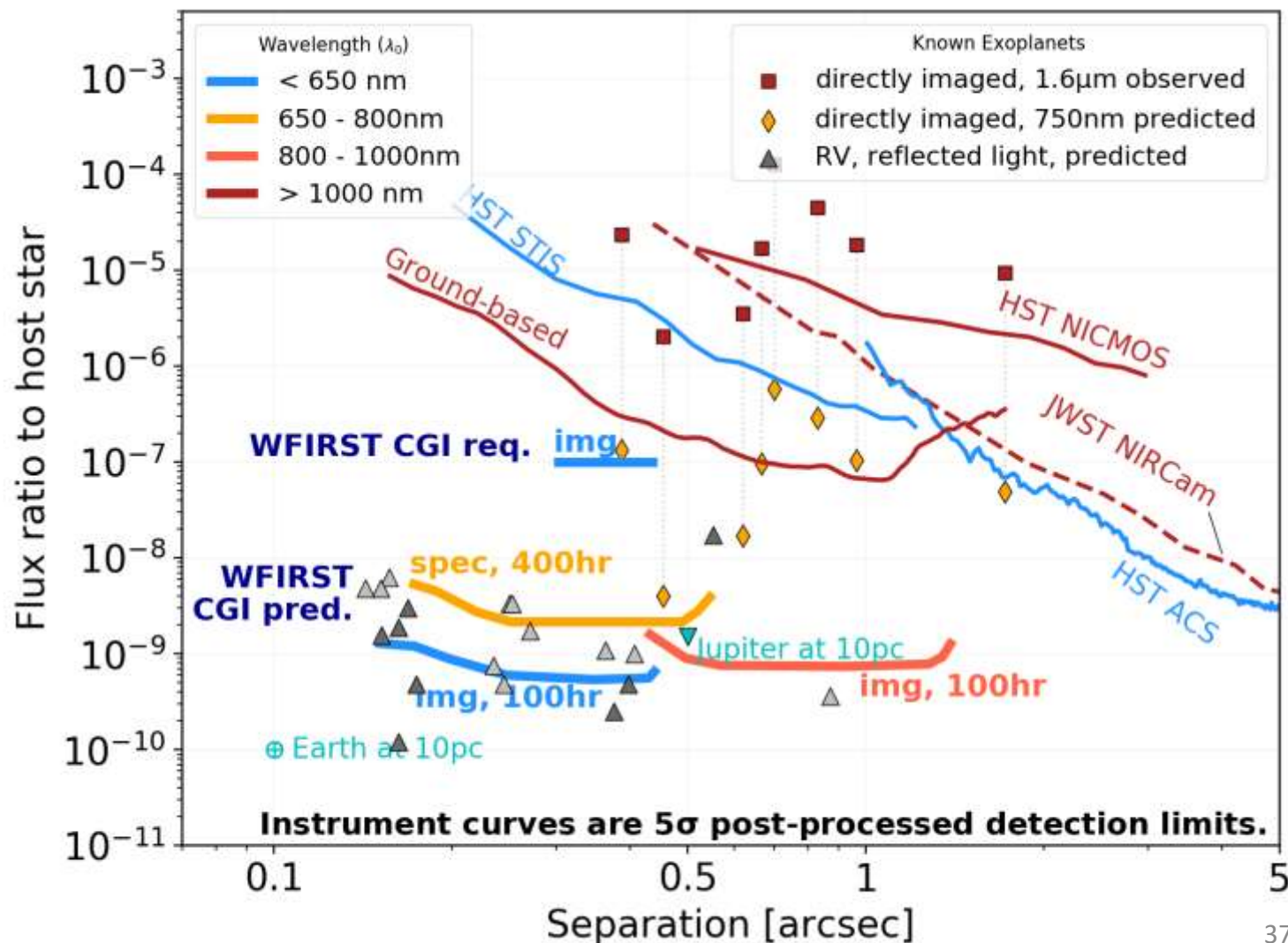


Technology / Instrument

CGI minimum performance requirement



L1 Threshold
requirement
(trigger cancellation
review if do not meet)



Timeline: pre-launch



- Preliminary Design Review (Q3/Q4 2019)
- Critical Design Review (Q1 2021)
- Deliver CGI for Payload I&T (Q3 2023)
- Launch (Q4 2025)

Potential enhancements

- Improve confidence in **instrument lifetime** with additional component testing and ground support equipment

CGI Observing Modes

λ_{center} (nm)	BW	Mode	FOV radius	Polarimetry?
575	10%	Imager	0.14" – 0.45"	Y
730	15%	Slit + R~50 Prism	0.18" – 0.55"	-
825	10%	Imager	0.45" – 1.4"	Y
630	15%	Slit + R~50 Prism	0.17" – 0.5"	Y
H α	1%	Imager	0.17" – 0.5"	Y
575	10%	Imager	0.35" - 1"	Y
825	10%	Imager	0.2" - 0.65"	Y

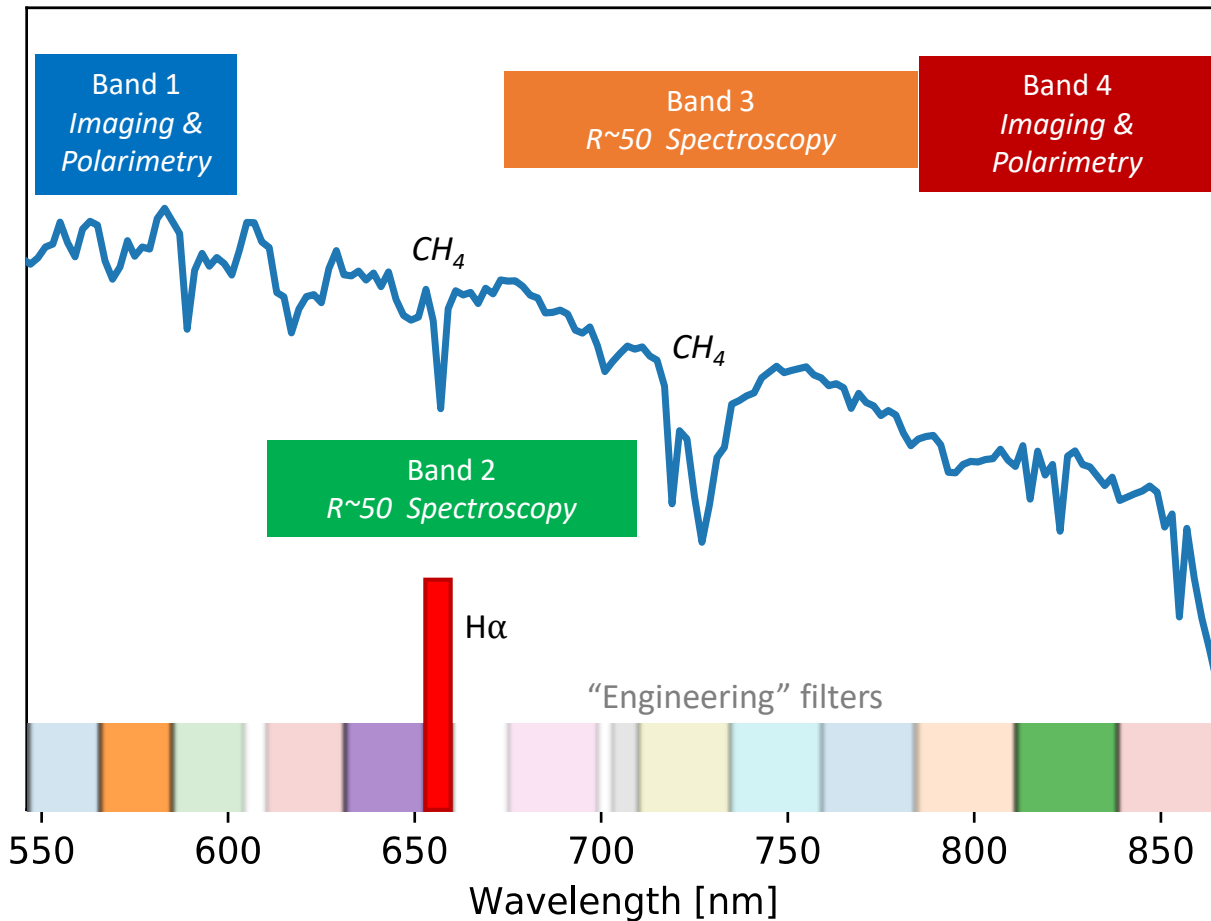
Additional narrow *sub-bands* (2.5-3.5%) installed

Three modes will be fully tested prior to launch.

Additional modes installed but not fully tested before launch

Potential enhancement
more pre-flight testing

WFIRST CGI Passbands



Three official modes
will be fully tested prior
to launch.

Additional modes will be
installed but not fully
tested before launch

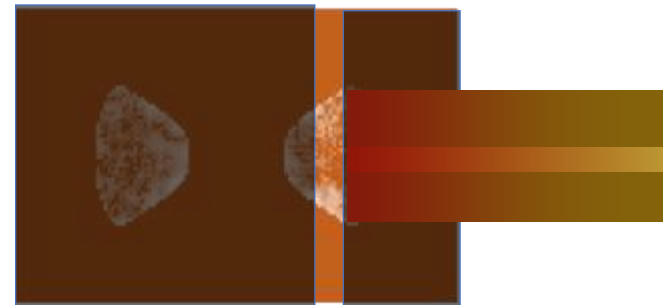
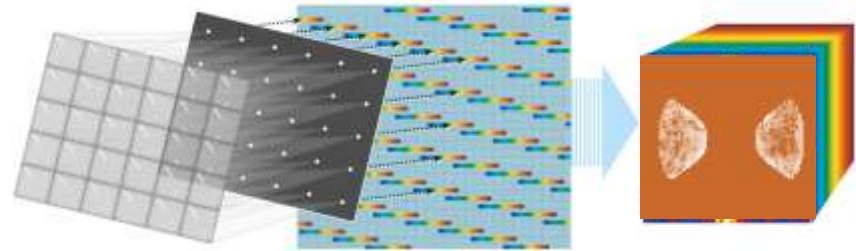
Potential enhancement
Add'l unofficial mode
combinations
and/or
more pre-flight testing

2019 spectroscopy change: IFS => slit+prism



Tyler Groff (GSFC)
Neil Zimmerman (GSFC)

- **IFS:** $R=50$. Sampled across FOV.
- **Slit:** $R \sim 35-70$. Sampled in slit only.
 - Fewer optics => higher throughput
- **CGI science capabilities largely unchanged**
 - Exoplanets: comparable
 - Not expecting to observe multi-planet systems
 - Disks: Not planned
 - More time consuming, but was never planned, because no spectral features of interest.
 - Operations: Increased alignment/calibration complexity, but solvable



EMCCD Degradation



- **Concern: How much does detector performance degrade due to radiation damage over the 5 year mission?**
- Custom chip design significantly mitigates radiation susceptibility vs. commercial version.
 - Performance is being validated in lab with radiation source.
- Detector performance is more important when sensitivity is photon- or detector noise-limited vs speckle-limited (ie: spectroscopy).
 - For spectroscopy mode, expect only a small ($<10\%$ *relative* decrease) in effective quantum efficiency between 0yrs and 5yrs.
 - assumes long (~ 120 s) exposure times, consistent with spec observations. During long exposures, dark current helps to fill traps, reducing their effect.
 - The relative reduction in QE would be larger when shorter exposures are used (ie: in speckle-limited imaging mode). But in this case, detector noise and traps are not the limiting factor.
 - Dark current could increase by ~ 3 x over 5yr. (still <5 e-/px/hr)

For more details, see Patrick Morrissey's 2019 SPIE presentation:

<https://www.spiedigitallibrary.org/conference-proceedings-of-spie/11117/111170J/Flight-photon-counting-EMCCDs-for-the-WFIRST-coronagraph-Conference-Presentation/10.1117/12.2529758.full>

Science Summary Slides

Potential CGI Exoplanet Science Contribution

Question	WFIRST can	During Tech. Demo Phase	Additional Science Phase (may not do all cases)
What are the cloud properties of young massive planets? How inflated are these planets?	Fill out SEDs with photometry and spectroscopy at $\geq 600\text{nm}$	1-2 test cases	Additional filters and/or more known planets
Are cold Jupiter analogs cloudy or clear?	Measure albedo at short wavelengths	1-2 planets	Additional filters and/or up to ~ 10 more planets
Are Jupiter analogs metal rich?	Distinguish 5x vs 30x Solar CH_4	1 planet * 730nm spec. only	+1 planet <i>OR</i> improve 1 st planet w/ 660nm spec or better SNR in 730 spec.
Are there Neptune-like planets orbiting nearby stars?	Survey nearby systems, informed by Gaia & RV	No	5-10 best systems

* Clear atmosphere planets may be too dim for spectroscopy

Potential CGI Disk Science Contributions

Question	WFIRST can	During Tech. Demo Phase	Additional Science Phase (may not do all cases)
Where does circumstellar material come from and how is it transported?	Map morphology of disks in the transport dominated density regime.	2-3 disks	Additional disks with a variety of properties
What is the composition of planetary dust in the inner regions of debris disks?	Map color, degree of forward scattering, and the degree of polarization.	1-2 disks	Additional disks with a variety of properties
How bright is exozodiacal dust in scattered light? Will it affect exo-Earth detection with future missions?	Probe low surface density disks in habitable zone of nearby stars	Opportunistic (as part of known exoplanet observation)	Survey best 25-50 potential exo-Earth targets for future missions
What are the accretion properties of low-mass planets in formation?	Measure H-alpha at high contrast	0 – 1 test observations	Observe transition disks with gaps in CGI FOV

Science Yield vs Instrument Performance

	10^{-9}	3×10^{-9}	10^{-8}	10^{-7}
Jupiter analog spectra	Some	A few	No	No
Jupiter analog Images	Yes	Yes	Possibly	No
Young GP optical spectra	Yes	Yes	Yes	Few
Young GP optical images	Yes	Yes	Yes	Some
Exo-Zodi Disks optical images	~2 zodi	~5 zodi	~15 zodi	~100 zodi

Exoplanet backups

Today

$\geq 1 \mu\text{m}$

Self-luminous, hot, super-Jupiters

WFIRST/CGI

550 – 880 nm

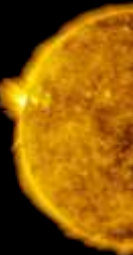
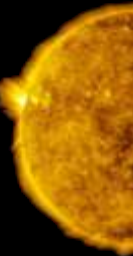
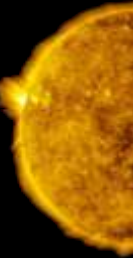
Reflected light Jupiter analogs

Self-luminous planets in visible light

Future Missions

0.3 – 1 μm

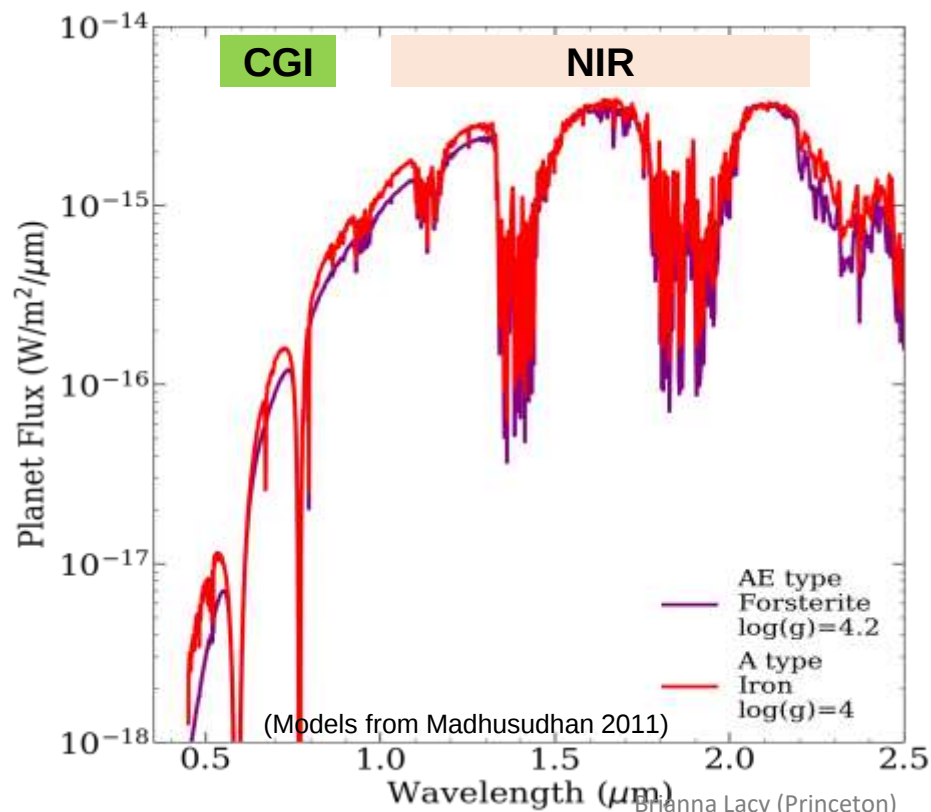
Earth-like, potentially habitable, planets



Young, self-luminous massive planets.

CGI complements ground-based NIR

- **Q:** What are the cloud properties of young massive planets? How inflated are they? Are they metal rich?
- **CGI can:** Fill out SED with broadband photometry and spectroscopy
- **During TDP:** 1-2 systems
- **Beyond TDP:** Additional bandpasses and/or survey more known planets



Brianna Lacy (Princeton)

First reflected light images of a mature Jupiter analog



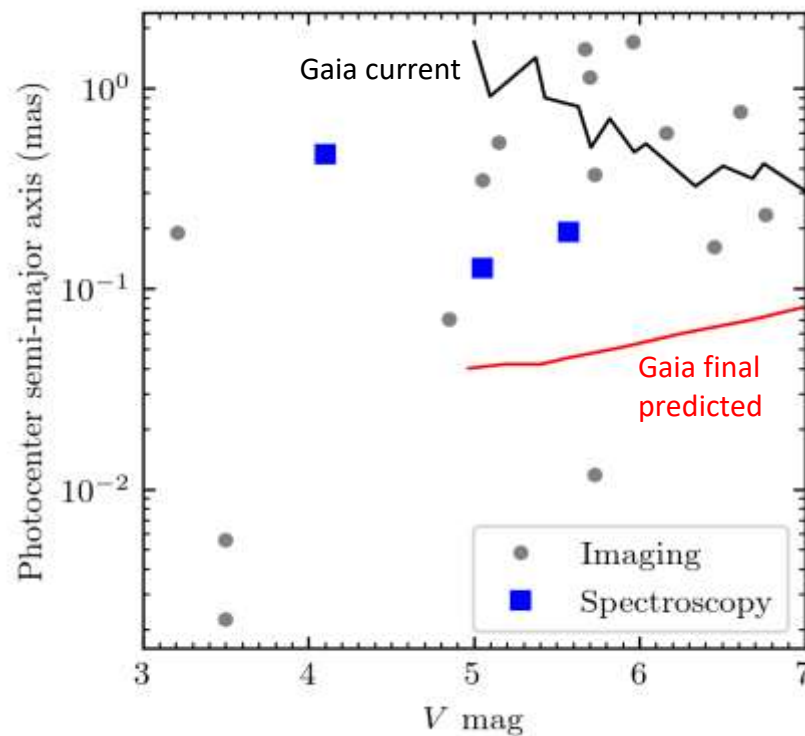
Rob De Rosa (Stanford)
Julien Girard (STScI)
Stephen Kane (UCR)
Eric Nielsen (Stanford)
Maggie Turnbull (SETI)
Neil Zimmerman (GSFC)

...



- limited observing time
⇒ target known RV Jupiter analogs
- **Q:** What is the mass of the planet?
- **CGI can:**
 - constrain inclination with 2-3 imaging epochs
 - combine with Gaia for better constraints

Astrometric signal for CGI RV targets



First reflected light images of a mature Jupiter analog



- **Q:** Are cold Jupiter analogs cloudy or clear?
- **CGI can:** Measure albedo at short wavelengths
- **During TDP:** 1-2 planets
- **Beyond TDP:** Additional narrowbands and/or survey more known planets

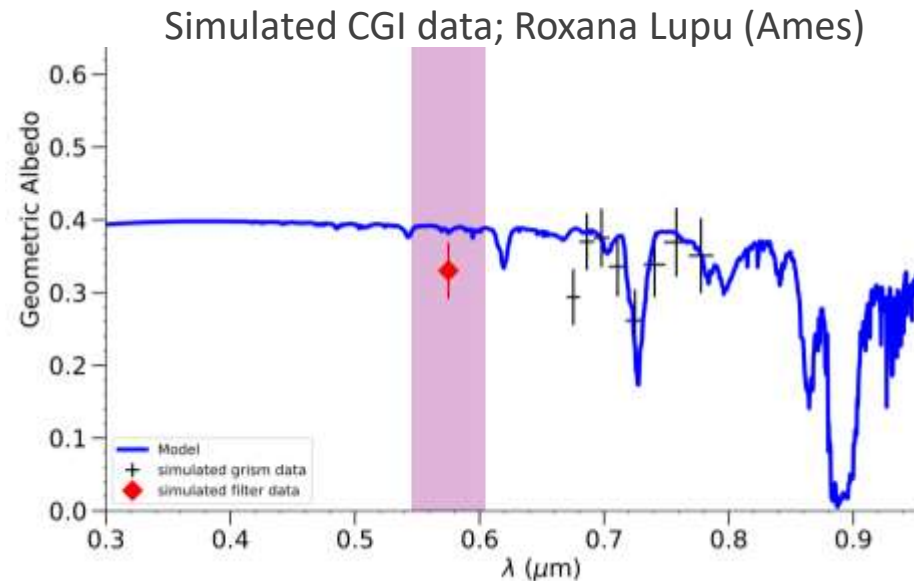
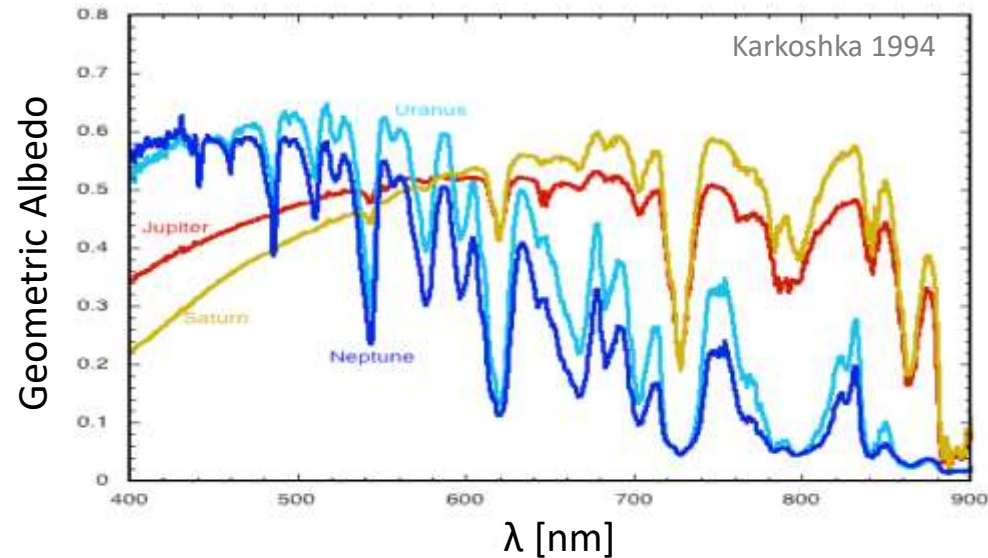


Characterization of a mature Jupiter analog



Increase confidence that we can detect molecular features in faint, high-contrast, reflected light spectra before we attempt exo-Earths

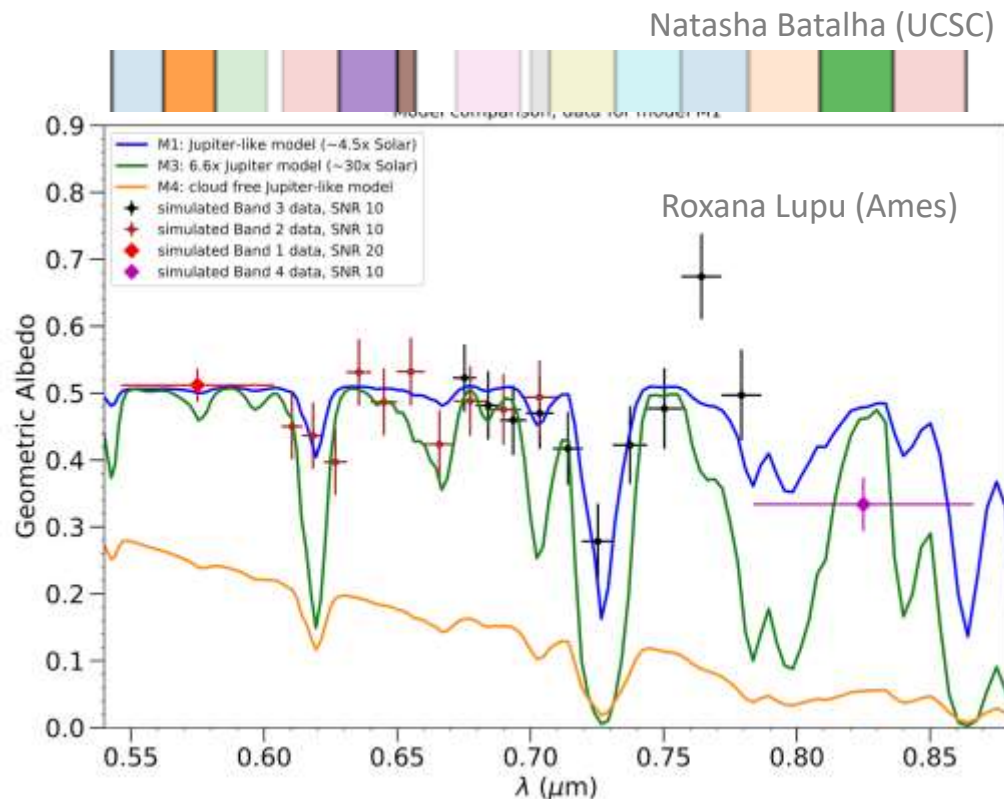
- **Q:** Are Jupiter analogs metal rich?
- **CGI can:** Coarsely constrain metallicity (5x vs. 30x Solar) if cloudy (high albedo)
- **During TDP:** 1 planet with 730nm spectroscopy



Characterization of a mature Jupiter analog

Increase confidence that we can detect molecular features in faint, high-contrast, reflected light spectra before we attempt exo-Earths

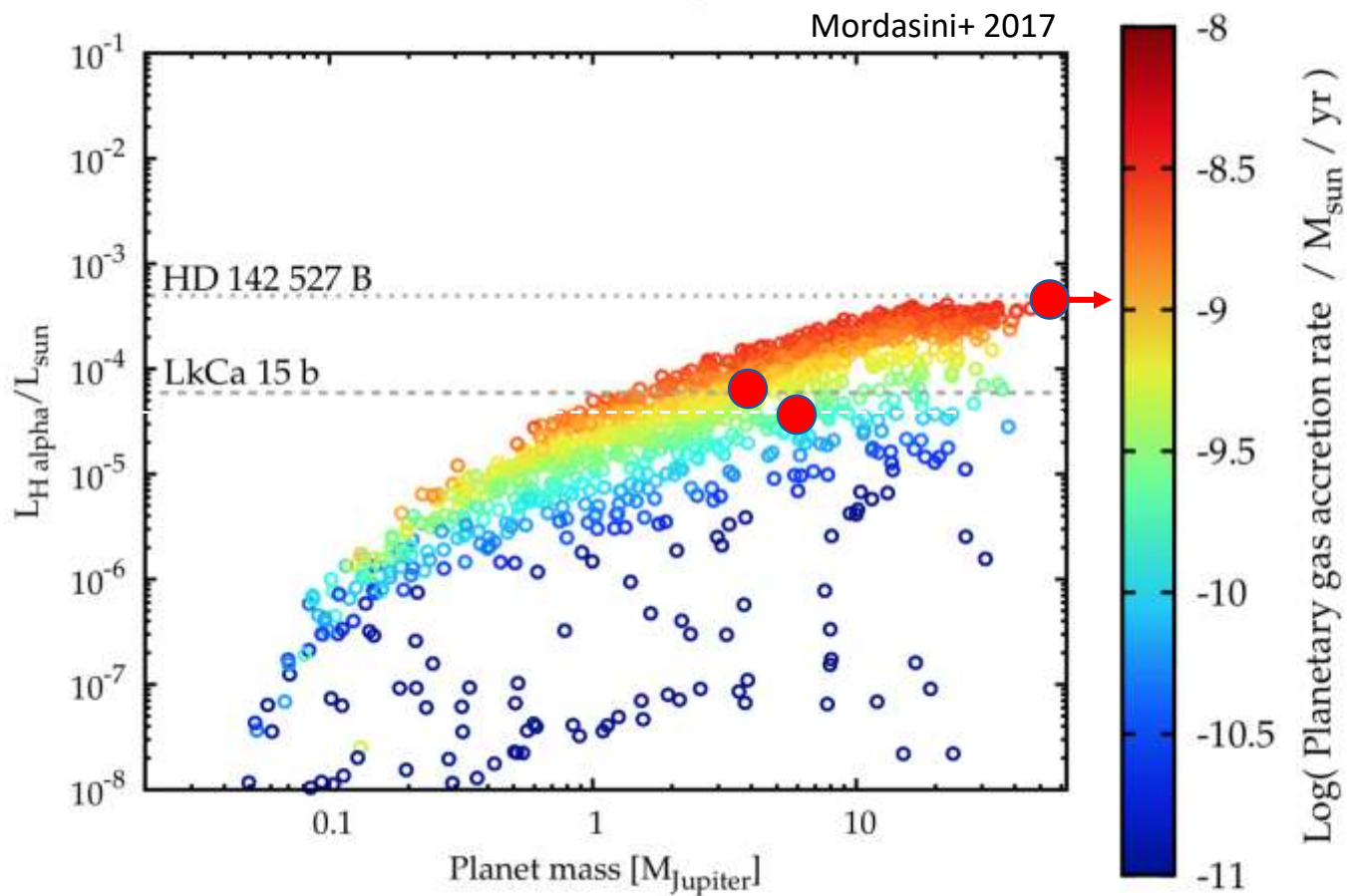
- **Q:** Are Jupiter analogs metal rich?
- **CGI can:** Coarsely constrain metallicity (5x vs. 30x Solar) if cloudy (high albedo)
- **During TDP:** 1 planet with 730nm spectroscopy
- **Beyond TDP:**
 - +1 planet
 - *OR* improve SNR of 1st planet
 - *OR* obtain narrowband photometry or 660nm spectroscopy of 1st planet.



Caveat! 660nm spectroscopy and 825nm narrow field imaging are NOT officially supported observing modes

H-alpha Imaging of Protoplanets

High-contrast H-alpha measurements will test these predicted core accretion luminosities.

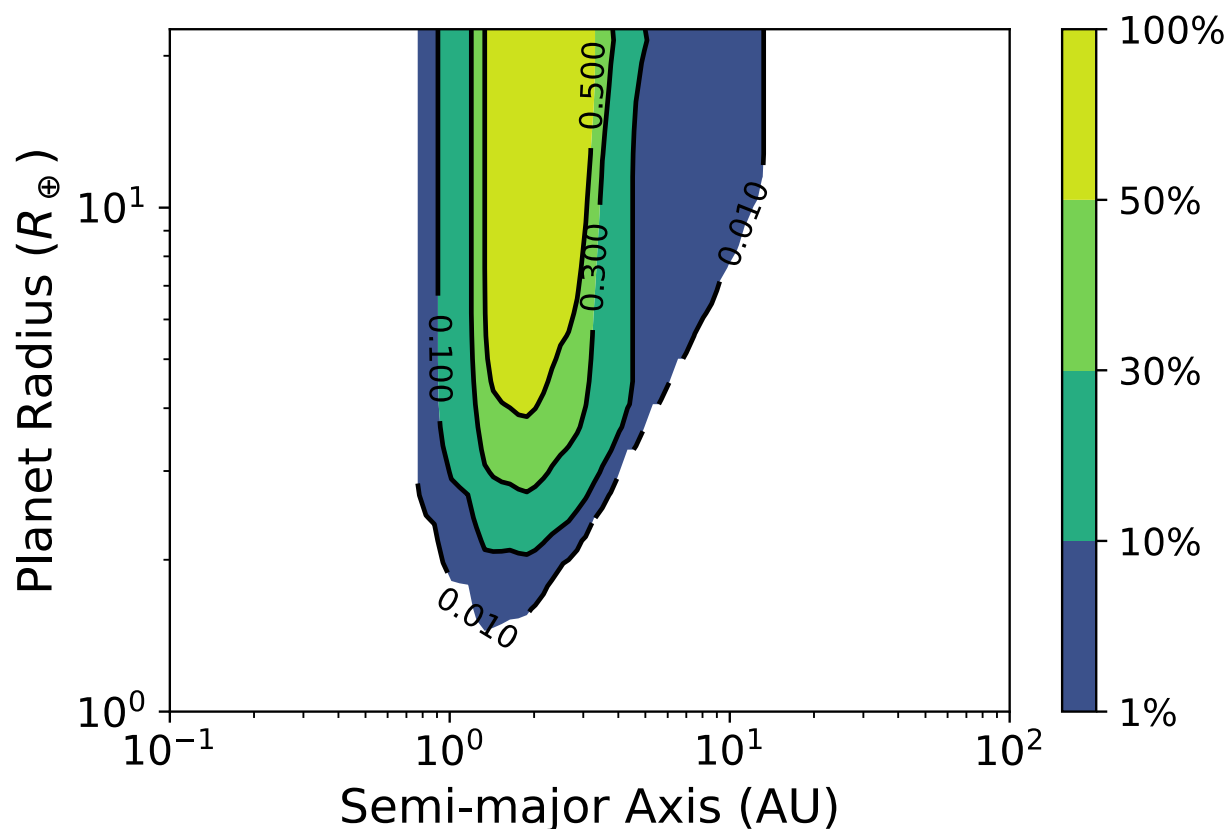


Beyond TDP: Search for small planets

CGI completeness
for 10 best targets

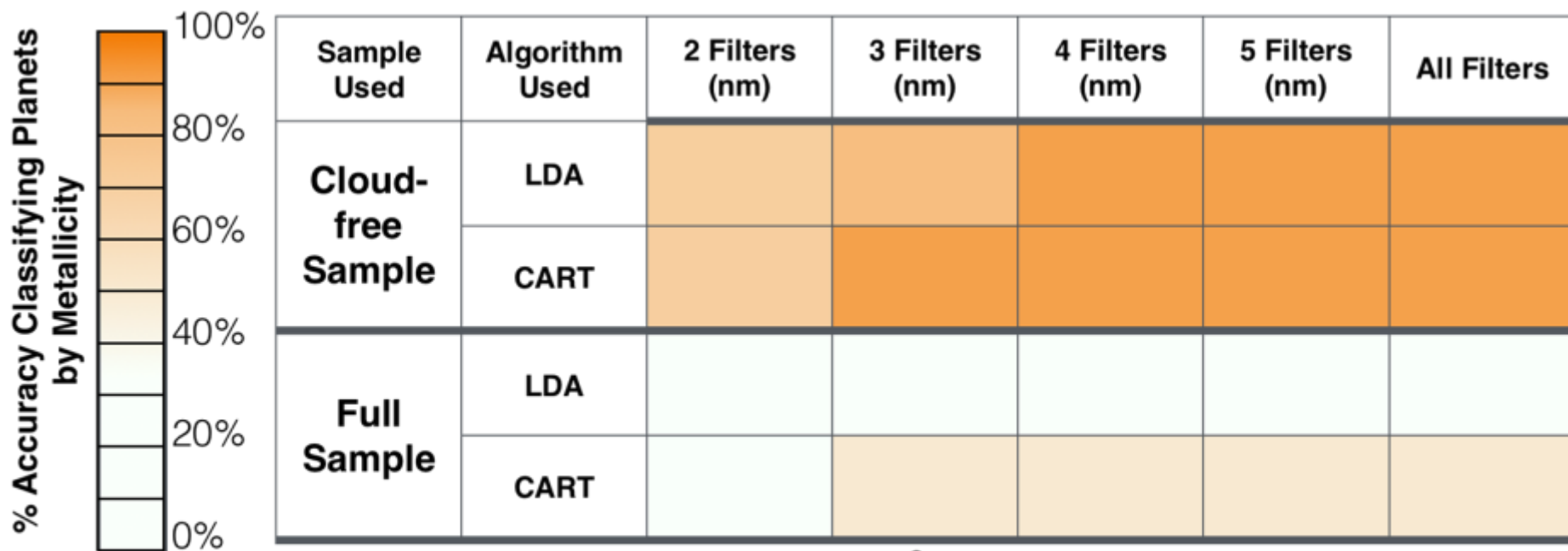
- ~100hr imaging
per target

Informed by Gaia
and RV limits



Beyond TDP: Multi-band photometric survey of reflected light planets.

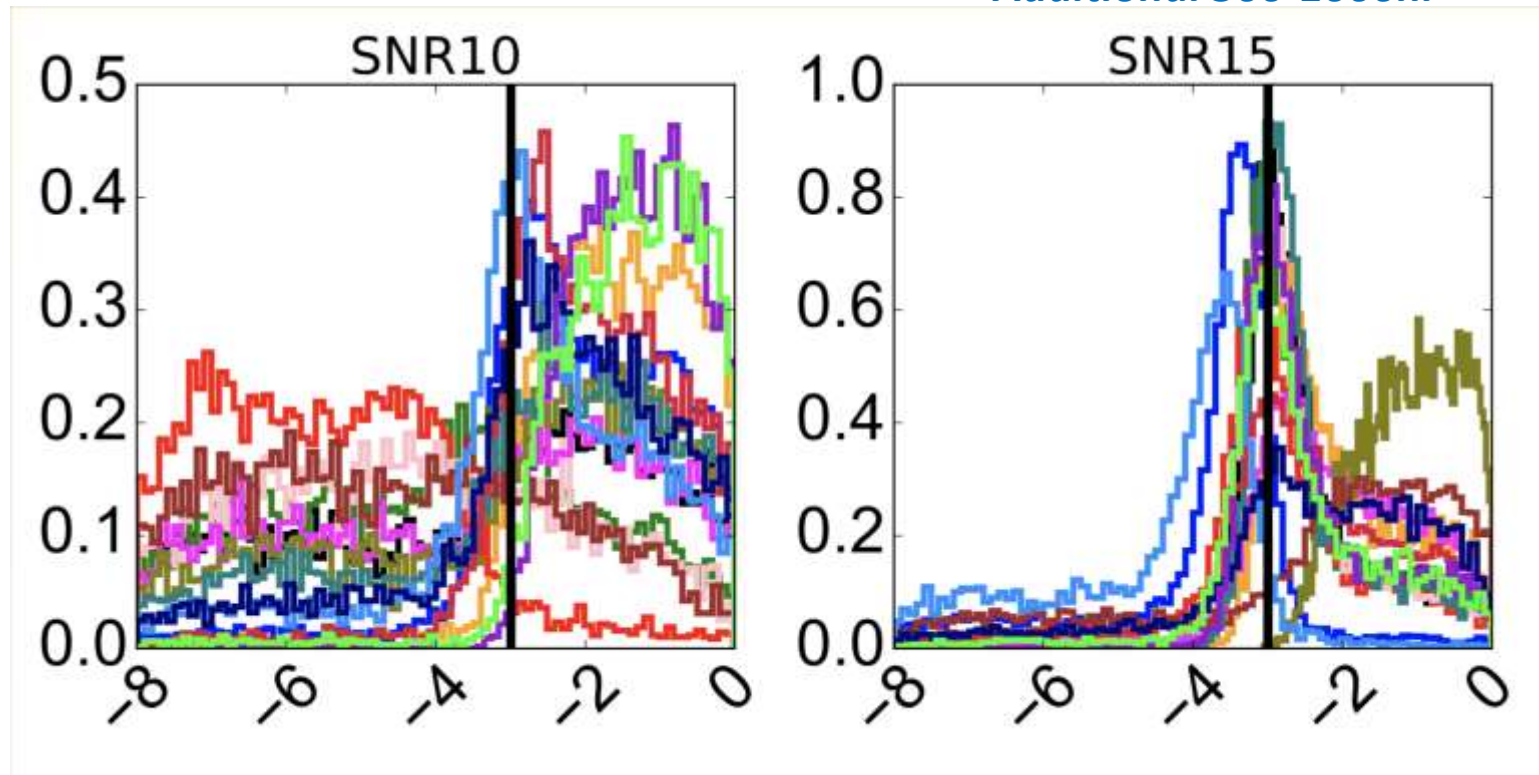
Metallicity?



Addition of 3rd filter gives largest gain in success rate for classification

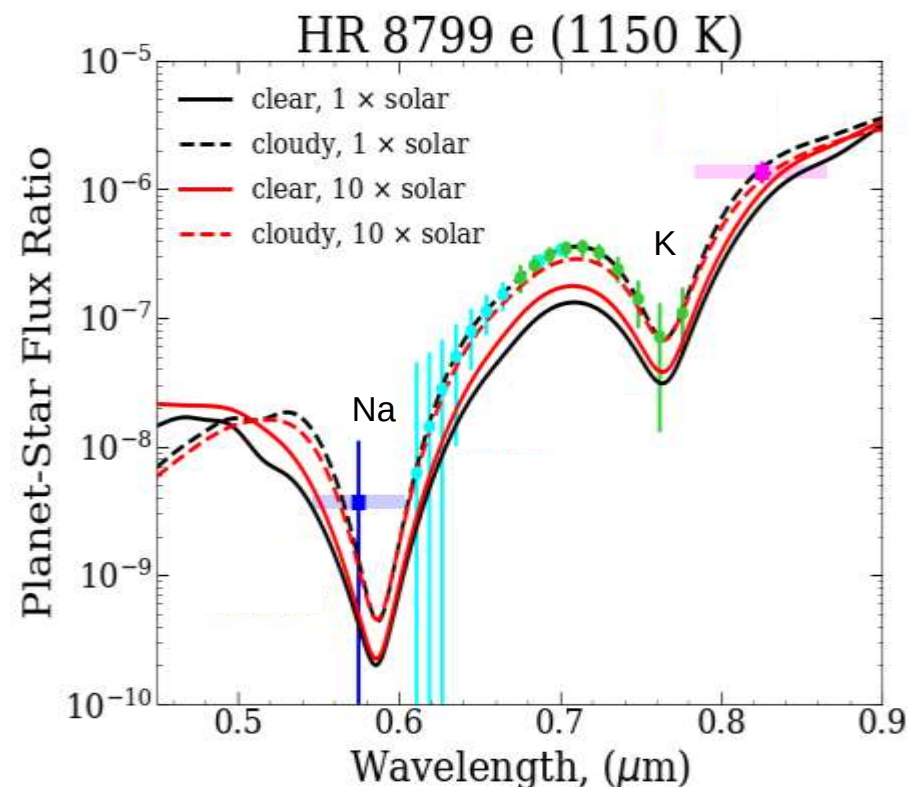
Beyond TDP:
improve SNR of reflected light planet
spectrum for CH₄ abundance

Additional 500-1000hr



Beyond TDP: Na and K in self-luminous planets

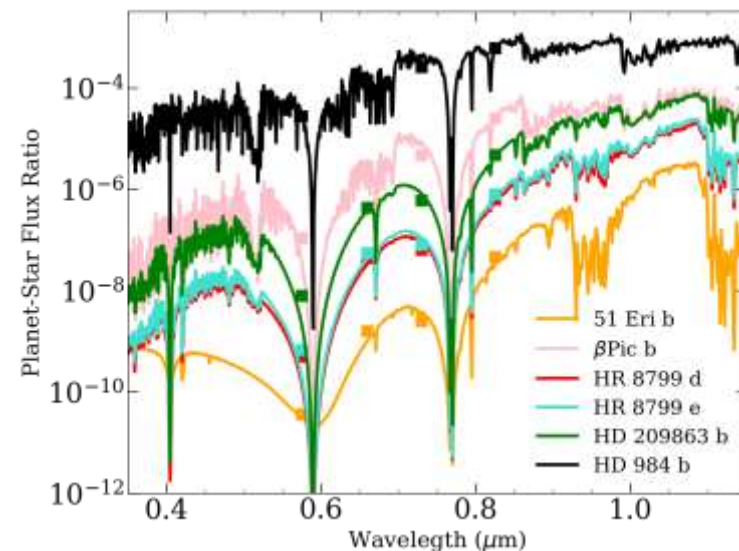
- Detect Na and K
- combine with NIR to help constrain:
 - the species, spatial extent, and particles sizes of condensates
 - the planet's effective temperature, surface gravity, and radius
 - the atmospheric metallicity



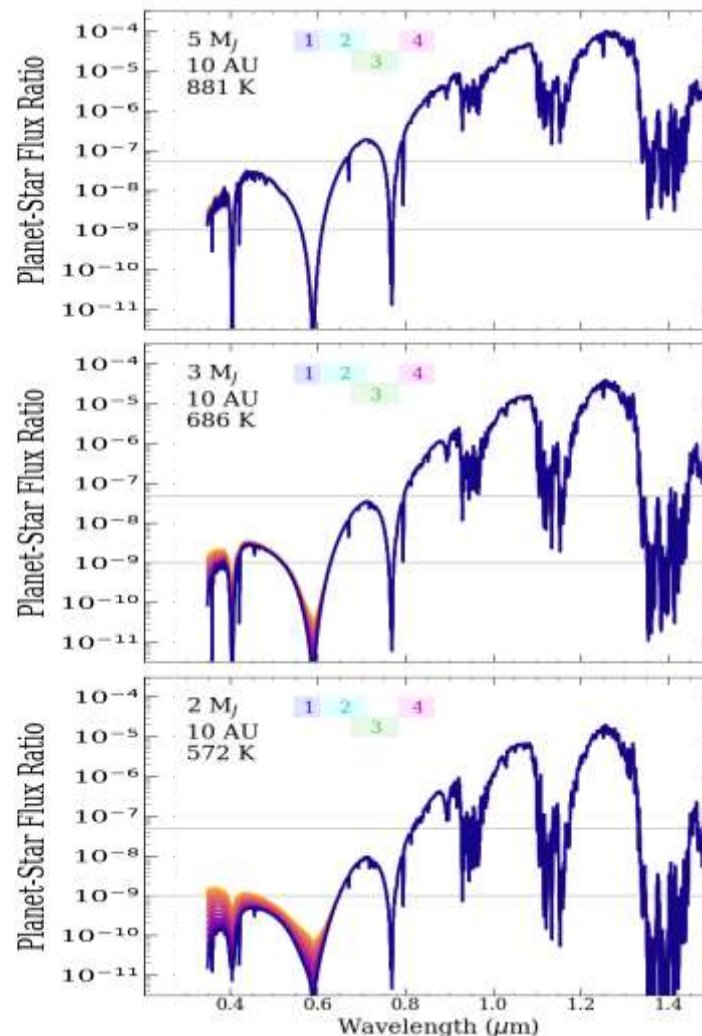
Self-luminous planet flux ratio in CGI bandpasses

Brianna Lacy (Princeton)

Object	Band 1	Band 2	Band 3	Band 3
51 Eri b	3.7E-11	1.6E-09	2.5E-09	4.6E-08
*Beta Pic b	1.1E-07	2.9E-06	4.7E-06	2.5E-05
HR 8799 d	5E-10	4.4E-08	6.4E-08	6.3E-07
HR 8799 e (cloudy)	6.8E-10 (2.7E-09)	5.6E-08 (1.6E-07)	8.1E-08 (2.1E-07)	7.9E-07 (1.4E-06)
HD 206893	7.9E-9	4.4E-07	6.1E-07	4.7E-06
HD 984 b	2.7E-05	1.4E-04	2.6E-04	6.1E-04



Reflected light is negligible for self-luminous planets



Larger mass
+ Warmer

Lower mass +
Cooler

Colors in the plot indicate phase angle.

CGI-Gaia synergies



Rob De Rosa (Stanford)

- CGI-Gaia synergies:
 - Prior to launch:
 - Constraints on inclination -> better constraint on mass (i.e. differentiate planet/BD) to refine target selection for reference mission.
 - During demonstration phase:
 - Help reduce inclination degeneracy with a single epoch of CGI relative astrometry to further constrain mass. Most useful for observations near line of nodes where single CGI epoch tells you nothing about inclination (see work by Eric Nielsen).
 - Potential GO program:
 - Identify promising blind search targets based on astrometric signature of massive orbiting companion (joint RV+astrometry constraints on mass/sma of companion).

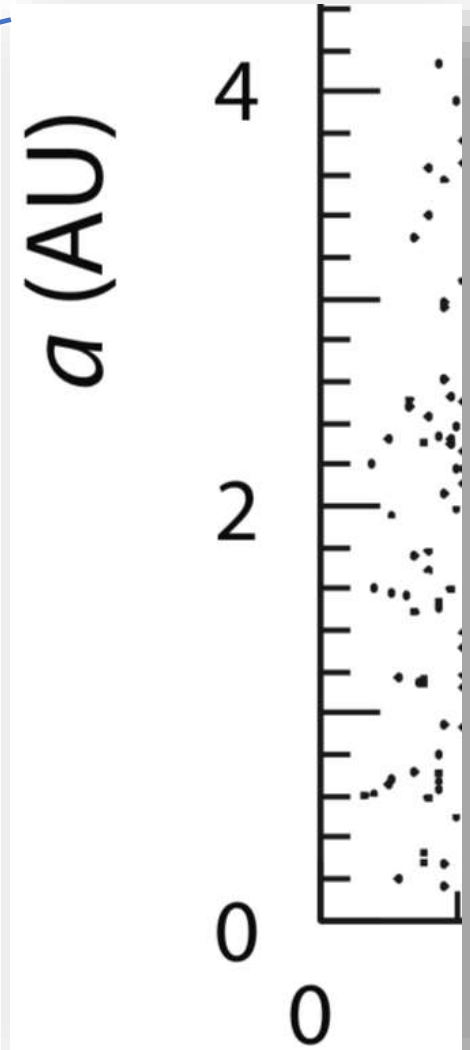
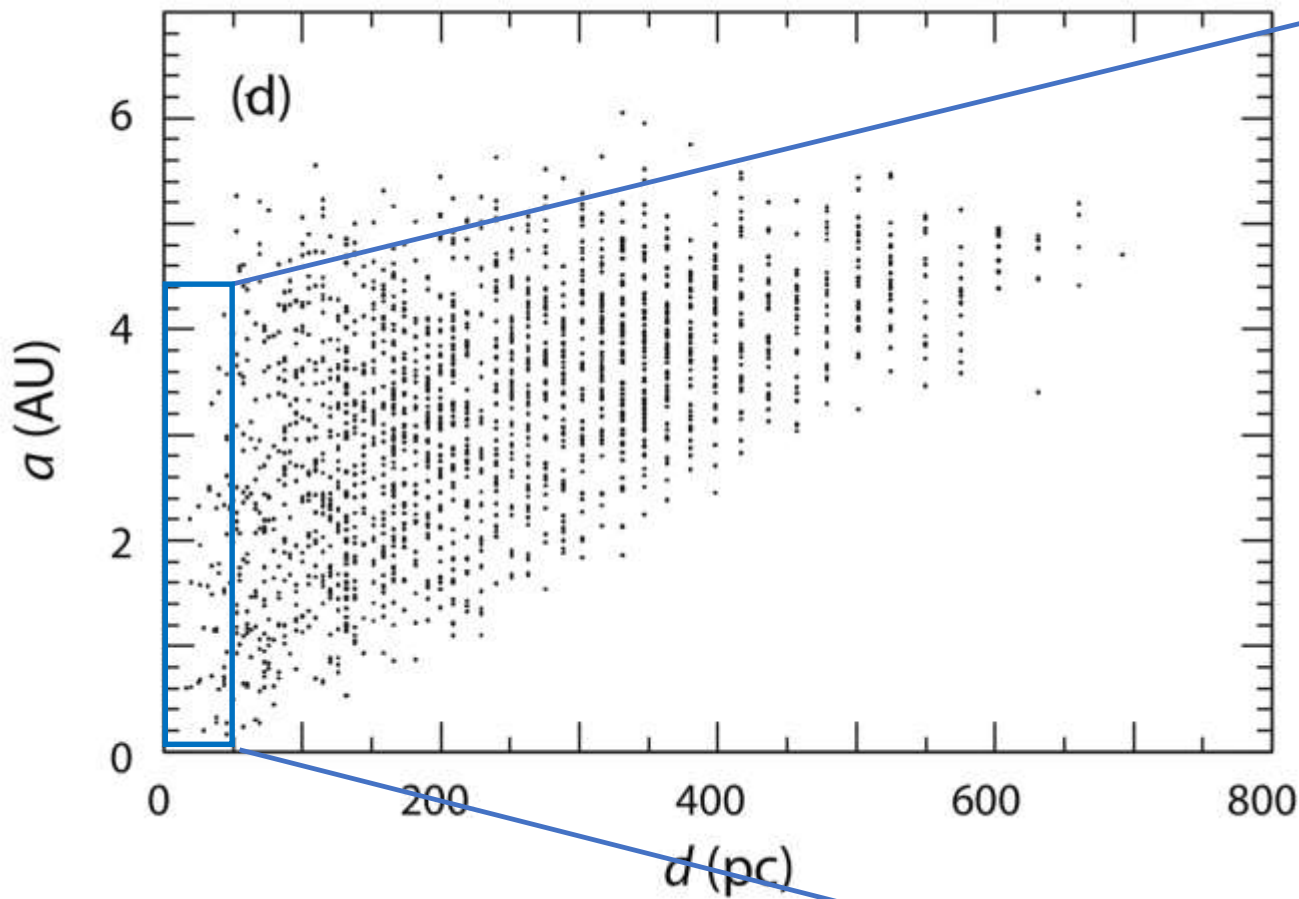
Gaia constraints on CGI targets



Neil Zimmerman (GSFC)

Rob De Rosa (Stanford/ESO)

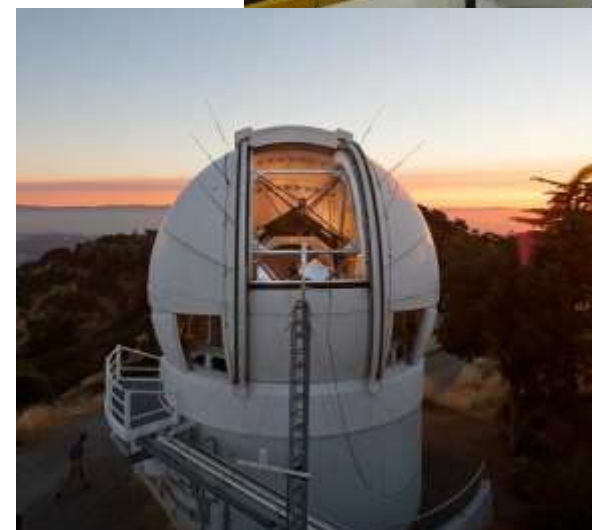
- Most CGI known-RV planet targets have expected astrometric amplitudes (semi-major axis) of 200 microarcsec (μ as), with several as greater than 500 μ as
 - Easily detectable with Gaia's predicted snapshot precision of 50--80 μ as for $V=5$ --7 stars.
- CGI's formal requirements are for $V<5$ stars, but Gaia's final capabilities on $V<5$ stars are not yet well understood.
 - There have been efforts to develop specific data processing strategies for recovering the photocenters of bright stars (Sahlmann, et al, 2016); however the implementation of such methods in the final Gaia data release is not guaranteed.
 - CGI's technology demonstration would greatly benefit from stronger collaboration with the Gaia Data Processing and Analysis Consortium's (DPAC) in this area.



Perryman et al. 2014.
Gaia 5 year mission new detections.
For clarity, only 1 in 10 planets are plotted.

RV precursors aid exoplanet target selection

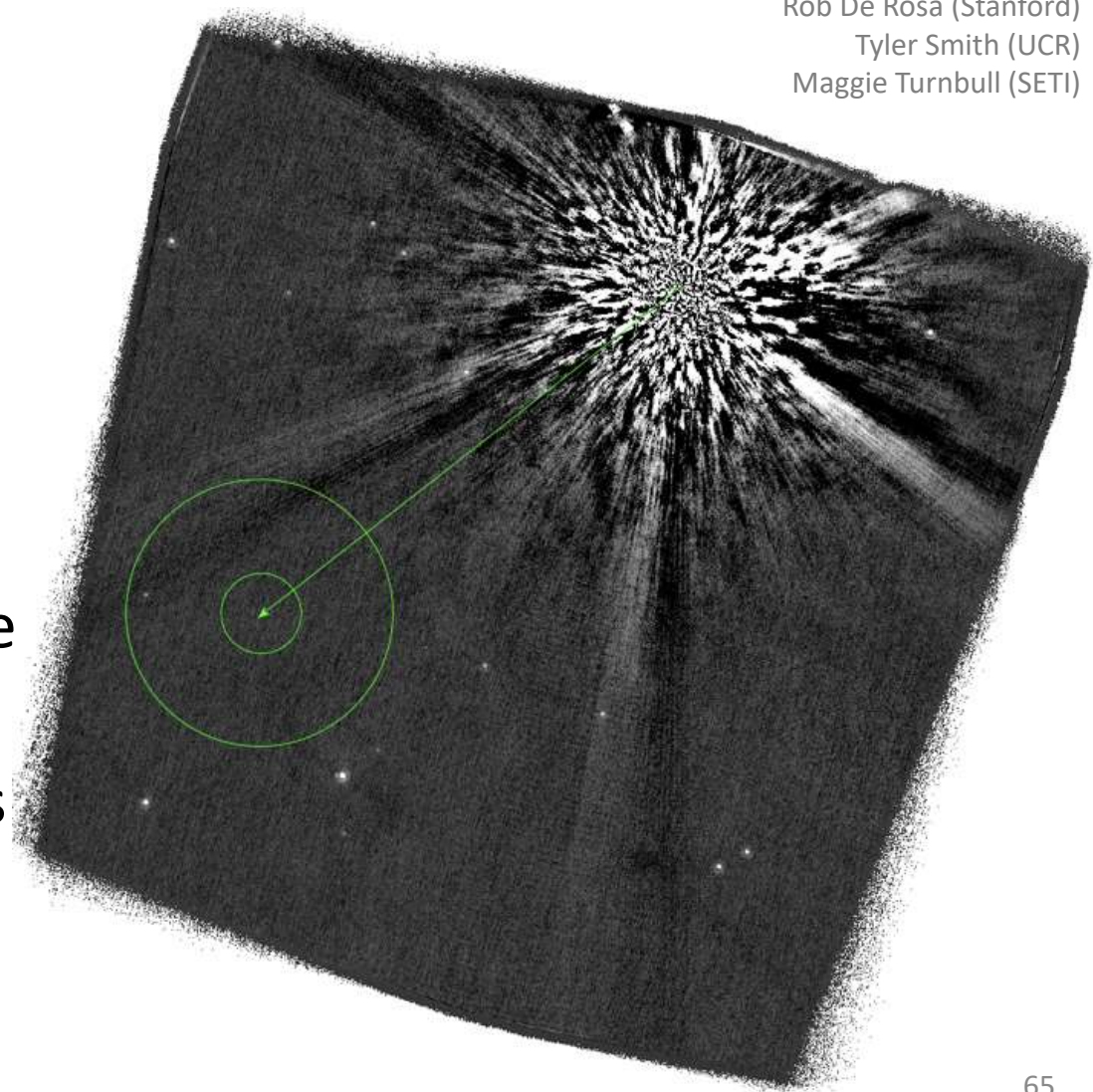
- RV precursor work needed to:
 - Refine ephemerides for CGI RV planets
 - Needs: 1-2 nights per year for next several years
 - Survey nearby stars discover more RV planets
 - Would need: ~2 weeks on NEID per year until launch
 - Also aids future missions
- Automated Planet Finder now underway
- Potential NASA resources:
 - Keck, NEID time & Key Projects
 - southern facilities (MINERVA, CHIRON)



Imaging precursors aid exoplanet target selection

Vanessa Bailey (JPL)
Misty Craycraft (STScI)
Rob De Rosa (Stanford)
Tyler Smith (UCR)
Maggie Turnbull (SETI)

- CGI target stars near the Galactic Plane could be contaminated by background stars
- Keck/NIRC2 precursor imaging of high-priority, high-proper motion CGI targets is mostly complete
- Required future work: survey CGI reference stars for binary companions

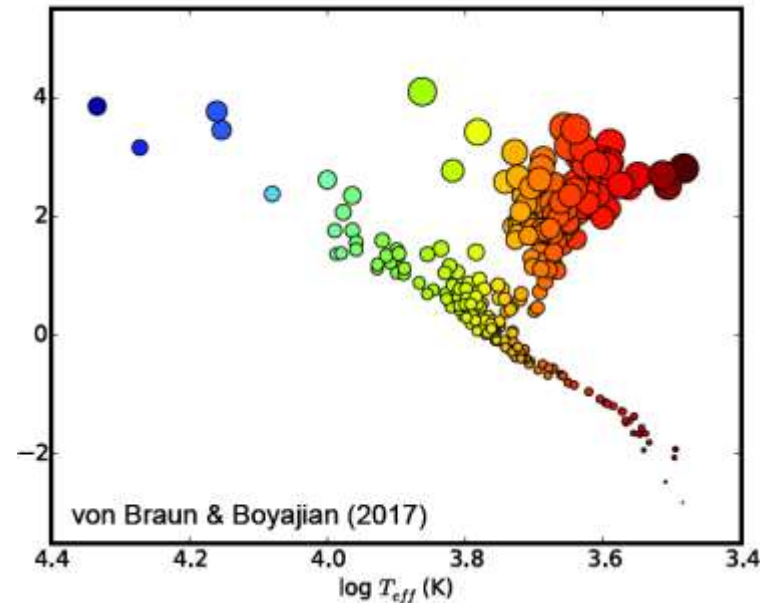


Ground-based optical interferometry to measure fundamental stellar parameters of CGI targets



Neil Zimmerman

- Georgia State University's CHARA Array has measured the precise radii of numerous exoplanet host stars, including $\sim 1/3$ of the 20 best CGI targets.
- An observing campaign to complete such measurements on all top-priority targets would add value to CGI in two ways:
 - For RV planet targets, the uncertainty in the mass of the star can be a significant contribution to the error in the semi-major axis of the planet's orbit. An independent estimate of the stellar mass can refine the global fit of the orbit parameters (e.g., von Braun, et al. 2012) and thereby assist in predicting the observability as a function of time.
 - If CGI acquires reflected-light photometry and spectroscopy of a planet, more precise knowledge of the stellar radiation incident on the planet and of the system age can inform atmosphere modeling efforts, and the retrieval of specific parameters such as CH₄ abundance (Batalha, et al., 2019).



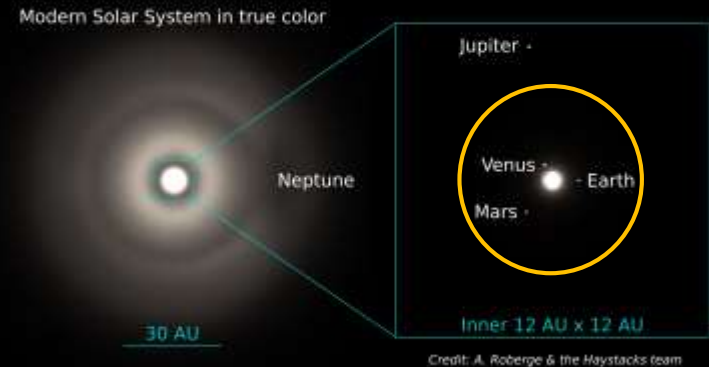
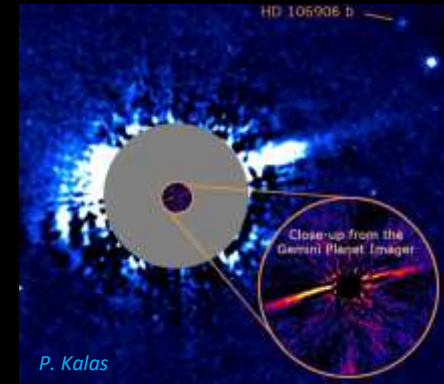
Empirical H-R diagram constructed using direct measurements of stellar radii

(von Braun & Boyajian, 2017).

Disk backups

Environment Matters

- Protoplanetary & Transition disks
 - Newly-forming planetary systems
- Debris disks
 - Remains of planet formation
 - Colliding or evaporating minor planetary bodies
- Exozodi disks
 - Can potentially shroud planets from observations



Known Cold Debris Disks

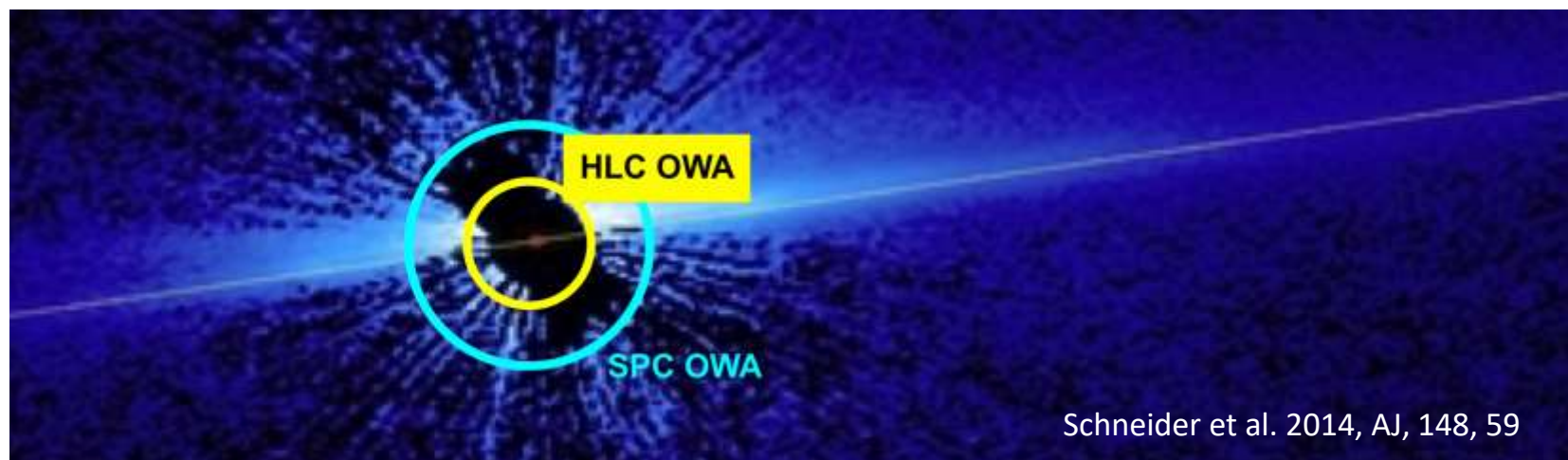


John Debes (STScI)

Ewan Douglas (UofAZ)

Bertrand Mennesson (JPL)

- **Q:** Where does circumstellar material come from and how is it transported?
- **CGI can:** Map morphology and scattered light flux of faint disks at smaller working angles than HST
- **During TDP:** 2-3 disks
- **Beyond TDP:** Additional disks with a variety of properties



Known Cold Debris Disks

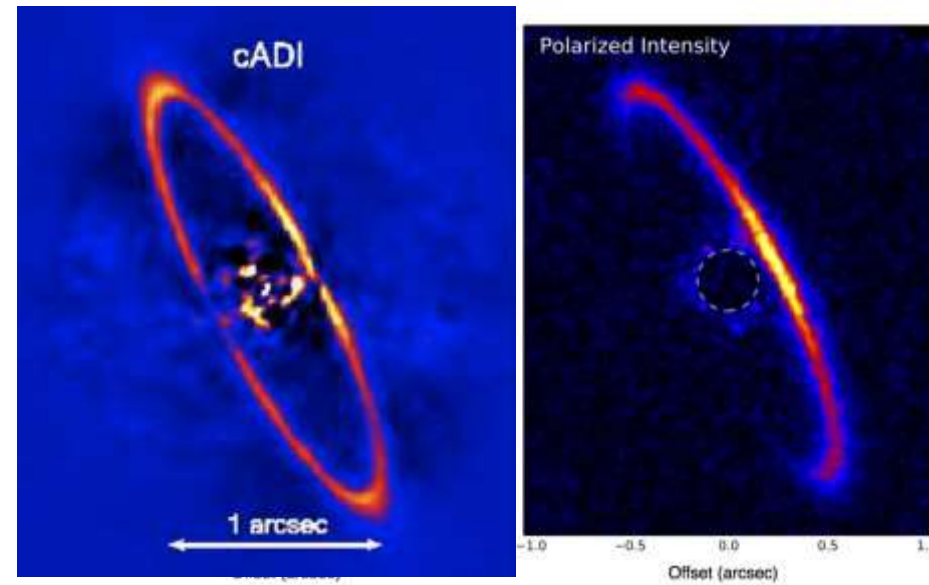


John Debes (STScI)

Ewan Douglas (UofAZ)

Bertrand Mennesson (JPL)

- **Q:** What is the composition of planetary dust in the inner regions of debris disks?
- **CGI can:** Map color, degree of forward scattering, and the degree of polarization.
- **During TDP:** 1-2 disks
- **Beyond TDP:** Additional disks with a variety of properties



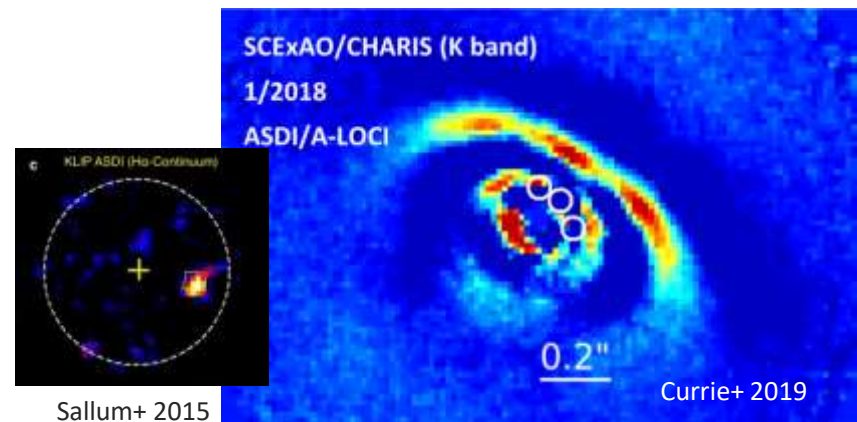
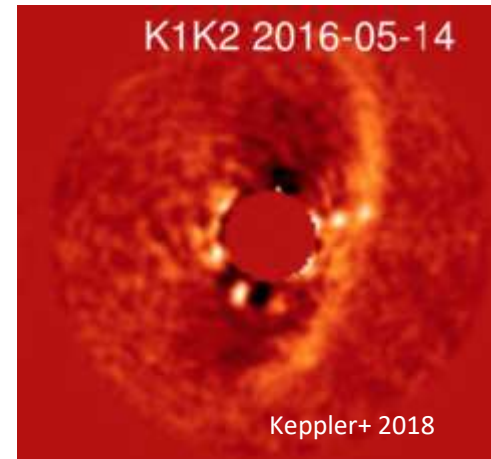
Perrin+2015
Milli+2017

Protoplanetary systems



Kate Follette (Amherst)
Ewan Douglas (UofAZ)

- **Q:** What are the accretion properties of low-mass planets in formation? How can we distinguish protoplanets vs. disk structures?
- **CGI Can:** Measure H-alpha at high contrast
 - Caveat: CGI will not achieve optimal performance on faint host stars. Performance modeling TBD.
- **During TDP:** *Perhaps* a test observation
- **Beyond TDP:** Observe transition disks with gaps in CGI FOV



First visible light images of exozodiacal dust



John Debes (STScI)
Ewan Douglas (UofAZ)
Bertrand Mennesson (JPL)

- **Q:** How bright is exozodiacal dust in scattered light? Will it affect exo-Earth detection with future missions?
- **CGI can:** Probe low surface density disks in habitable zone of nearby stars
- **During TDP:** *Opportunistic*, as part of exoplanet observations
- **Beyond TDP:** Survey best 25-50 potential exo-Earth targets for future missions



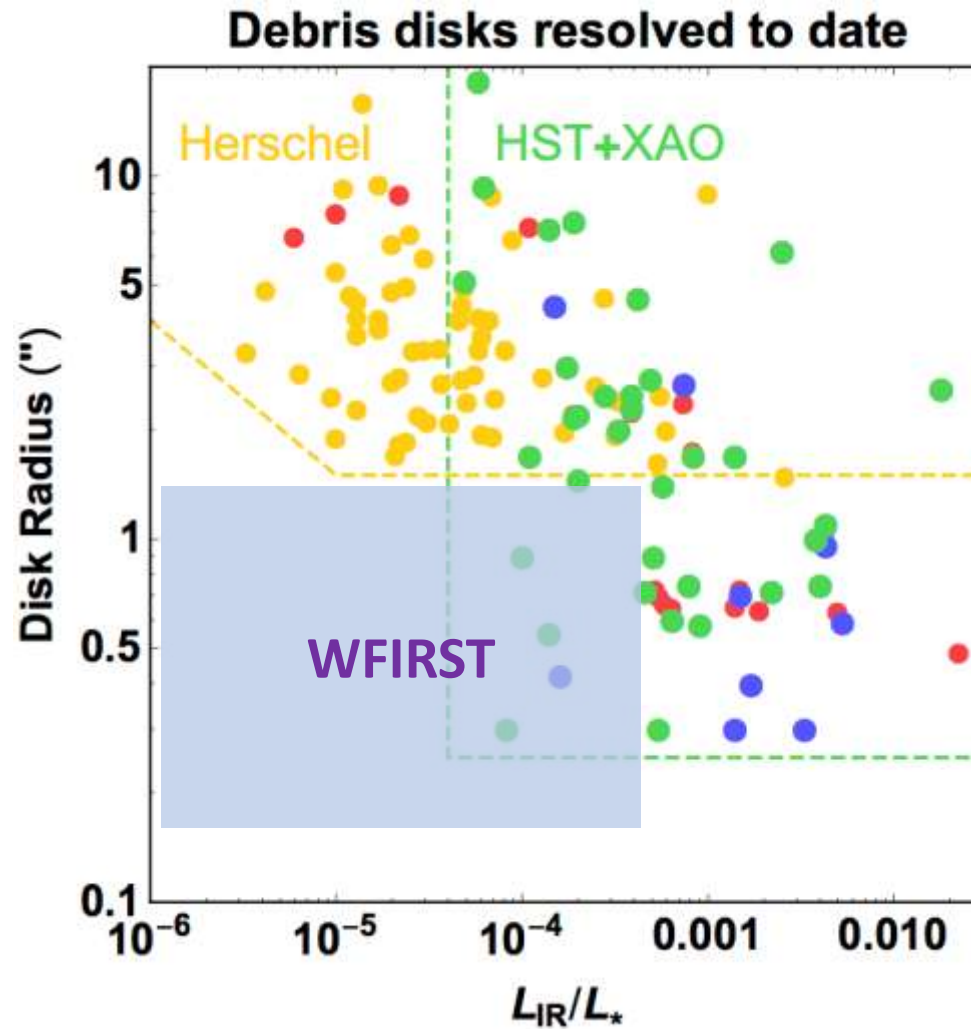
Summary of CGI Exozodi potential vs Instrument Performance



John Debes (STScI)
Ewan Douglas (UofAZ)
Bertrand Mennesson (JPL)

Case	IWA (mas)	5 σ Point Source Detection Limit per resel at IWA	Number of stars with dark hole intercepting HZ	Median Exozodi Sensitivity (per resel)	Nb of stars observable in 100h/ 500h
CBE	150	$\sim 10^{-8}$	76	15	15/58
Current Req't (ex BTR5)	200	2.5×10^{-8}	41	61	8/26
TTR5 at 575nm	300	10^{-7}	14	270	8/14
"TTR5 at 825nm"	430	10^{-7}	2	582	2/2

PRELIMINARY analysis!
Please do not post publicly

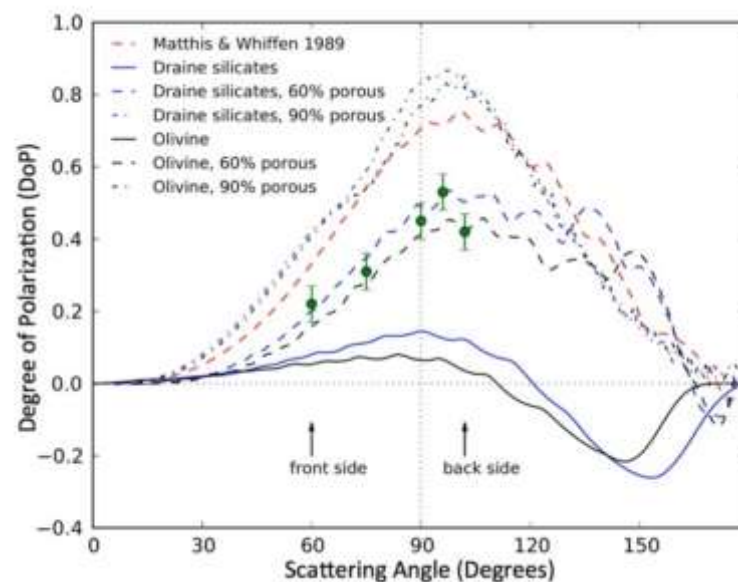
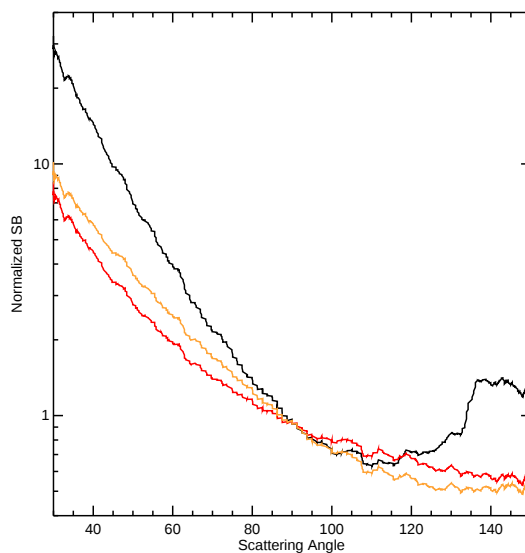
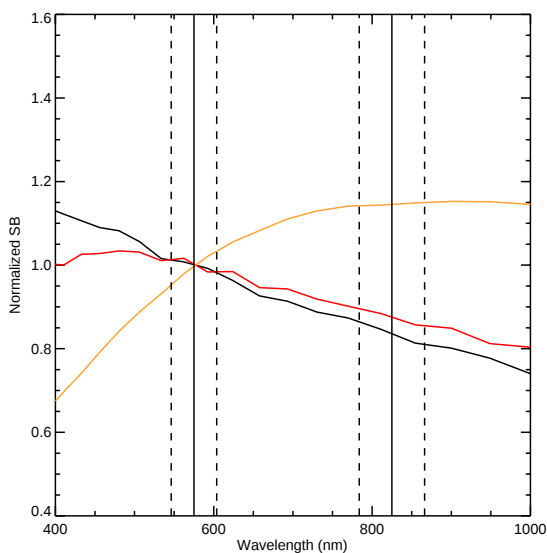


Dust Composition



- Combination of scattering efficiency, forward scattering, and polarization fraction (DOP) can constrain compositions

Water ice, organics, silicates



CGI vs FM requirements



Parameter	CGI vs. Future missions <i>unobscured aperture: HabEx & LUVOIR B</i>
5 σ Flux ratio at 3 λ/D (6 λ/D)	n/a ($\sim 10^{-7}$) vs. $5 \cdot 10^{-11}$ ** L1 Threshold Requirement

** NTE = not-to-exceed = requirement on max tolerable.

CGI vs FM requirements



Part 1

Parameter	CGI vs. Future missions <i>unobscured aperture: HabEx & LUVOIR B</i>
Pointing jitter control	Slightly worse CGI: ~0.5mas RMS V=5 star, FM: 0.3mas NTE**
Low order control (Z4-Z11)	~10x better (~100pm RMS) ~100pm RMS * vs ~1nm NTE **
EMCCD	Comparable: dark current, clock-induced-charge Worse: QE at UV/red. 21mo lifetime req.

** NTE = not-to-exceed = requirement on max tolerable.

CGI vs FM requirements



Part 2

Parameter	CGI vs. Future missions <i>unobscured aperture: HabEx & LUVOIR B</i>
Wavefront error sources	Comparable? Can't probe "new physics" (amplitude & polarization) as well at 10^{-7}
High order drift ($\geq Z12$)	10x Worse CGI: 50pm NTE**, FM: 5pm NTE**
# of DMs	Same (2)
DM stroke resolution	~8x worse (15pm vs 2pm) Engineering problem, not physics problem
DM actuator count	48x48 vs 64x64

* model, typically without Model Uncertainty Factors (MUFs)

** NTE = not-to-exceed = requirement on max tolerable.

International Contributions



WFIRST Program Level Requirements Appendix (PLRA), 2020-01-30

International partners may^b participate in the project by providing the following contributions:

JAXA	Coordinated, contemporaneous ground-based observations on Subaru
	Ground station for telemetry and tracking
	Polarization optics for the CGI
	Microlensing data from the MOA project
MPIA	Access to microlensing data from and observations with PRIME
	Precision mechanisms for the CGI
ESA	Batteries
	Star trackers
	Electron Multiplying Charge-Coupled Device (EMCCD) detectors for the CGI
	Ground station for telemetry and tracking
CNES	Superpolished optics for the CGI
	Grism data processing
	Coronagraph starlight suppression algorithms

Status Date: 1/26/20



Preliminary Disposition of Tiger Team Recommendations

#	Recommendation	Disposition			
		WFIRST Project	CGI	Preliminary Consensus	descope now or offramp later? HQ included in decision?
1	When faced with decisions, choose the side of simplest design or test that meets threshold, not the side of deeper contrast.	Yes	Yes	Yes	offramp No
2	Carry an incompressible test list that has only one mode (direct imaging) with test of function and model correlations.	Yes	Yes	Yes	offramp No
3	The WFIRST Level 2's state CGI has a 5.25 year life; this needs to be corrected to be consistent with the anticipated tech demo duration.	Consider	Yes	Yes	now Yes
4	HQ should clarify the timeline and hours available for tech demo completion in WFIRST mission, consistent with Class C reliability.	Consider	Yes	Yes	same scope Yes
5	Do all High Order Wave Front Sensing and Control (HOWFSC) calculations on the ground.	Consider	Yes	Yes	now No
6	Consider moving other processes such as phase retrieval and calibrations to ground	Consider	Yes	Yes	now No
7	Consider moving the MPIA/JPL interface. Specifically, have MPIA (with their industrial partner) deliver both PAM and PAME.	No	No	No	same scope No
8	If EDU schedule impacts FLT deliveries, be prepared to overlap the EDU and FLT	Yes	Yes	Yes	offramp No
9	The Mechanical WBS integrates and tests the FSM and FCM mechanisms and delivers in-place to the Adaptive Optics WBS which adds the flat mirrors and does more tests. Look for savings by integrating/merging the testing in the two WBS elements.	Yes	Yes	Yes	same scope No
10	The EDU and FLT EMCCD detectors come from the same lot. Get EDU earlier with minimal screening.	Yes	Yes	Yes	offramp No
11	Relaxing the star magnitude (Mv=4 or brighter... Level 1 says Mv=5), identify suitable brighter science targets, and for purely technical experiments consider possibility of even brighter targets and brighter reference stars. Potential gains will likely be mode-dependent.	Yes	Consider	Consider	now Yes
12	Increasing number of DM opens/shorts that can be tolerated (beyond 5/6 offramp already taken). Because impact depend on how they are distributed, run the models when the DMs get connectorized	Consider	Consider	Consider	offramp No
13	Relaxing DM precision and stability. 15-bit DAC linearity performance (without hardware change) is consistent with DM electronics stability of 1 mV (from 0.5mV). [CGI has adopted this already].	Yes	Yes	Yes	now No
14	Relaxing filter specs - 1% wide filters with high optical density could be relaxed...drives procurement.	Yes	Yes	Yes	now No
15	That timing/efficiency should not drive anything. WFIRST should be asked to give CGI the time that is needed. Use efficiency metrics to see if relief is worthwhile. Chopping cadence to reference can be optimized. CONOPS is a useful knob to buy back performance	Consider	Consider	Consider	same scope No
16	Have fallback hardware options wherever possible for both flight and EDUs.	Yes	Yes	Yes	offramp No
17	Safe to mate EGSE alternatives for any avionics that drive EDUs.	Yes	Yes	Yes	offramp No
18	Buy additional EDUs to add schedule robustness	Yes	Consider		offramp No
19	In case of a late EDU element, use existing testbed element for testing (project has adopted this offramp)	Yes	Yes	Yes	offramp No

Moving forward: balancing performance with constraints



- L1 - Threshold Technology Requirement:
 - “TTR5: WFIRST shall be able to measure (using CGI), with $\text{SNR} \geq 5$, the brightness of an astrophysical point source located between 6 and 9 λ/D from an adjacent star with a VAB magnitude ≤ 5 , with a flux ratio $\geq 1 \cdot 10^{-7}$; the bandpass shall have a central wavelength ≤ 600 nm and a bandwidth $\geq 10\%$.”
- CGI design is not changing as a result of L1 relaxation
 - Re-design to Threshold would hurt both cost & schedule
- However, if required to stay “in the box,” CGI will accept as-built performance and/or additional risk
 - Key decisions advised by stakeholders

CGI plan to stay off the critical path



- 0th line of defense - accept as-built performance
 - If schedule and/or cost margin are not available, demonstrated performance will be accepted
 - Use CGI integrated modeling & performance budget to assess impact
- 1st line of defense – aggressive schedule management
 - Instituted off-ramps with schedule work arounds to be used if necessary
- 2nd line of defense - adequate schedule reserve
 - Increased funded schedule reserve during II&T by postponing some verification by test to post launch (eg. stability tests)
- 3rd line of defense – rescope II&T test program to the Incompressible Test List that covers only one mode (L1 Threshold requirement)

Consequence of Class D: allowed to trade cost/schedule for risk

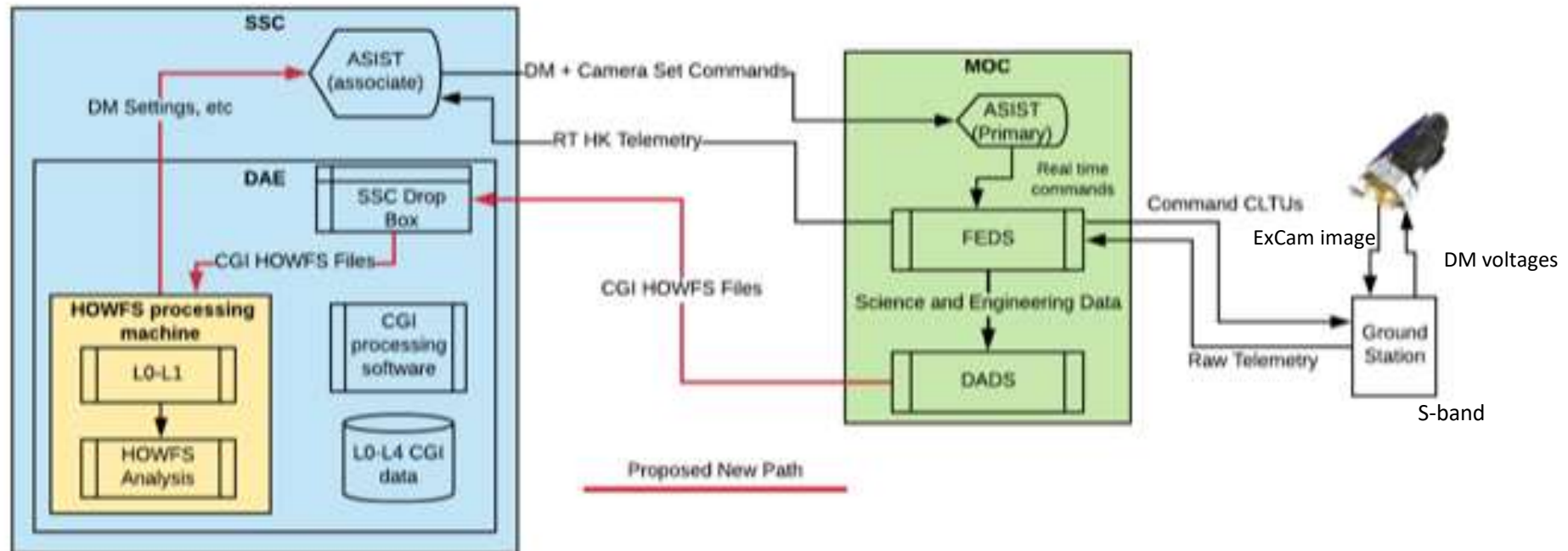
- Tailoring currently in progress.
- Example: electronics parts:
 - Many parts already in procurement => no benefit to reducing quality
 - May reduce some screening or conduct in parallel if schedule driver
- Example: simplify process and oversight
 - Drawing quality and review
 - Lower level sign-off for documents, reviews, etc.

Accepted Tiger Team Recommendation: HOWFS Ground-in-the-Loop



- Offload computations to ground (at IPAC/SSC)
 - Downlink images, uplink DM commands
 - Significant schedule risk reduction for CGI (avionics/FSW)
 - Consistent with the current WFIRST ground systems architecture
 - Existing CGI HOWFS/C timing requirements can be met with margin using S-band up/down link
 - Data volume, ground station coverage, and down/uplink rates
- Will bring to PDR maturity for WFIRST Ground Systems PDR in July 2020

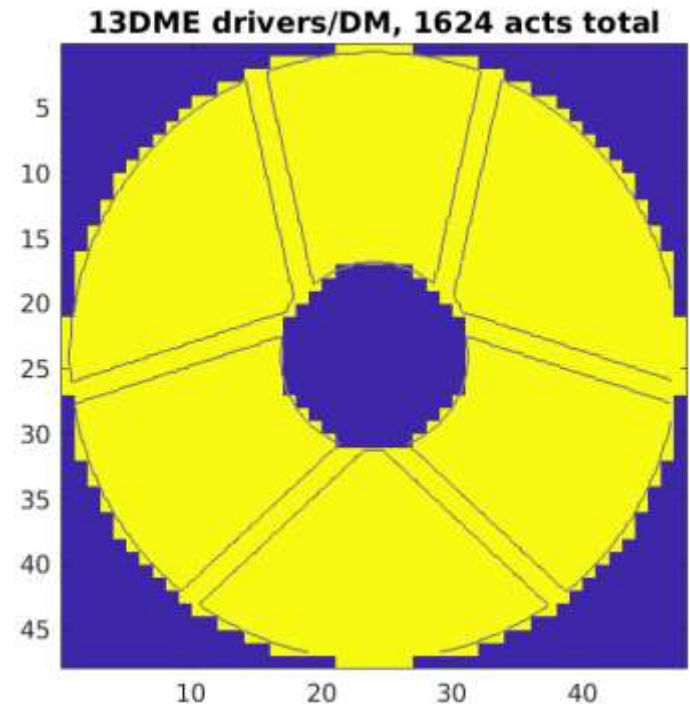
Data Flow for HOWFS Ground-in-the-Loop



Design change example:

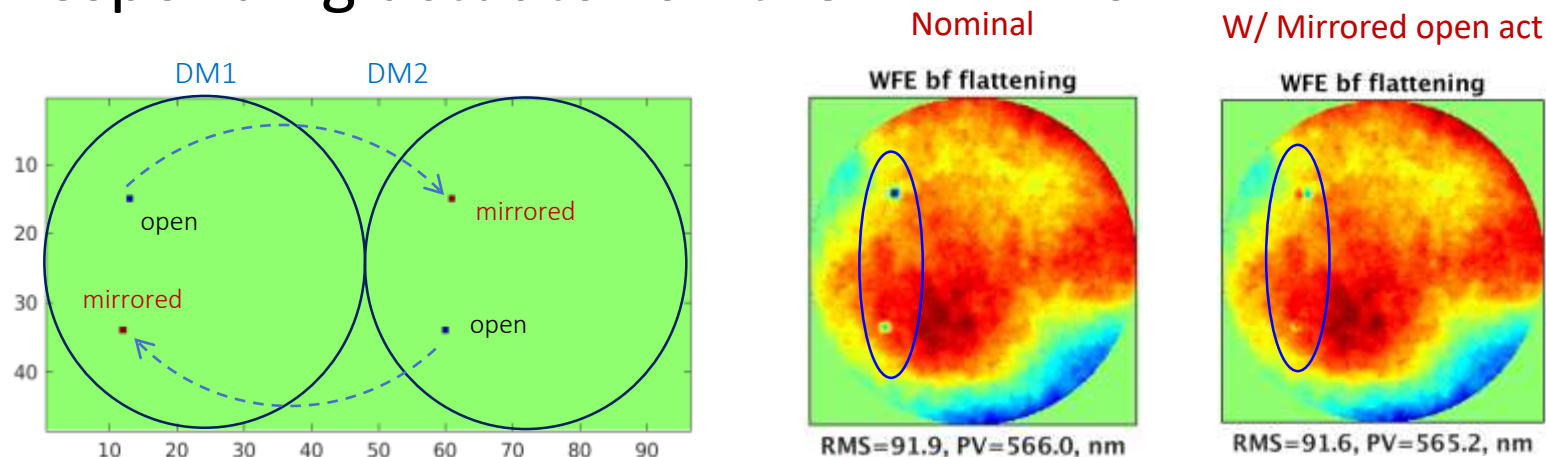
Do not drive unilluminated actuators

- Number of driver boards per DM : 16->13
- Reduces mass, power, cost, schedule for minimal performance risk



Accepted offramp example: accept up to 5 bad actuators per DM

- Open actuators on one mirror can be mitigated using the corresponding actuator on the 2nd mirror



- New acceptance criteria are based on integrated modeling
 - HLC & SPC bowtie (spectroscopy) minimally impacted
 - SPC wide FOV raw contrast may degrade up to ~4x, depending on exact distribution of bad actuators.
- Trades schedule robustness for modest performance risk