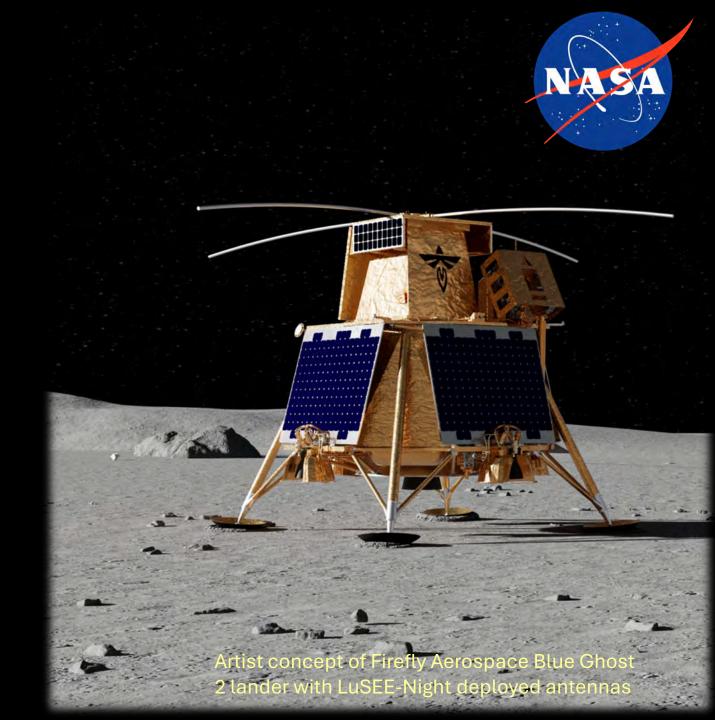
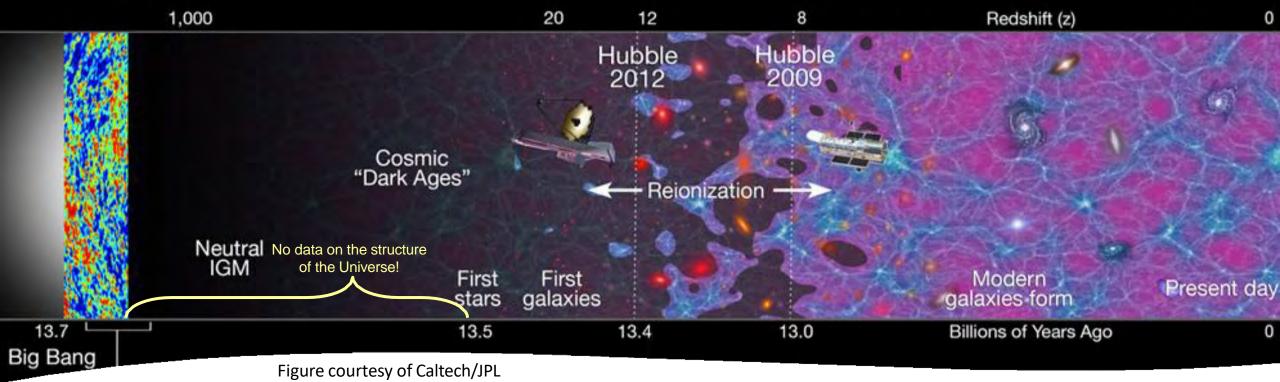
## LOW FREQUENCY RADIO TELESCOPES ON THE MOON

Jack O. Burns

University of Colorado Boulder

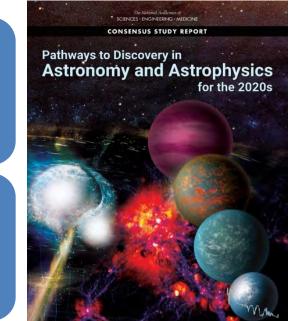
Key Non-Polar Destinations Across the Moon to Address Decadal-level Science Objectives with Human Explorers: Panel on Heliophysics, Physics, and Physical Science 10 June 2025





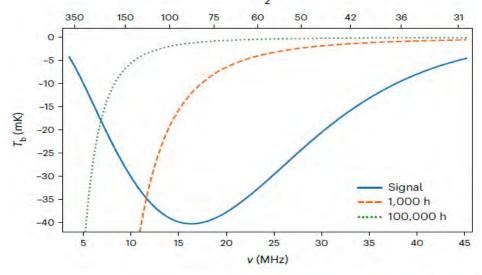
The Astro2020 Decadal Survey cosmology panel singled out the Dark Ages as THE Discovery Area in Cosmology; and identified "needed capabilities to include... both higher sensitivity and a better understanding of instrumental systematics and astrophysical couplings."

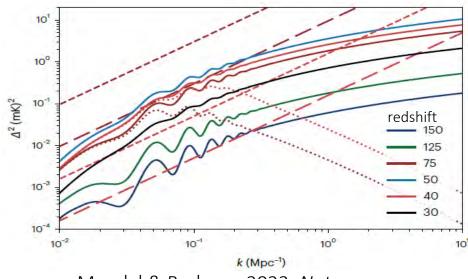
The Dark Ages is part of the Universe where JWST & other missions cannot explore – when the universe was filled with just neutral hydrogen – no stars or galaxies.



#### Exploring the Dark Ages of the Early Universe

- Global 21-cm spectrum can be measured by single antenna.
- Using an interferometer, structure power spectra & tomographic maps trace evolution of Universe from before/during birth of first stars.
- Spatial fluctuations, governed by linear structure formation, allows precise predictions for standard cosmology.
- Departure from these well-constrained predictions will provide insights into new physics possibly including:
  - Nature of Dark Matter (e.g., warm dark matter, sterile neutrinos)
  - Early Dark Energy
  - Masses of neutrinos and their hierarchy
  - Non-Gaussianity of initial density perturbations
  - Imprints of gravitational waves to reveal the energy scale of cosmic inflation.

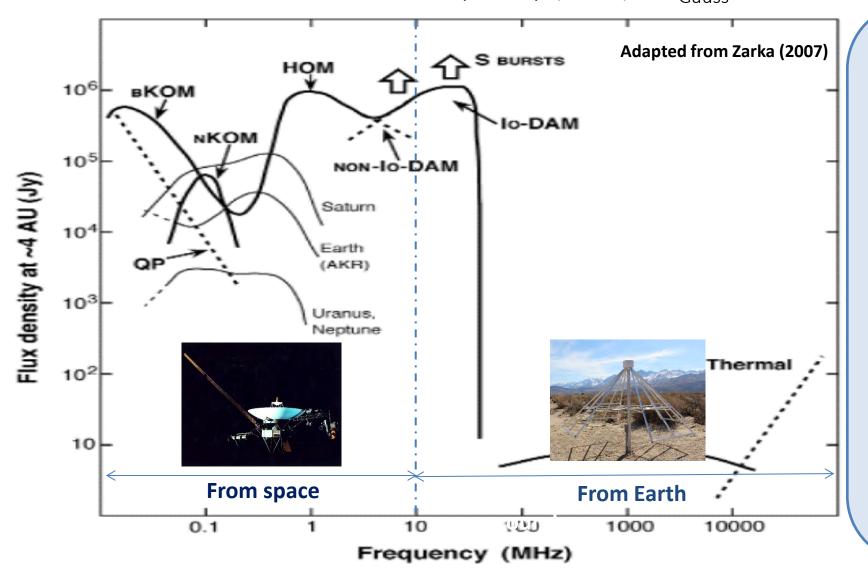




Mondal & Barkana, 2023, *Nature Astronomy*, 7, 10

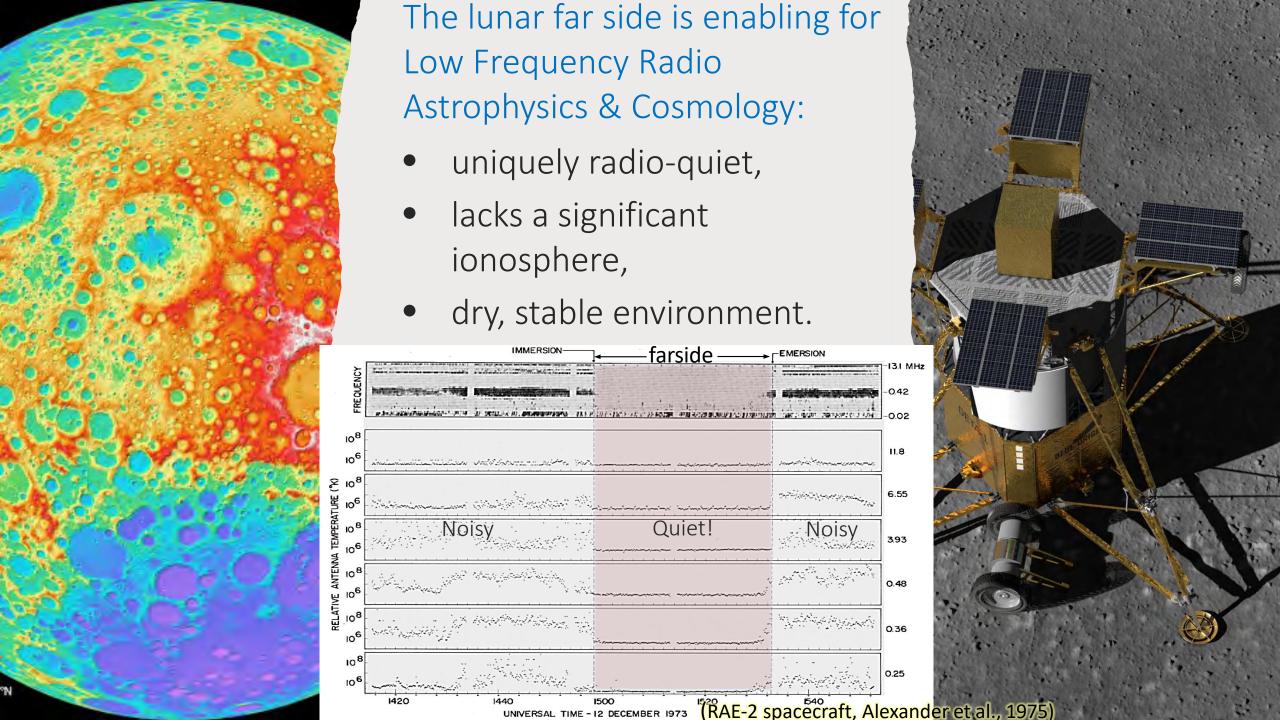
#### Measuring B-fields for Exoplanets

Electron cyclotron maser emission Frequency (MHz) =  $B_{Gauss} \times 2.8$ 



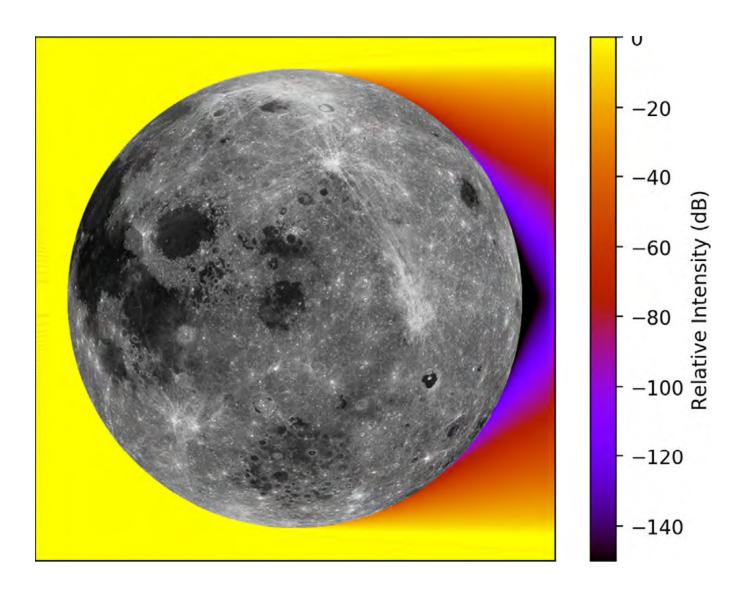
#### Planetary Science Decadal Recommendations on Magnetospheres & Exoplanets

- Excited aurorae in the waveband spanning **radio** to X-ray spectrum are a window into the workings of these magnetospheres.
- Strategic Research search[es] for magnetospheric activity at exoplanets with remote sensing of phenomena associated with magnetic fields, potentially including radio emission.
- M-dwarf stars which house exoplanets often exhibit extreme levels of stellar activity, which may adversely impact the habitability of orbiting planets.



#### Numerical Modelling of the Radio-Quiet Zone on the Lunar Far Side

Results of a 4000 x 4000 km FDTD simulation at 30 kHz. RFI incident from the left is attenuated behind the Moon on the right. An image of the Moon from the NASA LROC. Bassett, Rapetti, Burns et al. 2020, Ad. Space Research, 66, 1265.



#### NASA's Artemis & CLPS Program

















IM-1 with ROLSES antennas deployed



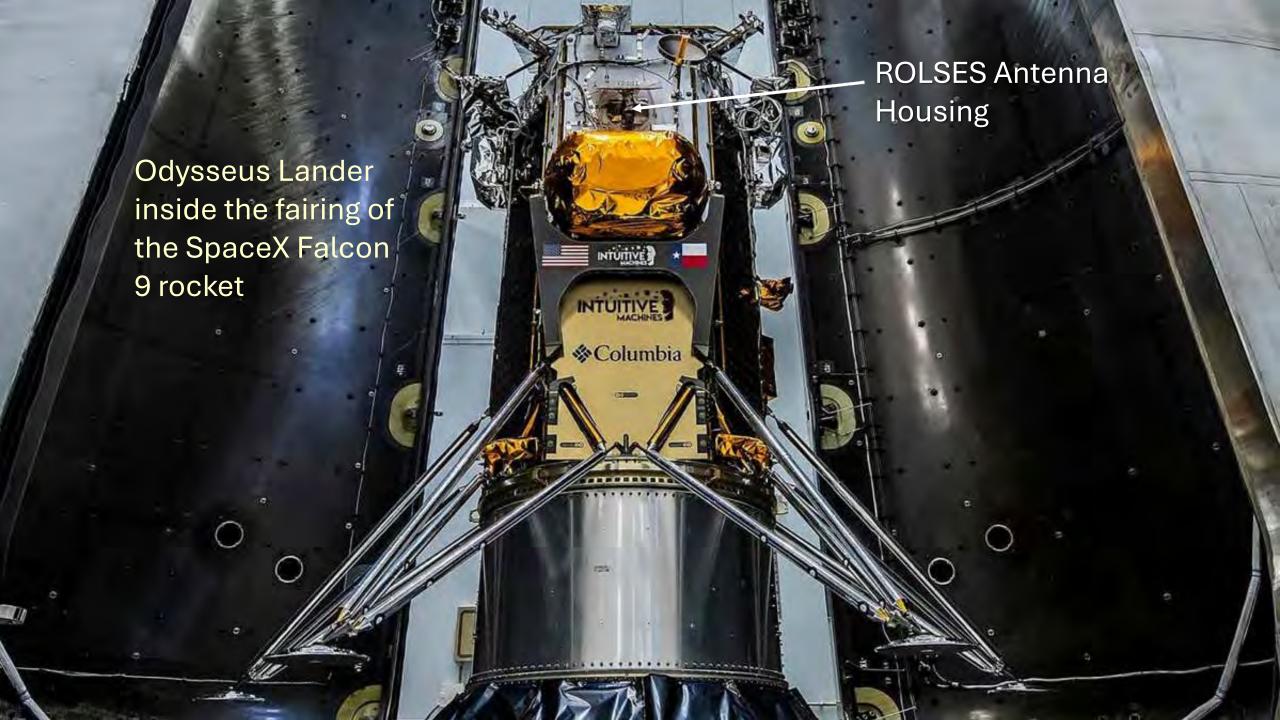
**Stowed STACER antennas** 



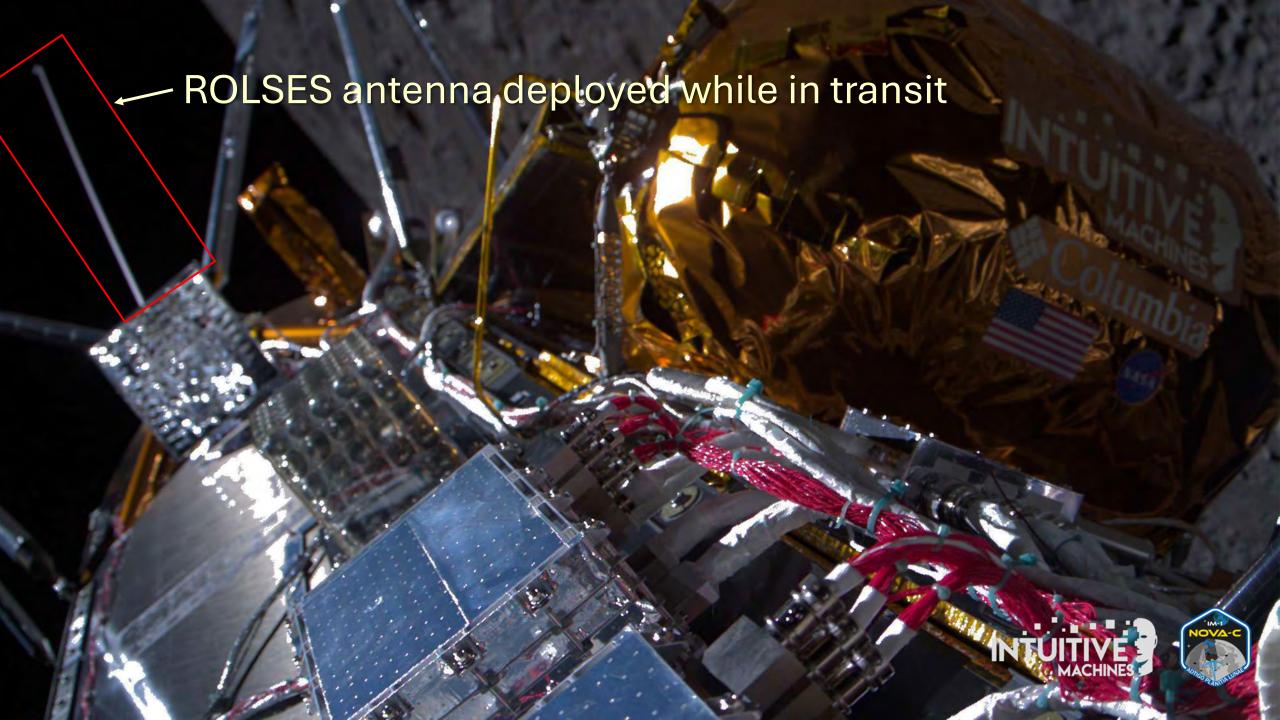
**ROLSES** spectrometer board

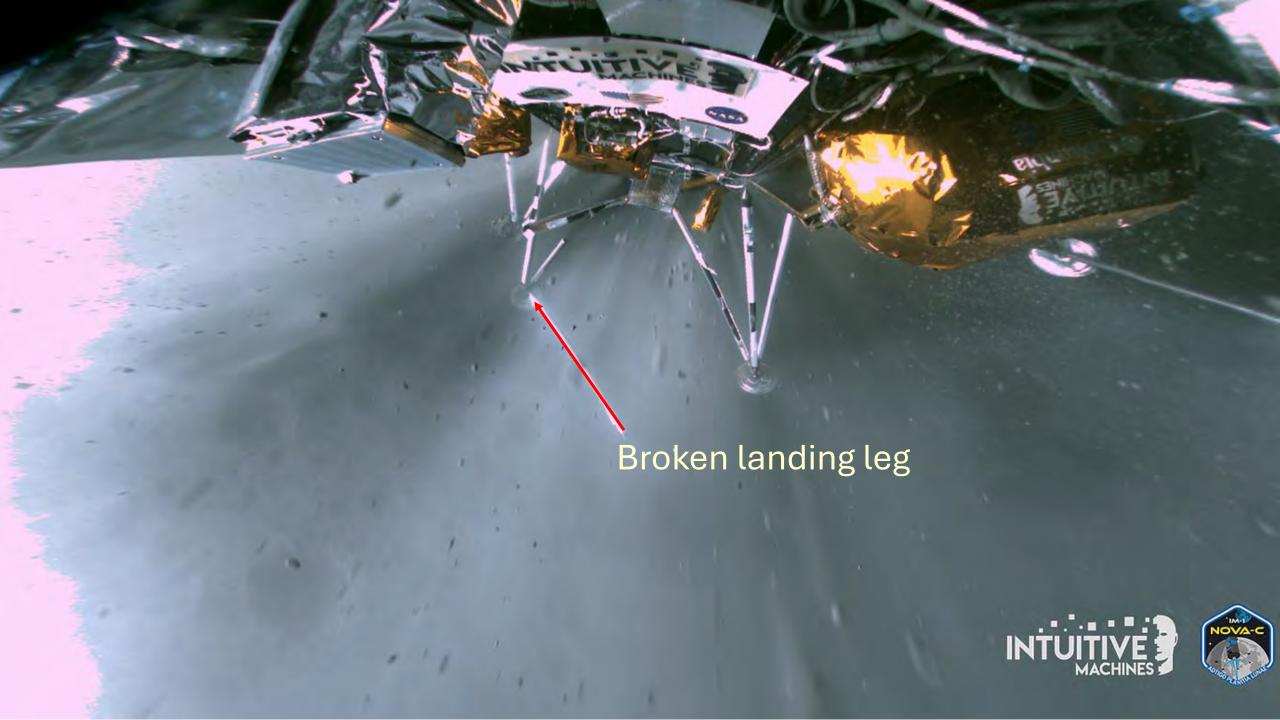
# Radio wave Observations at the Lunar Surface of the Electron Sheath (ROLSES)

- ROLSES Team: N. Gopalswamy (PI), J. Burns, R. MacDowall, W. Farrell, D. Bradley, M. Reiner, D. McGlone, M. Choi, S. Murphy, R. Katz, I. Kleyner, D. Rapetti, J. Hibbard, J. Dorigo Jones.
- ROLSES instrument is a new build with heritage from NASA SMAP Earth Remote Sensing satellite:
  - Four 2.5-m monopole antennas.
  - Radio spectrometer with 2 bands:
     10 kHz 1 MHz and 10 300 kHz 30 MHz.
- Landed near-side, South Pole region.









#### Results from NASA's First Radio Telescope on the Moon: Terrestrial Technosignatures and the Low-Frequency Galactic Background Observed by ROLSES-1 Onboard the Odysseus Lander

Joshua J. Hibbard, <sup>1</sup> Jack O. Burns, <sup>1</sup> Robert MacDowall, <sup>2</sup> Natchimuthuk Gopalswamy, <sup>2</sup> Scott A. Boardsen, <sup>2</sup> William Farrell, <sup>3</sup> Damon Bradley, <sup>3</sup> Thomas M. Schulszas, <sup>2</sup> Johnny Dorigo Jones, <sup>1</sup> David Rapetti, <sup>4,5,1</sup> and Jake D. Turner<sup>6</sup>

<sup>1</sup>Center for Astrophysics and Space Astronomy, Department of Astrophysical and Planetary Science, University of Colorado Boulder, CO 80309, USA

<sup>2</sup>NASA Goddard Space Flight Center, 8800 Greenbelt Rd, Greenbelt, MD 20771, USA

<sup>3</sup> DeepSpace Technologies Inc, Columbia, MD 21045, USA

<sup>4</sup>NASA Ames Research Center, Moffett Field, CA 94035, USA

<sup>5</sup>Research Institute for Advanced Computer Science, Universities Space Research Association, Washington, DC 20024, USA <sup>6</sup>Department of Astronomy and Carl Sagan Institute, Cornell University, Ithaca, New York 14853, USA

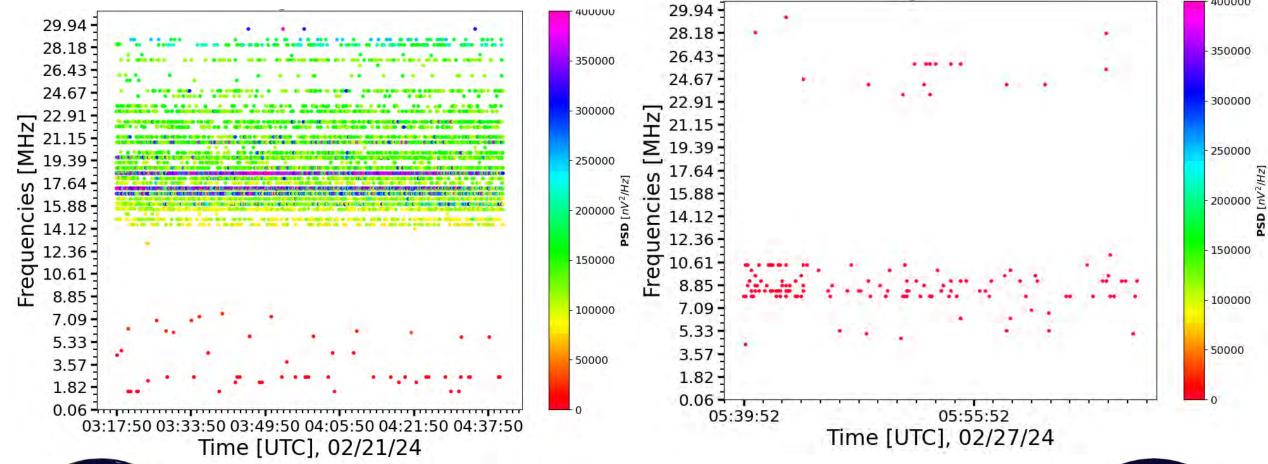
#### ABSTRACT

Radiowave Observations on the Lunar Surface of the photo-Electron Sheath instrument (ROLSES-1) onboard the Intuitive Machines' Odysseus lunar lander represents NASA's first radio telescope on the Moon, and the first United States spacecraft landing on the lunar surface in five decades. Despite a host of challenges, ROLSES-1 managed to collect a small amount of data over fractions of one day during cruise phase and two days on the lunar surface with four monopole stacer antennas that were in a non-ideal deployment. All antennas recorded shortwave radio transmissions breaking through the Earth's ionosphere—or terrestrial technosignatures—from spectral and raw waveform data. These technosignatures appear to be modulated by density fluctuations in the Earth's ionosphere and could be used as markers when searching for extraterrestrial intelligence from habitable exoplanets. After data reduction and marshaling a host of statistical and sampling techniques, five minutes of raw waveforms from the least noisy antenna were used to generate covariances constraining both the antenna parameters and the amplitude of the low-frequency isotropic galactic spectrum. ROLSES-2 and LuSEE-Night, both lunar radio telescopes launching later in the decade, will have significant upgrades from ROLSES-1 and will be set to take unprecedented measurements of the low-frequency sky, lunar surface, and constrain the cosmological 21-cm signal. ROLSES-1 represents a trailblazer for lunar radio telescopes, and many of the statistical tools and data reduction techniques presented in this work will be invaluable for upcoming lunar radio telescope missions.



#### Techno-signatures

#### Lunar Surface, Calibrated, Cleaned





#### Clear, good S/N measurements of Human-made Earth Transmissions

- Instrument frequency bandwidth corrections applied
- Flagged channels with internal EM interference
- Corrected for instrumental shelf noise



Hibbard, Burns, MacDowell, Gopalswamy et al. 2025, submitted to ApJ

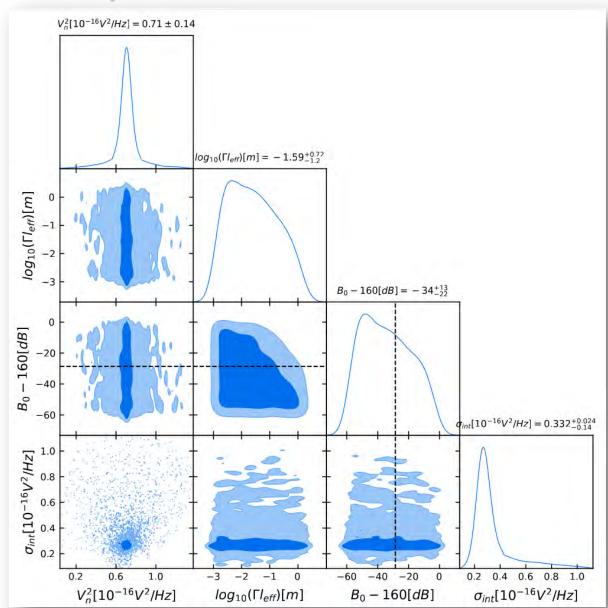
#### First Detection of the Galaxy from the Moon

#### Parameter Constraints

$$V_r^2 = V_{\text{noise}}^2 + \frac{2\pi}{3} Z_0 \Gamma^2 l_{\text{eff}}^2 B_{\nu}(\nu)$$

where  $Z_0$  is the impedance,  $\Gamma$  is antenna efficiency,  $I_{eff}$  is the effective length of the antenna, and B is the Galaxy brightness

- At different levels of frequency averaging, there is a slight preference for a model with the Galactic spectrum over one without (comparing KS-test p-values).
- Black dashed line shows the expected value of the Galactic amplitude from the Novaco and Brown 1978 spectrum.



#### Radio Astronomy Payloads on NASA CLPS Missions

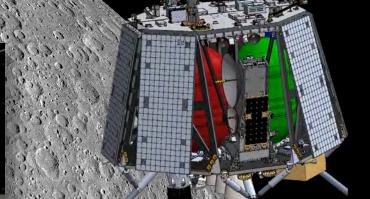
see Burns+ 2021, Planetary Science Journal, 2, 44B



Intuitive Machines delivered ROLSES (Radio wave Observations at the Lunar Surface of the Electron Sheath) to the lunar South Pole in February 2024.



Firefly Aerospace will land LuSEE-Night on far side in 2026. It probes the Dark Ages of early Universe.



iSpace/Draper Apex lander will bring LuSEE – Lite to the Schrödinger Impact Basin on far side in 2027. Science includes measuring plasma waves & surface-geospace interactions.



ROLSES-2 will be delivered to near-side Gruithuisen domes in ~2028 on Firefly Blue Ghost 3 lander.



Courtesy of Firefly Aerospace

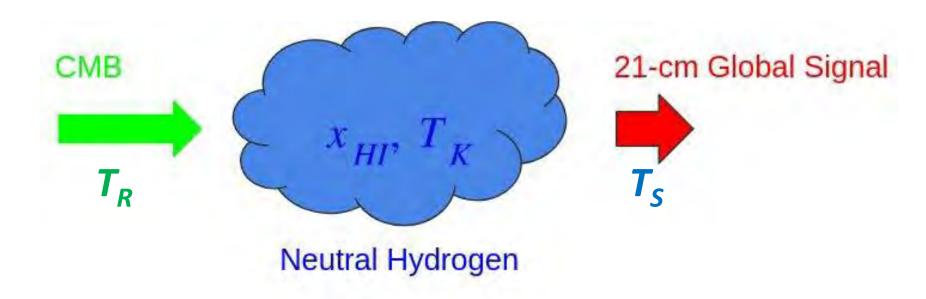
## Lunar Surface Electromagnetics Experiment (LuSEE-Night)

The first 21-cm cosmology observations from the lunar far side at night

Stuart D. Bale (PI), Jack Burns, Johnny Dorigo Jones, Keith Goetz, Christian Hellum-Bye, Sven Herrmann, Joshua Hibbard, Milan Maksimovic, Ryan McLean, Raul Monsalve, Paul O'Connor, Aaron Parsons, Marc Pulupa, Rugved Pund, David Rapetti, Kaja Rotermund, Ben Saliwanchik, Anže Slosar, David Sundkvist, and Aritoki Suzuki

Bale+, 2023, https://arxiv.org/abs/2301.10345.

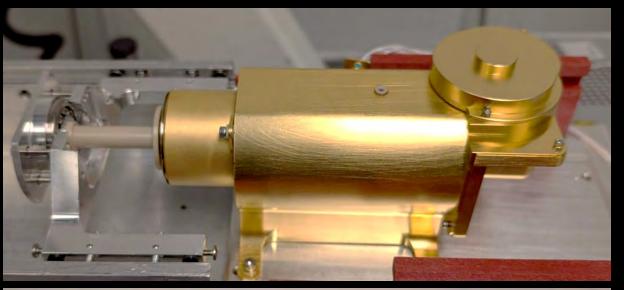
#### Differential Brightness Temperature



$$\delta T_b(\nu) \alpha x_{HI} \left(1 - \frac{T_R}{T_S}\right)$$

#### Stacer antennas

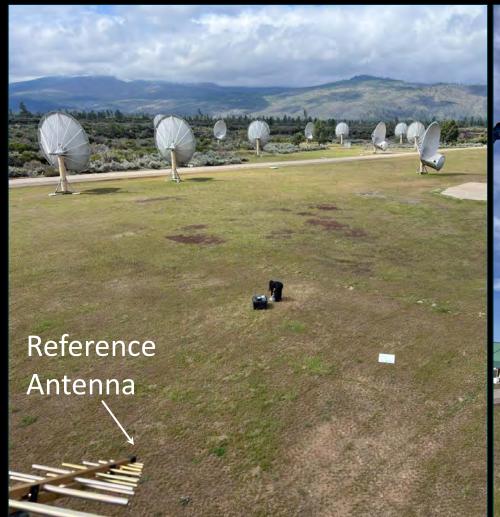


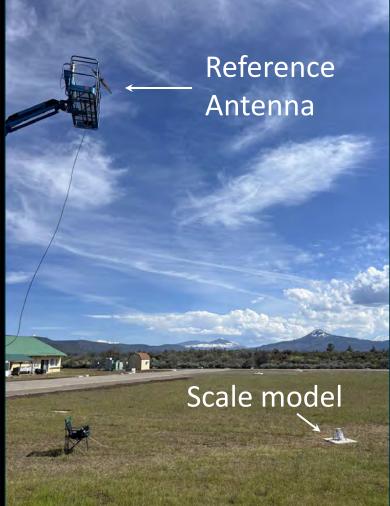


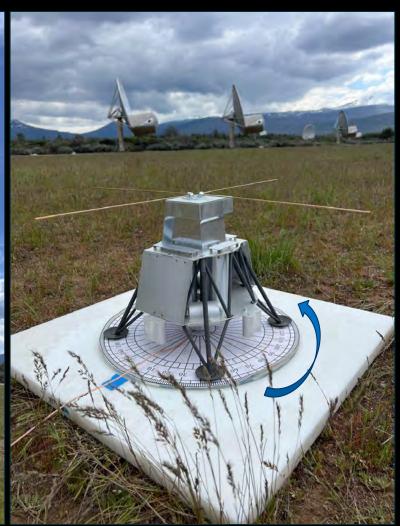


Rolled BeCu sheet, Stowed during launch, deploy as rolled flat spring

#### Beam pattern measurements



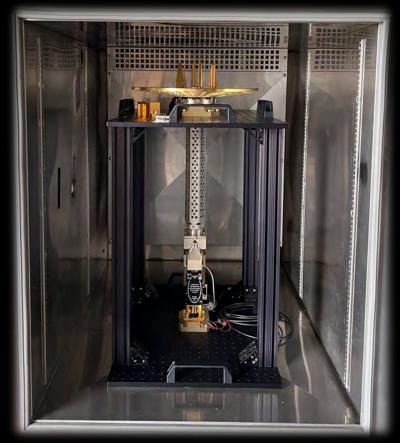




- Fixed reference antenna, rotated the LuSEE-Night scale model
- Measure slices of beam and compare against simulation

#### Flight model Assembly, Integration & Test







Assembly

Thermal test

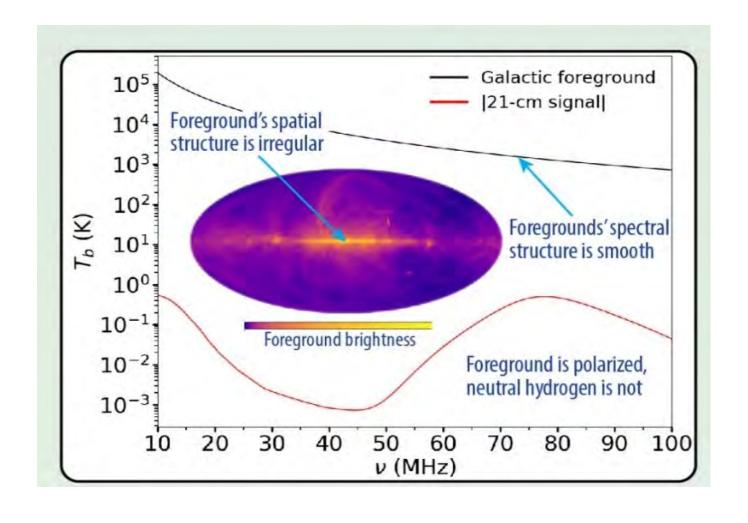
Integration

- Flight model parts assembled, torqued, tested
- We ended up gold plating almost every metal part

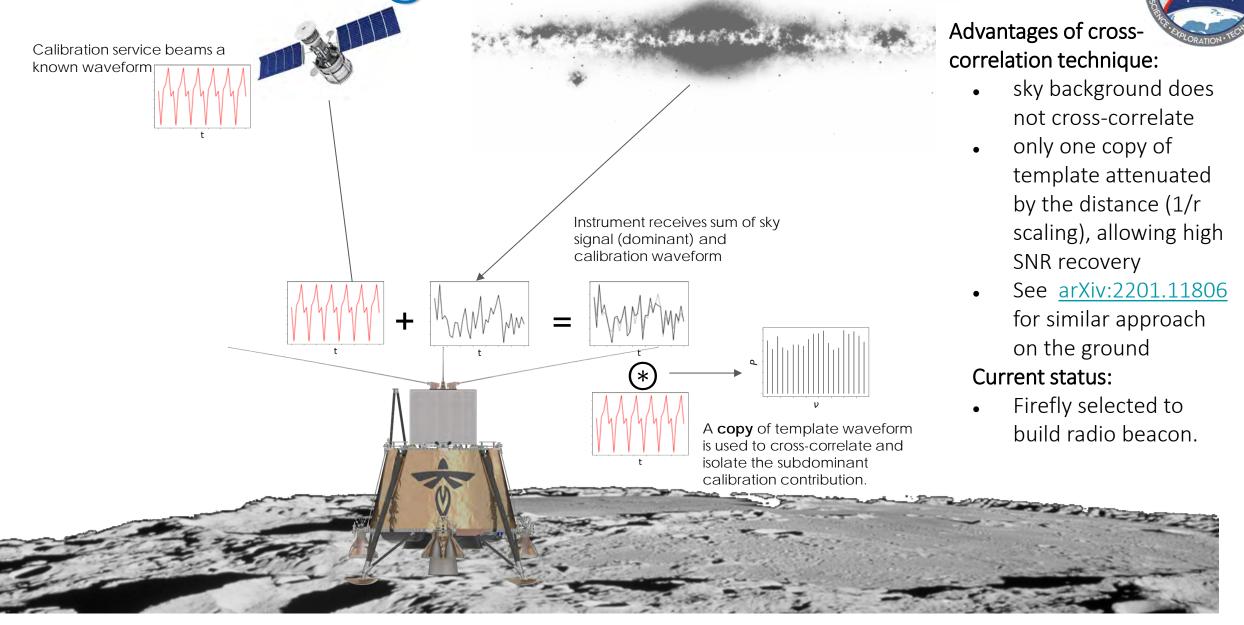
Even from the Moon, a Global 21-cm
Cosmology
Experiment is
Challenging because of Bright Foreground



Digital Control Board for LuSEE-Night



LuSEE-Night Far-Field Calibration









Phase II Concept Study funded by NASA Innovative Advanced Concept (NIAC) program An in-situ manufactured lunar far side, 100,000 dipole radio array concept for 21-cm Dark Ages cosmology

**Jack Burns** for the Science Team University of Colorado Boulder

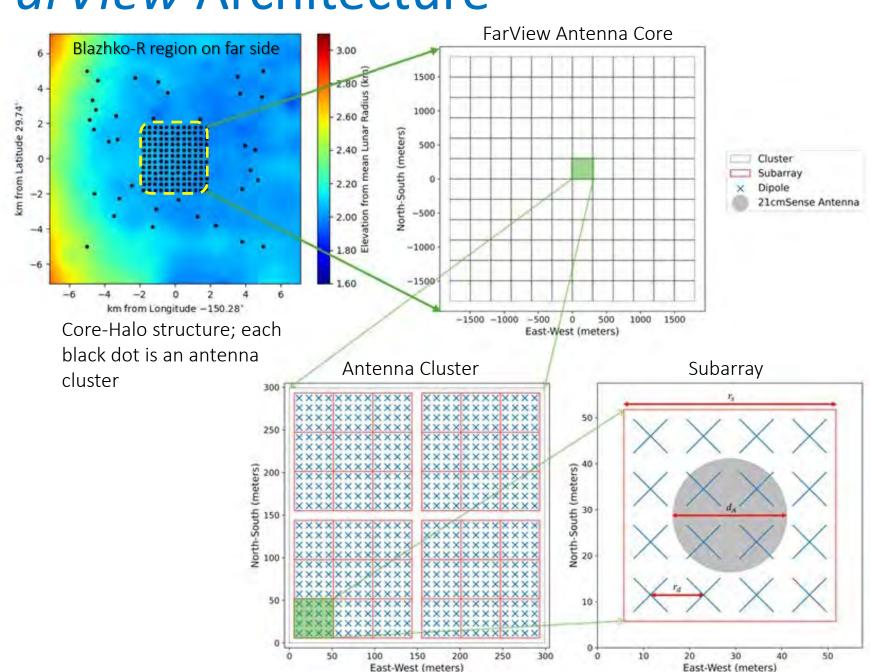
Alexander Hegedus, SRI International Jonathan Pober, Brown University Gregg Hallinan, Caltech

Nivedita Mahesh, Caltech
Tzu-Ching Chang, Caltech/JPL
Judd Bowman, Arizona State University
Ron Polidan, Lunar Resources, Inc., Pl

#### FarView Architecture

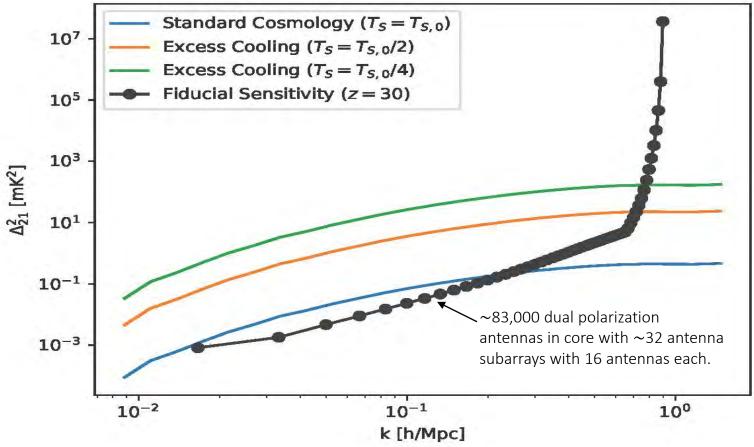
FarView architecture is hierarchical with top level in a core-halo (83,000-20,000 antennas).

- Halo is 14×14 km to measure Foreground.
- Core is 4×4 km. Antennas gathered into
  - Subarrays with 16 cross-dipoles.
  - Clusters with 36 subarrays.
- For the fiducial array  $r_d = 11.5$ m,  $r_s = 46$ -m, and  $d_A = 24.7$ -m.



#### **New Simulations of Sensitivities with Beamforming**





New numerical simulations using *21cmSense* code suggests that to achieve required sensitivity for Dark Ages, we need

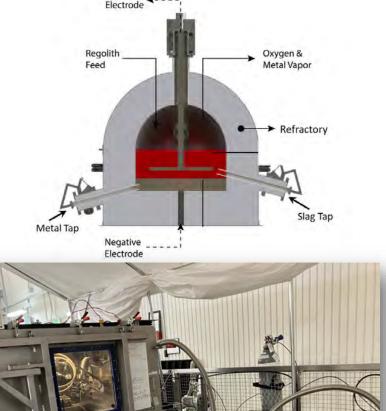
- Smaller subarrays with closer spacing between antennas.
- More subarrays and more antennas devoted to core.



### FarView Lunar Manufacturing: Molten Regolith Electrolysis

LUNA R

- Metal Extraction Electrolysis Process
  - o Extract: Iron, Silicon, Aluminum, Magnesium
- Utilize Aluminum for antennas and Silicon and Aluminum for thin film silicon solar cells
  - o Also use Magnesium for antennas
  - Collect and hold Iron for future use
- FarView Metals Requirement
  - Approx 45,000 kg
- ∴ Require 6 MRE Units Operating for ~6 yrs
  - ~6 tonnes of launch mass to gain 45 tonnes of metals
- We also developed a Wire Fabrication Process for the dipole antennas on the Moon

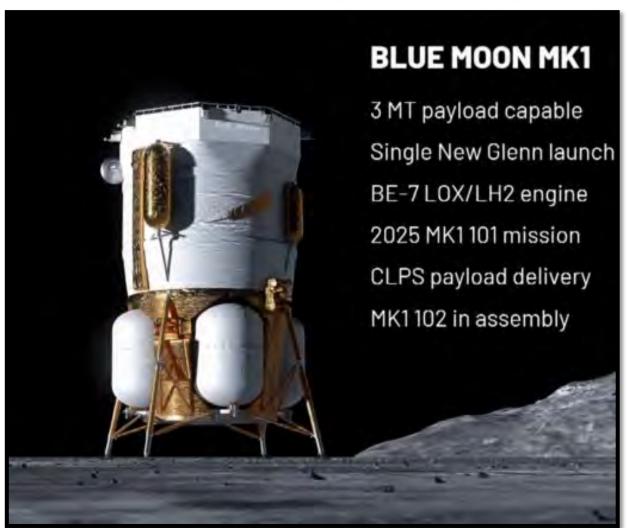


MRE undergoing testing in a vacuum chamber at NASA KSC





## A Single *Blue Moon* Lander Can Deploy a 400 Antenna Cluster & LTV on the Moon







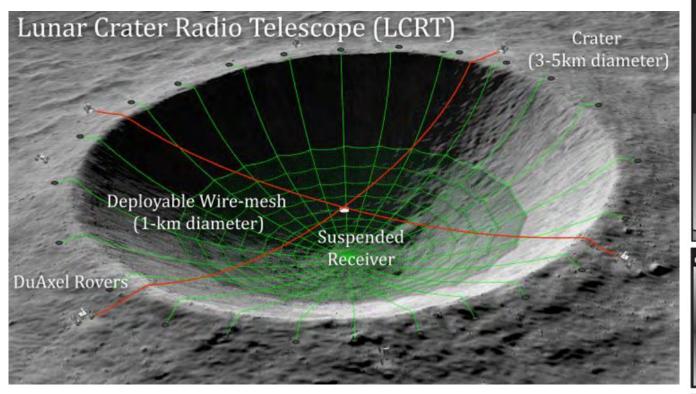


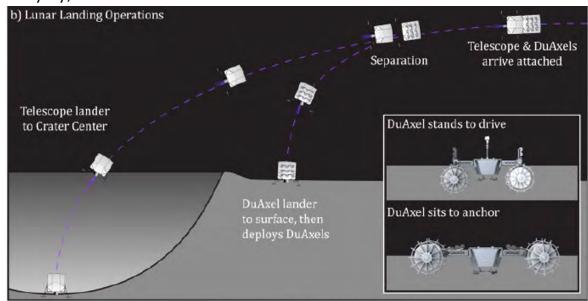
#### FarView and Astronauts – Early Stages

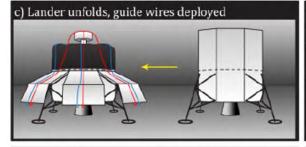
- In-situ assessment/survey of the surface and subsurface conditions to safeguard that the *FarView* site meets all the suitability criteria.
  - Ensure no small scale (~1-3 m) hazards (rocks, craters) that could not be removed with surface processing.
  - Assessment of the subsurface structure with a ground penetrating radar.
  - Deliver a small prototype array (5-20 antennas) to surface to fully evaluate the site.
- Astronaut support for *FarView* for initial landing & setup of hardware.
  - Initial landing could deliver all the hardware for *FarView* using a large lunar lander such as Blue Origin's Blue Moon landers. Pre-deploy these assets with the astronauts.
  - Checkout of extraction & manufacturing hardware may require combined astronaut/robotic support to deploy and assemble hardware and bring into operation.

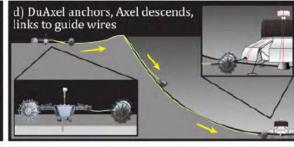
#### Lunar Crater Radio Telescope

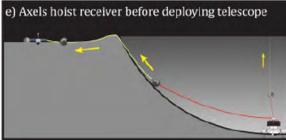
Saptarshi Bandyopadhyay, JPL

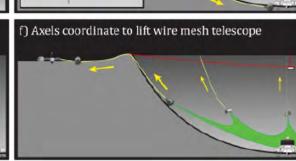












#### Radio Leakage from Starlink Satellites – Implications for the Moon

"Observations with the LOFAR radio telescope between 10 to 88 MHz and 110 to 188 MHz show broadband emission [from Starlink satellites]...The spectral power flux density of this broadband UEMR varies from satellite to satellite, with values ranging from 15 to 1300 Jy, between 56 and 66 MHz,", Bassa et al. 2024, A&A, 689, L10.



Distances and power flux density measurements of Starlink satellites that passed through the beam pattern of the two 1 h LOFAR observations.

Observed satellite (ordered by NORAD ID)

10 Jy

O 100 Jy

Y < 10 lv

40

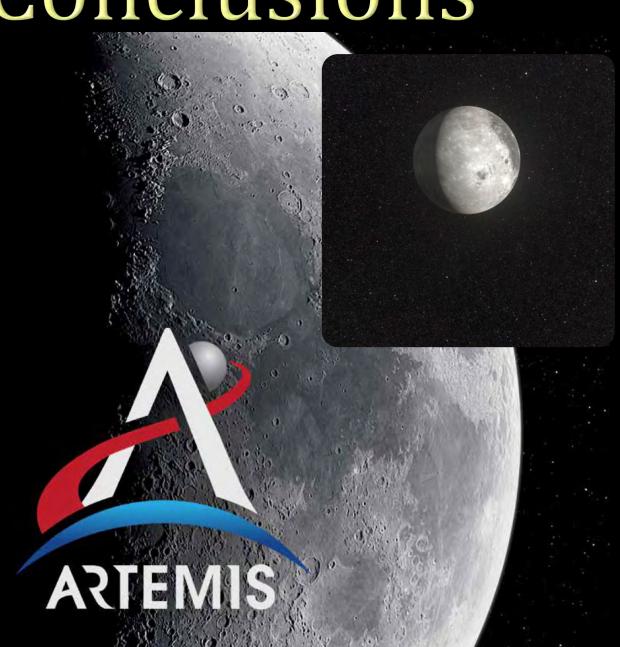
1000 Jy

200

See also Krolikowski & Elvis, 2024, *Phil. Trans. Roy. Soc.*, in press.

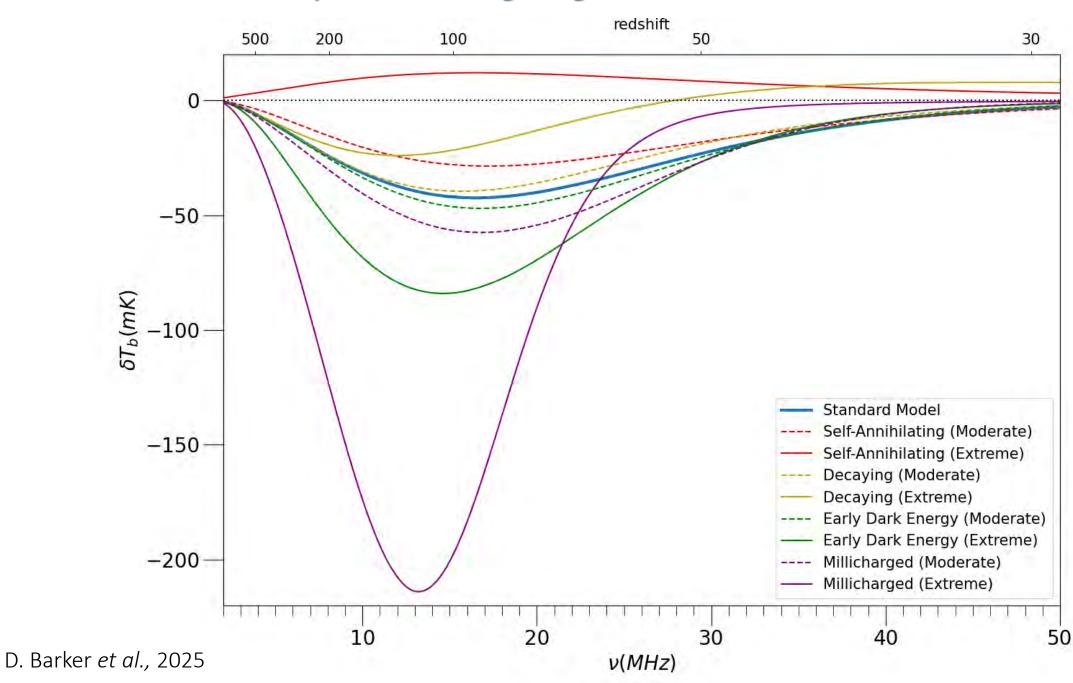
Summary & Conclusions

- NASA CLPS delivered the first radio telescope to the surface of the Moon in February 2024.
- ROLSES-2 & LuSEE-Night will further investigate lunar plasma environment & cosmic Dark Ages in next few years.
- FarView will measure 3-D spatial/spectral fluctuations to explore new physics potentially including dark matter, neutrinos, & inflation.



### Extra Slides

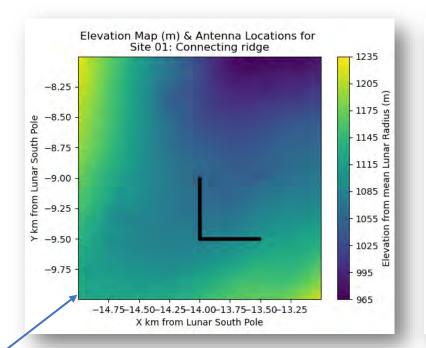
#### Effects from Exotic Physics on Dark Ages Signal: Alternative Dark Matter & Dark Energy Models

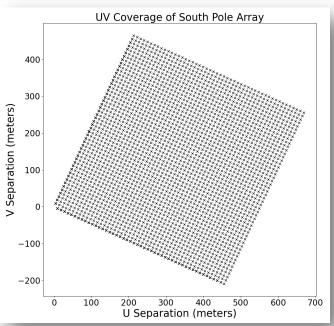


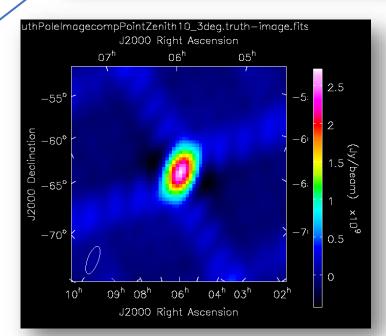
#### Low Frequency Radio Interferometric Array at Artemis Landing Site

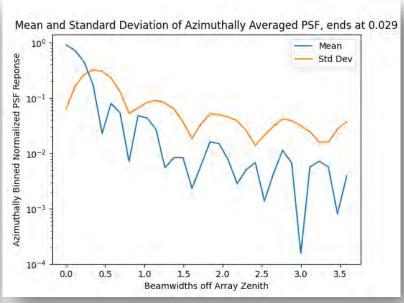
 A ~1-km tether can house up to 100 radio dipoles, each 10-m tip-totip, to create a unique, sensitive synthetic aperture radio interferometer (Hegedus+ 2023).











Point response function for Tethered Array, 1-min integration, at 10 MHz

#### **Trade Study for Beamforming**



				•	
Criteria	Weight	Sub-array	Score	Individual dipoles (no sub-arrays)	Score
Computational requirement	25%	Reduces the computations required by reducing the u-v points/number of visibilities.	5	$N^2$ computational steps with $N = 100,000$ is large. Best case with EPIC is still $N*log(N)$ .	3
Deployment	Easy rover operations		4	More distance & time travelled by rover due to larger number of cables	3
	10%	Fewer cables (less mass and volume)	5	Cables from each dipole to the central node	2
	10%	Fewer correlators on the surface (6400 correlators)	3	100,000 correlators	2
Technology Progression	5%	Building correlators for ~6400 elements is doable in the coming decade. We are already expanding to 10 <sup>7</sup> in this decade	4	Expanding to $10^{10}$ correlations is more difficult in the foreseeable future	3
Field of View (FOV)	5%	Reduces the instantaneous FOV from all-sky to $\lambda/[(N_{sub})^{0.5}*17m] \sim 25^{\circ}$ at 10 MHz	4	All-sky field of view	5
Cadence of coverage	5%	Reduced	4	Maximum possible	5
Blind transient detection rate	5%	Decreased		Best possible	5
Sample/Cosmic variance	5%	Higher	4	Lower	5
Foreground contamination	5%	Lower contamination	5	More contamination due to larger FoV	4
Chromaticity	10%	Higher	3	Lower	5
Operational Complexity	5%	Steerable beams	3	Drift scan	5
Totals	100%		4.15		3.3



FarView Mission Design Parameters Derived from the Science Analyses				
Parameter	Value	Rationale		
Location	Radio-quiet zone on the lunar far side	The radio quiet zone is a ~65° radius circular area centered on 0° latitude, 180° longitude) on the lunar far side where the natural and anthropogenic low frequency radio signals from the Earth (a major noise source for FarView) are almost totally blocked. This very low noise environment is required to detect the Cosmic Dark Ages signal.		
Frequency Band	5 - 50 MHz	For detection of highly redshifted 21-cm radiation from the Cosmic Dark Ages.		
Total Number of Dipoles	100,000	Minimum necessary to achieve cosmology sensitivity requirement.		
Antenna Distribution	Core/Halo distribution	Core should have ~80% of total number of antennas. This distribution is required to supply sensitivity in the core to detect the Dark Ages signal and enough sensitivity in the halo to adequately remove the foreground.		
Antenna Length	~10 meters	Optimal length to produced required sensitivity.		
Antenna Separation	11.5 meters	Optimized separation that maximizes sensitivity and is viable given the antenna length and rover maneuverability.		
Antenna Distribution	Zig-Zag Pattern	Cross-dipole antennas are required to measure polarization. Non-cospatial cross dipoles can be achieved by deploying dipoles with one orthogonal to the next in a zig-zag pattern.		
Positional Tolerance for Placement of Individual Antennas	~2 meters in all three axes	Tolerance needs to be <0.1λ at 15 MHz and flexibility to avoid small boulders & craters.		
Orientation Tolerance for Placement of Individual Antennas	≈10 degrees	Maintain coherence for polarization with post-processing corrections.		
Polarization Measurement	Equal number of antennas for both orientations	Allows calculation for full Stokes parameters enabling polarization measurements.		
Lunar Surface	Relatively smooth and flat with undulations of <2 m in height	Tolerance needs to be <0.1λ at 15 MHz over the area spanned by 3 dipole antennas.		
Lunar Subsurface	High crustal thickness (≥ 25 km)	Minimize subsurface reflections of the incoming radio signal that will distort the radio beam.		
Visible Horizon	Avoid sites with nearby mountains	A jagged horizon, one with lots of mountains, will compromise the data analysis. Map of horizon will be needed for data analysis to remove horizon effects.		

#### **Estimated Data Rates for FarView**



#### FarView will generate prodigious amounts of data. To estimate the data rates, we assume:

- 100,000 cross-dipoles within subarrays with bandwidth of 5-50 MHz.
- 5 kHz spectral resolution with  $\sim$ 9,000 spectral channels over 45 MHz bandwidth.
- Core is 4 km & halo is 14 km in diameter. Resolutions at 45 MHz for core is ~5.7' and for halo is 1.7'.
- Beamforming of subarrays reduces down correlations from 10<sup>10</sup> to N<sup>2</sup><sub>stations</sub>.
  - N<sub>stations</sub> for core = 144 \* 36 = 5184
  - $N_{\text{stations}}$  for halo = 35 \* 36 = 1260
- Integrate for ~1 min & 2.4 mins for longest and shortest baselines, leading to 1440 and 600 integrations every 24 hrs.
- Assume 64 bits to store complex visibilities.

#### With all these parameters, the data rate per day to be downlinked to Earth is:

- Data-rate/24<sup>h</sup> (halo) =  $N_{stations}(N_{stations}-1)/2 * N_{channels} * N_{bits} * N_{pol} * N_{integrations}$ = 793,170 \* 9000 \* 64 \* 4 \* 1440 = 329 TB/day
- Data-rate/24<sup>h</sup> (core) = 1,342,433\* 9,000 \* 64 \* 4 \* 600
   = 232 TB/day

#### We are exploring methods to reduce data rate including:

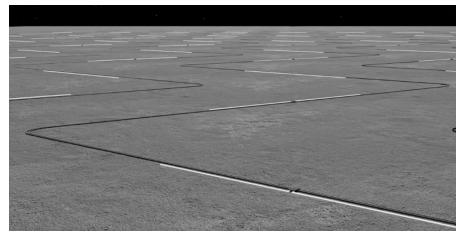
- Aggressive baseline-dependent averaging / fringe-rate filtering.
- Real-time averaging of redundant visibilities.
- **EPIC** (E-Field Parallel Imaging Correlator). Reduces # of computations from N<sup>2</sup> to N\*log(N).



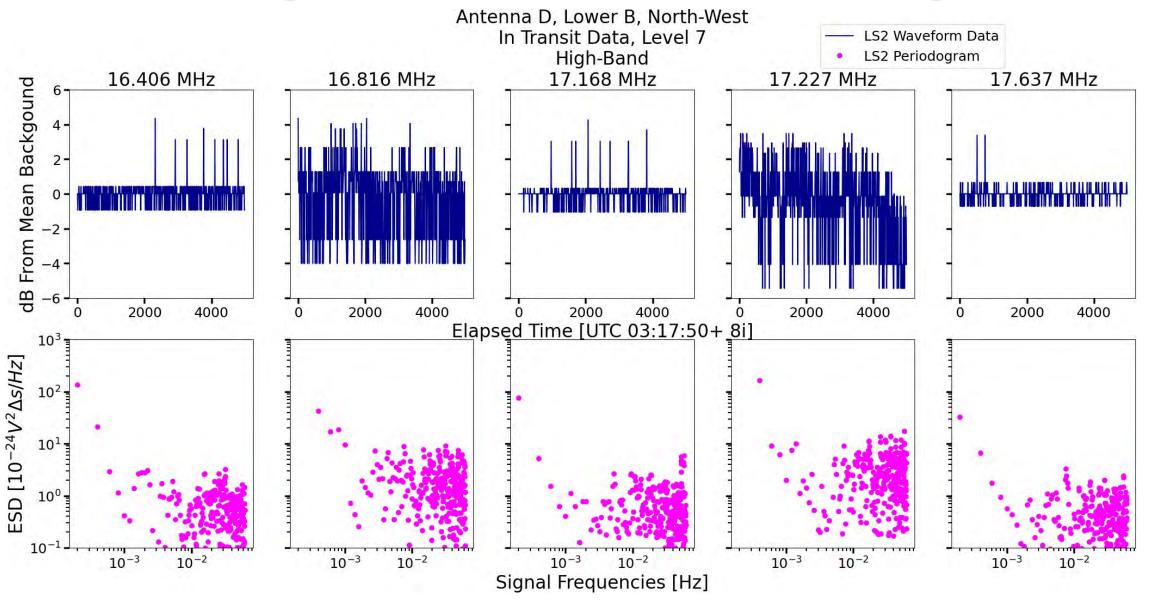


#### FarView Lunar Manufacturing

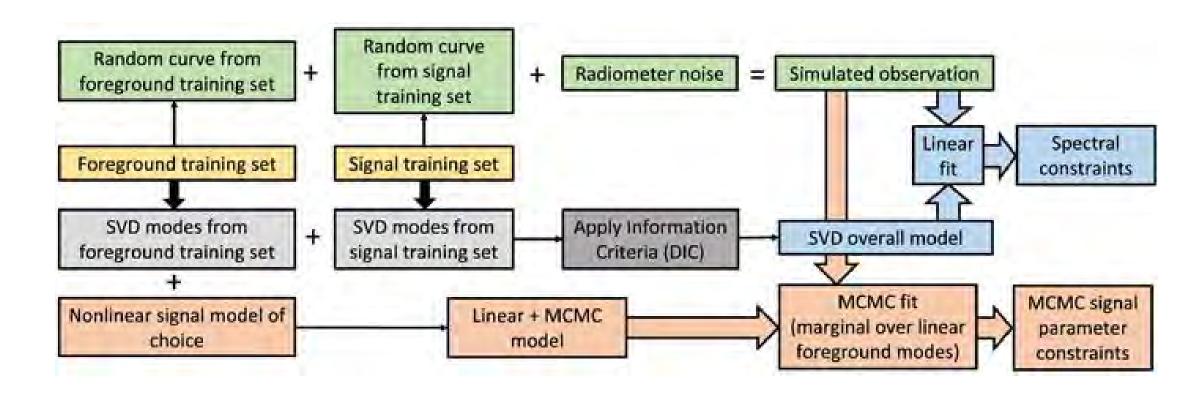
- FarView Observatory Incorporates 100,000 dipole antennas
  - In addition, Power Cables, PreAmps, Data & Comm, Power System.....
- Metallic Components: antennas and power cable wires
  - 1,000,000 m 'wire' for antennas
  - 3,200,000 m wire for power cables
  - Wire mass ~ 0.01 kg/m
  - Total Observatory 'wire' mass ~42 tonnes
- Why bring metallic wire to the Moon?
  - Lunar regolith contains aluminum (aluminum oxide)
  - Reduce oxide and extract aluminum metal on the Moon
  - Use extracted aluminum metal for manufacturing 'wire' for FarView on the Moon
  - Make Observatory more Cost Effective/less Cost Prohibitive



#### Viewing Earth in the Radio: Technosignatures



#### Signal Extraction Pipeline – The forward-model approach



Rapetti et al. 2020, ApJ, 897, 174

## Lessons Learned & Upgrades for ROLSES-2

ROLSES-2 assigned to the CP-21 CLPS manifest, Gruithuisen Domes on lunar near side, in ~2028.

- Upgrade antenna actuator for higher temperatures.
- Internal Cal source to monitor gain variations.
- Improved shielding from internal interference.

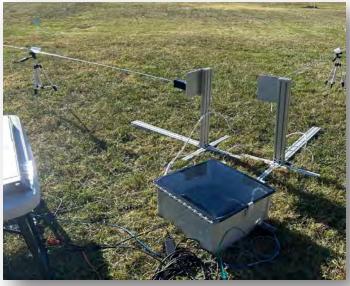
 4 antennas operate as cross dipoles to measure polarization.



Stacer antennas deployed in the Lab.



Spectrometer board with FPGA



Recent field test of Stacer antennas